# Marine Corps Systems Command Combat Support Logistics Equipment and Training Systems Directorate Quantico, VA 22134-5010

January 30, 1997

# Reconnaissance Surveillance Targeting – Vehicle (RST-V) Concept/Requirements Report

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#### 1. Introduction

Although Lockheed Martin Defense Systems (LMDS) has been acquired by the General Dynamics Corporation, this report continues to use LMDS as our company name rather than our new name which is General Dynamics Defense Systems (GDDS). This will be necessary until our Reconnaissance Surveillance Targeting - Vehicle (RST-V) study contract is officially transferred to General Dynamics. This official transfer (contract novation) is expected to occur in February 1997.

This report is considered to be an interim report for the RST-V Study Contract. A final technical report is to be provided in May 1997 upon completing this study contract. The final technical report will include the results described within this report as well as the progress made between now and the end of contract.

The basic top level objectives of the RST-V Study are to:

- Identify concepts, architectures, and technologies supportive of the RST-V System/Segment Specification (Draft, Revision 03, 26 March 1996).
- Identify those currently defined requirements that are not considered to be realizable with technologies that are either available now or will be available for effective use within the RST-V Advanced Technology Demonstrator (ATD) which is scheduled for fabrication and initial testing in FY 1999.
- Provide definition and availability estimates for technologies supportive of RST-V
  Specification requirements. Technology availability will be categorized into three time
  frames: Can be built today or is relatively low risk for RST-V ATD; Can support RST-V
  ATD upgrades (3-5 years out); Can only support developments beyond 5 years out.

Our basic approach to meeting RST-V Study objectives is summarized in Figure 1-1 and includes:

- Assessing information from a number of related activities and source documents.
  - Analyzing user missions and identifying vehicle capabilities supportive of these missions.
  - Formulating and selecting (through trade-offs) preferred concepts at the subsystem and system level.
  - Providing comparisons of the performance capabilities of the preferred concepts to the current RST-V Specification requirements and proposing requirements changes as appropriate. Primary focus is to identify performance requirements and associated technologies necessary to realize capability improvements beyond HTMMP, AEDT, and JTEV as well as being suitable to support the FY 98 RST-V ATD startup.
  - Characterizing and documenting preferred concepts.

This report summarizes RST-V Study progress to date and focuses on the following topics which were suggested at the RST-V Kickoff Presentation held April 10, 1996.

- Requirements Analysis
- Mission Analysis
- Concepts Developed
- System Trade Studies
- Vehicle Weight Analysis

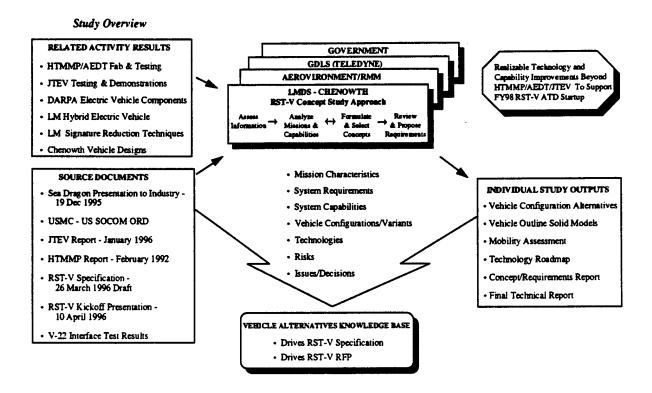


Figure 1-1. RST-V Concept Study Overview

#### 2. Requirements Analysis

Requirements associated with the source documents and some of the related activity results listed in Figure 1-1 were assessed to determine the key capabilities and critical thresholds that new RST-V vehicle and subsystem concepts should support. An overview of these capabilities and thresholds is shown in Figure 2-1. Specific performance requirements are primarily documented in the RST-V Specification (26 March 1996) draft. As subsystem and vehicle level concepts are defined and traded off, their performance is compared to required performance thresholds and concepts are modified or specification changes are recommended as appropriate. In some cases, requirements are relative to performance achieved to date by a related activity. For instance, the RST-V Kickoff Presentation recommended performance beyond HTMMP, AEDT, and JTEV. Therefore, test results of these related activities have been reviewed to ensure concepts considered have estimated performance in excess of these already evaluated test beds.

As this study defines concepts, comments to the RST-V Specification are generated. Comments identified to date are contained in Appendix A and include both performance and document organizational comments. A summary of current findings follows. Final comments to the specification are planned to be issued near the end of the study and will include suggestions for changing performance parameter values based on findings from the design and integration phase of this study.

- Specification is relatively well structured.
- Partitioning performance characteristics by major subsystem is a good idea.

- Suggest RST-V work breakdown structure be updated to be consistent with major subsystem partitions.
- Qualitative statements (minimize, easily, fuel efficient) should be eliminated or quantified wherever possible.
- Paragraphs should be structured so that each numbered paragraph only contains a single requirement. This leads to a clearer requirement set and improves traceability between requirements, validation methods, and associated test plans and procedures.

#### Critical Thresholds/Features Key Capabilities (1) Weight GVW Curb Pavload USMC 6500 5000 1500 · Mobility (Continued) Weight SOCOM 8000 5000 3000 - Ride Quality\* - GVW - Range - Curb Speed Survivability - Payload - Acceleration Mobility - Acquisition Avoidance - Top Speed\* - Penetration A voidance - Ride Quality\* - Speed - Kill Avoidance • Trafficability - % NoGo\* Forward\* · Dash (Top\*) NBC Range Acceleration\* • Firepower (2) - USMC 300 Miles - Tractive Effort\* - SOCOM 450 Miles • C4I (2) - Mobility Rating (%NoGo\*) · > 8 Miles Silent Operation Transportability Terrains • Transport - Tactical\* - Obstacles - Air Transportation • V-22 Internally Transportable Longitudinal Slopes\* Width Side Slopes\* · Height Mobility Verucal Step Flexibility & Expansion - Minimum ≥ HMMWV Fording - Modular Subsystems & - Objective ≥ Maneuver Force - Maneuverability Components Obstacles Turning (Dynamic) Structural & - Tractive Effort\* - Turning (Stanc) Connect/Disconnect Features - Vehicle Cone Index - Slope Operations\* - Growth • Weapons Up in 15 Seconds - Interfaces Logistics Approach Angle - Operations & Support Cost Hybrid Electric Drive Departure Angle • 25 kw Excess Auxiliary Power - Average Unit Rollaway Cost · Ground Clearance · 4 Variants (Personnel, Weapons, (1) Initial Focus for Establishing Common System Baseline Concept (2) No New Concepts Being Developed (Assessed to Establish Space, Weight, & Power Claims) Sensors, Litter Carrier) "Critical Factors" From Kickoff Presentation

Figure 2-1. Key Capabilities & Critical Thresholds

### 3. Mission Analysis

A variety of missions have been assessed and the essential vehicle characteristics of each have been identified. Characteristics include payload, weapons, sensors, communications, and range as well as the relative importance of stealth, mobility, firepower and protection. Missions included USMC Combat Missions for Light Vehicle Operations (Amphibious Raid, TRAP, Airfield Seizure, Recon/Surveillance, Limited Objective Attack), Special Operations Missions (Recon-Special Ops, RST-Short Duration, RST-Scout Mission, RST-Small Unit Ops), Operational Maneuvers from the Sea (Troop Carrier, Fire Support, Command and Control, Logistic Support, Medical Evacuation, Deep Recon and Targeting), and Operations Other than War (Rescue Noncombatants, Show of Force, Peacekeeping). A summary of mission/vehicle characteristics is shown in Table 3-1.

Table 3-1. Mission/Vehicle Characteristics Summary

			SENSORS	RANGE	RANGE	
MISSION	PAYLOAD	WEAPONS	& COMMS	OFF	ON	CHARACTERISTICS
				ROAD	ROAD	
<b>ESTABLISHE</b>		MBAT MISS	IONS FOR L	GHT VEE		
Amphibious	3 Crew	MK19/M2	TTS, NVG	250 mi	100 mi	Stealth, Dash Speed,
Raid	2500 lbs	M60/M240	Comms			Firepower
TRAP	3 Crew	MK19/M2	TTS, NVG	250 mi		Stealth, Offroad Mobility
	1500 lbs	M60/M240	Comms			
Airfield	3 Crew	MK19/M2	TTS, NVG	150 mi		Stealth, Dash Speed,
Seizure	2000 lbs	M60/M240	Comms			Firepower
Recon /	3 Crew	MK19/M2	TTS, NVG	100 mi		Stealth, Dash Speed,
Surveillance	1500 lbs	M60/M240	Comms			Offroad Mobility
Limited	3 Crew	MK19/M2	TTS, NVG	250 mi	100 mi	Stealth, Dash Speed,
Objective	2500 lbs	M60/M240	Comms			Firepower
Attack						
SPECIAL OP	ERATIONS M	AISSIONS (co	mpiled by NS	WC Carde	rock)	
Recon,	3/4 Crew	MK19/M2	TTS, NVG	350 mi	100 mi	Stealth, Mobility,
Special Ops	2000 lbs	M60/M240	Comms			Firepower, Endurance
RST: Short	3/4 Crew	MK19/M2	TTS, NVG	100 mi	50 mi	Stealth, Mobility,
Duration	2000 lbs	M60/M240	Comms			Firepower, Dash Speed
RST: Scout	3/4 Crew	MK19/M2	TTS, NVG	300 mi	100 mi	Stealth, Mobility,
Mission	2000 lbs	M60/M240	Comms			Firepower, Dash Speed
RST: Small	3/4 Crew	MK19/M2	TTS, NVG	300 mi	20 mi	Stealth, Mobility,
Unit Ops ,	2000 lbs	M60/M240	Comms			Firepower, Endurance
OPERATION.	AL MANEUV	ER FROM T	HE SEA - GR	OUND MA	NEUVE	
Тгоор	2 Crew	M60/M240	NVG	150 mi	100 mi	Stealth, Mobility,
Carrier	4 Infantry	Pers Wpns	Comms			Dash Speed
Fire	3 Crew	MK19/M2	TTS, NVG	150 mi	100 mi	Stealth, Mobility,
Support	2500 lbs	Mortars	Comms			Dash Speed
Command and	3 Crew	MK19/M2	TTS, NVG	150 mi	100 mi	Stealth, Mobility,
Control	2500 lbs	M60/M240	Comms			Dash Speed
Logistic	2 Crew	M60/M240	TTS, NVG	200 mi	100 mi	Stealth, Mobility,
Support	3000 lbs	Pers Wpns	Comms			Dash Speed
Medical	2 Crew,	M60/M240	TTS, NVG	150 mi	100 mi	Stealth, Mobility,
Evacuation	4 Litters	Pers Wpns	Comms			Protection, Dash Speed
Deep Recon	3 Crew	MK19 / M2	TTS, NVG	200 mi		Stealth, Mobility,
& Targeting	3000 lbs	M60/M240	Comms			Dash Speed, Endurance
<b>OPERATION</b>	S OTHER TH	AN WAR				
Rescue Non-	2/3 Crew	MK19 / M2	TTS, NVG	100 mi	250 mi	Mobility, Dash Speed,
Combatants	1500 lbs	M60/M240	Comms			Firepower, Protection
Show of	3 Crew	MK19/M2	TTS, NVG	100 mi	250 mi	Deterrence: Speed
Force	2500 lbs	M60/M240	Comms			Firepower, Protection
Peacekeeping	3 Crew	MK19/M2	TTS, NVG	100 mi	250 mi	Stealth, Dash Speed,
_	2500 lbs	M60/M240	Comms			Firepower, Protection

These missions and their essential characteristics were then grouped into four mission areas: Reconnaissance Surveillance Targeting (RST), Strike, Personnel, and Litters/Logistics as shown in Table 3-2. The essential capabilities/features for each mission area were defined across the major subsystems of mobility, firepower, survivability, and C4I (Command, Control, Communications, Computer and Intelligence). Qualitative rankings of mobility capabilities were established across the mission areas and all capabilities were divided into three categories: Capabilities common to all mission areas (Green table entries); capabilities unique to one mission area (Red table entries); and capabilities common to two or three mission areas (Black table entries).

As subsystem and vehicle level tradeoffs are made, these mission/capability categorizations will be used to support initial vehicle configuration selections. In addition to the above mission/capabilities categorizations, our project team has reviewed, discussed and filled out a critical parameters matrix provided by Carderock at the RST-V Concept Study In-process Review #1. Participants included systems, safety, and field support engineers. Survey results for the four functional categories of lethality, supportability, survivability, and mobility are shown in Table 3-3. A summary of our findings follows and specific findings are contained in Appendix B.

- Survey is valuable (filling it out forces priorities to be established).
- Survey participants should be interviewed to ensure rationale for priorities is understood.
- Participant's background and rationale should be documented.
- Present survey form (general in nature) appropriate prior to definition of mission specific vehicle and subsystem concepts.
- Separate survey forms are recommended (each tailored to particular mission area).
- Separate forms, tailored to each mission area of interest could provide more accurate survey conclusions.
- Minimum set of top level filters (mandatory requirements) will help focus developer and could reduce development costs if other requirements are not too stringent or are defined as goals.
- Present method of scoring causes participants to spend too much time dealing with math (suggest rounded percentages).

A more detailed user perspective is needed before making final configuration selections during the design and integration phase of this study. It is planned to obtain this perspective by interviewing ground weapons personnel at Quantico. Topics to be discussed are contained in Appendix C.

Table 3-2. Capabilities/Features Comparison by Mission Area (1 of 2)

	0	. v v
Payload	3 -4 man crew 2000 - 3000 lb payload 100 - 350 mi offroad range 0 - 100 mi onroad range	3 man crew 1500 - 2500 lb payload 100 - 250 mi offroad range 100 - 250 mi onroad range
C4I 0 (7)	Pos Nav Svstem Sincgars (2 Radio) CVC2 Processor? Control & Display Panels EPLRS TAC Satellite? Low Freq Long Distance?	Pos Nav System Sincars (2 Radio) CVCQ2 Processor? Control & Display Panels EPLRS TAC Satellite? Low Freq Long Distance?
Survivability .1 (7)	Acquisition Avoidance: Signature Management Acoustic Engine Suppression Themal Engine Suppression IR False Target Generator? RF Jammer? Ventilated Face Piece Micro Clim ate Cooling? Hit Avoidance: Laser Warning Receiver Radar Warning Receiver Dash Speed Penetration Avoidance: Protection against mines (floors & wheel wells) Kill Avoidance: NBC Standoff Detector Chemical Agent Detector	Acquisition Avoidance: Signature Management Acoustic Engine Suppression Thermal Engine Suppression IR False Tanket Generator? RF Jammer? Ventilated Face Piece Micro Climate Cooling? Hit Avoidance: Laser Warning Receiver Radar Warning Receiver Critical dash speed Penetration Avoidance: Configurable Armor Protection against mines (floors & wheel wells) Kill Avoidance: NBC Standoff Detector Chemical Agent Detector
Firepower .2 (7)	Target Acquisition Capabilities: Acoustic Sensor Seismie Sensor Ref Target Acquisition Equipment Low Light Level TV Visible Video FLIR? Radios S81? Vehicle ID System Laser Rangefinder Driver's Thermal Viewer? 10 Meter Mass? Sight Head w/Azi Drive & Stab Prim ary: MK 19, M2, Javelin or TOW [3] -and-	Target Acquisition Capabilities: Acoustic Sensor Seismic Sensor Seismic Sensor RF Target Acquisition Equipment Low Light Level TV Visible Video FLIR? Radiac Set? Vehicle ID System Laser Rangefinder Driver's Thermal Viewer 10 Meter Mast? Sight Head w/Azi Drive & Stab Prim ary: MK 19, MZ, Javelin or TOW or mortar (41 (3) -and-
Mobility (1)	Dash Speed is in portant: 70 mph 121  Central Tire Inflation Sustained Low Speed is necessary, 3 mph @ 3 hrs [4] Forward Speed is very important [3] %NoGo is very important [3] Tractive Effort is in portant [2] Slope Stability is very important [3] Ride Quality is very important [3] Semi-Active or Active Suspension?	Dash Speed is necessary; 70 mph [4]  Central Tire Inflation Sustained Low Speed is not in portant [0] Forward Speed is very important [3]  %NoGo very important [3] Tractive Effort very Important [3] Slope Stability very in portant [3] Slope Stability is important [2] Semi-Active or Active Suspension?
LCC	·	
Mission Area	RST: Recon/Surveillance Recon/Surveillance RSTA: Short Duration RSTA: Scout Mission RSTA: Small Unit Ops Deep Recon & Targeting	Strike: Amphibious Raid TRAP (2) Arfield Seizure Limited Obi, Attack Fire Support Show of Force Peacekeeping

Table 3-2. Capabilities/Features Comparison by Mission Area (2 of 2)

					,	Doolood
	,	Mobility (I)	Firepower	Survivability	0 (7)	a a section of
Mission Area	֝֝֝֝֝֝֝֝֝֝֝֝֝֝֝֝֝֝֝֝ ֓֓֞֞֞֞֞֞֞֞֞֞֞֞֞֞֞֞	1 (7)	.2 (7)	777	Control & Display Panels?	2/3 man crew
	4 (7)		Secondary: M 60 or M 240G	Acquisition Avoidance:		/ 4 infantry
Descound!		Speed is very timportain		Signature Management?		
		<u> </u>	Man Dainess Dennired	Acoustic Engine Suppression?		1500 lb
			No rillianty Nodello	Thermal Engine Suppression		Professional
Decine Non-		Central Tire Inflation	The Target Aconisition	IR False Target Generator?		200.684
Nescue item			Limited in Ber Addition			150 mi
Companie		Sustained Low Speed is not	Equipment.	Hit Avoidance: N/R		offroad range
		important [0]				100 - 250 mi
				Penetration Avoidance:		onroad range
		Forward Speed 15 very till politails		Protection against mines (110015		
		[3]		& wheel wells)		
		13) Instruction in contrast (3)				
		None in the state of the state		Kill Avoidance:		
		Tractive Effort is important [2]		Chemical Agent Detector		
		Slope Stability is important [2]				
		Ride Quality is very important				
				A acutation A voidance:	Control & Display Panels?	Z man crew
Litters / Lopistics:		Dash Speed is important [2]	Secondary: M60 or M2400	Signature Management?		4 litters
		i	No Primary Required	Acoustic Engine Suppression		3000 lb
Logistic Support	_	Central Tire Inflation		Thermal Engine Suppression		payload
Medical Evacuation		Con of Lease S and I to a second	Limited Target Acquisition	IR False Target Generator		
		Sustained Low opera is not	Enuipment:			150 - 200 mi
		im portant   U	Vehicle ID System	Hit Avoidance: N/R		offroad range
		Forward Speed is very important		penetration Avoidance:		
-		[3]		Protection against mines (floors		100 mi
				& wheel wells)		9
		% NoGo is importanct [2]				
		Tractive Effort has minimal		Kill Avoidance: NBC Standoff Detector		
				Chemical Agent Detector		
		Stope Stability has minimal importance [1]				
		Ride Quality is very important				
		[3]				

Mobility requirements are ranked in order of importance. These rankings are shown by a number in brackets ([4] Mandatory/Critical; [3] High Value/Very Important; [2] Medium Value/Moderately Important; [1] Limited Value; [0] No Value).
 TRAP requires the capability to pick up and transport troops (on the order of 1 - 3).

(3). Javelin, TOW or Mortar weapons are for Indirect Strike missions.

(4). A question mark following an entry means that the item's need is unknown.
(5). The Command & Control Mission is unique and doesn't fit into one of the four main mission areas.
(6). All mission areas require personal heaters.
(7). These values are the current weightings from the draft RSSV Specification.
(7). These values are the current weightings from the draft Roson aftributes. and black for attributes that support two or three mission areas.

Table 3-3. Critical Parameters Survey Results

LSV Survey Participant User Average Systems Field Safety Systems Systems **Function** Systems Value Engr Engr Engr Engr Engr Engr (4)(3) (2)(1)(1) (1)35 .40 .30 35 .30 .40 Lethality 18 .20 .20 .20 .20 10 Supportability 18 20 25 .25 10 Survivability 10 .29 .30 30 .25 30 .30 Mobility

			RST	<u>A-V</u>				
			Sur	vey Participa	int			4
Function	Systems Engr (1)	Systems Engr	Systems Engr (1)	Safety Engr (2)	Field Engr (3)	Systems Engr (4)	User	Average Value
Lethality	.10	.25	.20	.20	.15			.18
Supportability	.20	.15	.20	.20	.25	_		.20
Survivability	.40	.30	.30	.40	.25			.33
Mobility	.30	.30	.30	.20	.35	-		29

			Gen	eral				
			Surv	vev Particina	int			4
Function	Systems Engr (1)	Systems Engr (1)	Systems Engr (1)	Safety Engr (2)	Field Engr (3)	Systems Engr (4)	User	Average Value
Lethality	.20	.20	.25	.35		.14		.228
Supportability	.30	.10	.20	.15		.08		.166
Survivability	.20	.30	.25	.20	-	.18		.226
Mobility	.30	.40	.30	.30	<u> </u>	.60		.38

- (1) Land Combat Systems Engineering
- (2) Safety Engineering
- (3) Field Support Engineering
- (4) Mission/Vehicle Capabilities Analysis

# 4. Concepts Developed

### 4.1 Technical Approach

An overview of our technical approach is shown in Figure 4-1. The Key initial tradeoff is between tracked and wheeled vehicle topologies followed by standard drive train and initial vehicle level solid modeling and associated assessments. These trades and selections are driven by key RST-V specification requirements, mission analysis results, earlier described related activity test results, as well as a survey of today's technologies. Starting with a relatively well known standard propulsion system allows a solid basis of essential constraints and performance characteristics to be established before advanced propulsion (including hybrid electric drives and active suspension) systems are concepted and assessed both at the subsystem and vehicle levels. The NATO Reference Mobility Model (NRMM) will be applied to confirm standard propulsion system assessments already performed, as well as the advanced concepts being assessed during the design and integration phase of this study. In addition, the Hybrid Electric Combat Vehicle (HECV) Software is being applied to evaluate alternate hybrid electric drive configurations. A

survey of advanced technologies in the areas of survivability, firepower and C4I is also being conducted and the preferred technologies will be selected based upon an assessment of their performance against requirements. Selected technologies will also be incorporated within vehicle level solid models to assess space and weight claim impacts prior to defining preferred concepts. Results to date are covered in the following paragraphs.

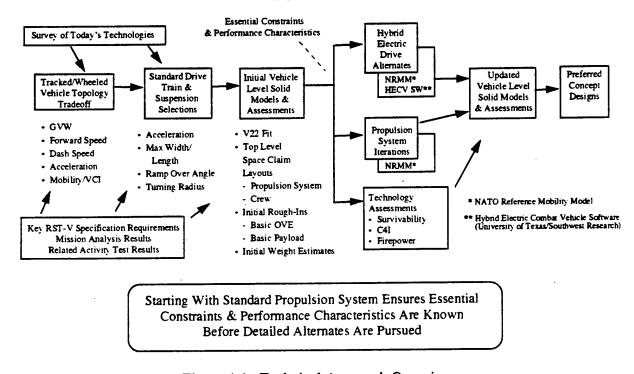


Figure 4-1. Technical Approach Overview

#### 4.2 Standard Propulsion System

The selection of standard propulsion system components was driven primarily by speed and mobility requirements. Starting with the ground contacting elements, it was determined that in order to meet the mobility and VCI requirements, a minimum of a 4x4 all wheel drive (AWD) configuration with 33" diameter tires would be required. At an assumed 95% tire radius, the estimated "static load driving radius" equals 1.31 feet and the tire's revolutions per mile is 611. Power to the wheels would be obtained by a conventional four wheel drive approach as shown in Figure 4-2.

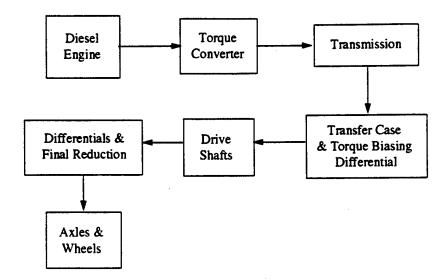


Figure 4-2. Diesel Propulsion System & Drive Train Block Diagram

Sizing of the remaining drivetrain and propulsion system components was calculated with the following assumptions:

- curb weight = 3500 lb (max GVW = 6500 lb)
- torque is biased to equal the weight distribution under all circumstances
- frontal area = 80% (W x H) = 60° x 66° X  $0.8 / 144 = <math>22 \text{ ft}^2$
- overall drivetrain efficiency (flywheel to tire patch) = 80%
- drag coefficient = 0.75 (note passenger cars range from 0.35 0.45, trucks = 0.7 and trailers = 1.2)
- coefficient of rolling resistance:

$$f_{sand} = 0.26$$

$$f_{\text{med soil}} = 0.10$$

$$f_{pavement} = 0.03$$

With the assumed gross vehicle weight, drag coefficient and rolling resistance, it was possible to estimate the required power to meet the acceleration requirements. This was found to be 168 hp to meet 0 - 30 mph in 6 seconds and 211 hp to meet 0 - 60 mph in 15 seconds. This power, together with the choice of being automatic, limits the choice of transmissions to either the GM 4L80E (a 4 speed automatic with overdrive) or the Mercedes Benz (5 speed automatic with overdrive). Each costs about \$2500 and weighs about 250 pounds. The GM was chosen over the Mercedes as it is currently being used in the HMMWV and offers better supportability in the US. It has the following gear ratios: first = 2.48:1, second = 1.48:1, third = 1.0:1 and fourth = 0.75:1. The remaining drivetrain ratios were assumed to be 4.56:1 as a final reduction, 1:1 in the transfer case and 1.9:1 in the torque converter at lockup conditions. The 4.56:1 final reduction is typically a 41 tooth ring gear meshed with a 9 tooth pinion gear at each front and rear axle. A 1:1 transfer case torque biasing capability is common and available through sources such as Borg-Warner, Dana and New Venture Gear. The torque converter is a hydro-kinetic unit with a

maximum multiplication of 1.9 at start-up conditions. This drops to 1:1 (no multiplication) typically around second or third gear for minimal losses. The acceleration requirement was found to be the dominant driver in terms of propulsion system sizing and is shown versus GVW in Figure 4-3. The remaining speed and mobility requirements were found to result in the following power demands:

- Drawbar Pull 197 ft - lb
- 75 mph Top Speed 109 hp (81 kW) @ 2613 rpm
- Start on 60% Slope 277 ft-lb
- Climb 5% at 40 mph 139 hp (103 kW)
- Tow equal up 40% slope 6.42 mph @ 139 hp

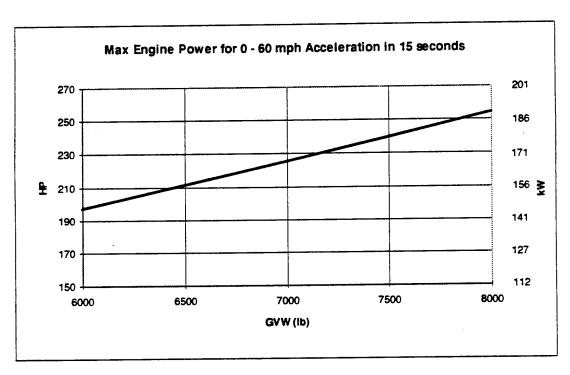


Figure 4-3. Maximum Engine Power (HP) vs. Gross Vehicle Weight (GVW)

An industry survey of diesel engines is included in Appendix D. From this listing, a preferred subset has been identified and is included in Appendix E.

The effects of efficiency on maximum vehicle speed is shown in Figure 4-4. From this it can be shown that with the above mentioned assumptions, a conventional propulsion system would require about 109 hp to achieve 75 mph top speed with 80% efficiency and at a gross vehicle weight of 6,500 lb while a more efficient propulsion system, like one might expect with a hybrid-electric system, the required power is reduced to 89 hp. Similarly, at a GVW of 8,000 lb, the required powers are 120 and 101 hp respectively.

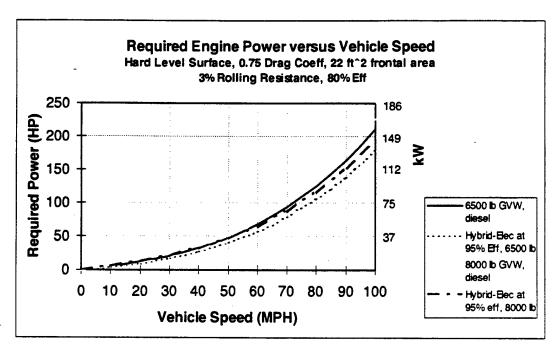


Figure 4-4. Required Engine Power vs. Vehicle Speed

In Figure 4-5, maximum vehicle speed is shown as a function of GVW for three given engine power sizes. This data is based on hard level driving as might be experienced over primary roads and superhighways.

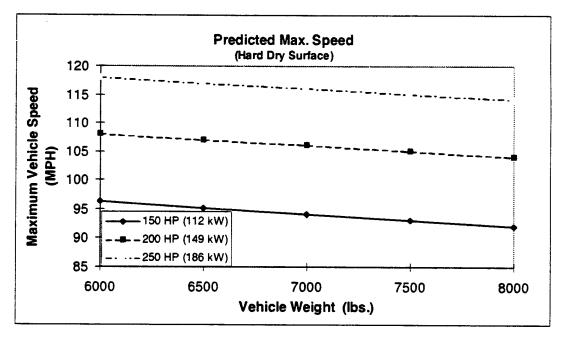


Figure 4-5. Predicted Maximum Speed (Hard Dry Surface)

Once off-road conditions are encountered, the maximum vehicle speeds must be reduced for various reasons such as driver comfort and safety. In order to be able to drive quickly on adverse surfaces, it may be necessary to employ a central tire inflation system (CTIS). The effects of

varying the tire pressure on vehicle speed are shown in Figure 4-6. This shows that a 6500 lb vehicle (GVW) at a typical tire pressure of 24 psi could travel at highway speeds with a tire deflection ratio of 10 - 20%. The VCI in this case would be about 22. CTIS is one alternative to enable the driver to travel on soils with a cone index of less than 22. For example, in the 4x4 configuration shown in Figure 4-6, the VCI can be lowered to about 17 by lowering the tire pressure to 15 psi which would reduce the deflection ratio to about 35% - 40% which is the maximum lower limit for which most tires are rated. This would reduce the vehicle's speed, perhaps to 5 - 10 mph, but would allow for continued driveability. VCI is covered in detail in paragraph 5. Of course, the tire's pressure can be manually lowered in pressure but this takes time and the driver must stop the vehicle to do so. CTIS allows the driver to adjust tire pressures up or down while on the run. The amount of time required to inflate the tires is a function of the size of the on board air compressor. One drawback of CTIS is that it increases the "unsprung" weight due to the rotary pneumatic joints required at each hub. Most modern CTI systems have an electronic controller which enables the operator to select desired pressure within the recommended limits from the tire's manufacturer.

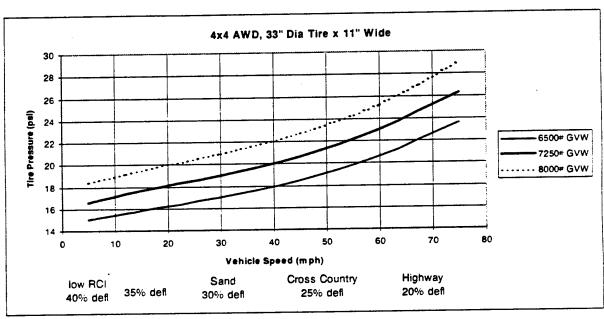


Figure 4-6 Effects of Tire Inflation on Mobility

Calculations used to determine performance characteristics necessary to achieve maximum speed, starting on 60% slope, climbing a 5% grade at 40 mph, towing self and equal weight up 40% slope, as well as acceleration requirements are contained in Appendix F.

#### 4.3 Hybrid Electric Drives

#### 4.3.1 Hybrid Electric Concepts

Four hybrid electric drive concepts have been considered for the RST-V application. All concepts provide both the ability to run directly off of an electric power generating prime mover as well as off stored energy required for stealth mode. These concepts are shown in Figure 4-7.

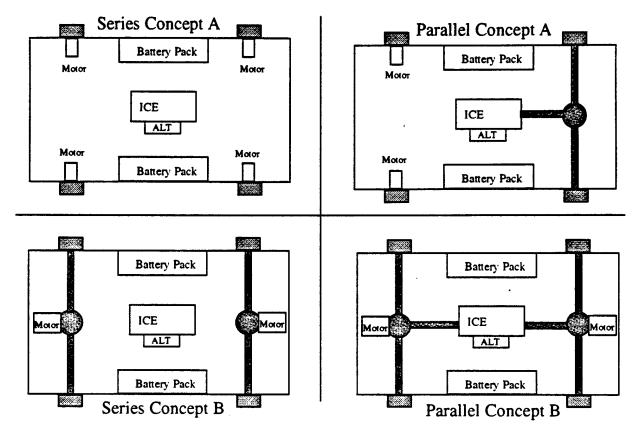


Figure 4-7. Hybrid Electric Vehicle Candidate Concepts

#### Series Concept A

This series concept utilizes an individual drive motor for each wheel with an APU sized for the average power demand. This configuration allows individual control of each wheel rate as well as torque. No mechanical linkage from the prime mover is used which allows the prime mover to be operated at its most efficient operating point. Stealth (or all electric mode) utilizes energy stored in batteries, flywheels, ultracapacitors, etc. Motors are sized for continuous power. Peak power is approximately 4 times continuous rating. Control of each wheel's speed is done by computer algorithm. Wheel speed during turning must be controlled fairly accurately to prevent counter torques. Wheel rate control allows traction and skid control. Rate control also prevents overspeeding/underspeeding of the wheels if the vehicle should leave the ground and avoids the counter torques responsible for sheering driveshafts on landing. The energy storage medium (typically batteries) is sized for prolonged stealth operation (20 miles). All four wheels are available for regenerative braking.

#### Series Concept B

This configuration is similar to Concept A except for the use of mechanical differentials for ease of control. Individual wheel rate control is not possible with differentials. Torque control is implemented. Two motors also simplify the electronics. As in Concept A no direct mechanical connection exists between the prime mover and the wheels. The use of differentials is a significant drawback due to added weight and the tendency to lockup or skid a wheel during low speed turning.

#### Parallel Concept A

Parallel configurations have the ability to employ both mechanical and electrical propulsion either individually or in concert. This concept utilizes a combination of standard mechanical linkage transmission for driving the rear (or front) wheels while the front wheels are driven by electric motors. The parallel connection is via the road. This configuration takes advantage of the fact that hybrid electric drives are more efficient for urban driving due to the regeneration of energy through braking, however are actually less efficient for highway driving due to the additional energy conversion which takes place. With this configuration, the prime mover provides the continuous power for the long haul across country while the electric drives are used for hard acceleration requirements and stealth mode (electric operation only). The prime mover would be sized for slightly more than average power.

#### Parallel Concept B

This configuration is a fully redundant topology which uses direct mechanical linkages and transmissions to route mechanical and or electrical power to the wheels. Here the electric motors would supply additional torque on demand as well as propulsion in stealth mode. The use of differentials prevents individual wheel control and the added weight due to the redundancy is a significant drawback. Redundancy, while providing total backup also adds twice the complexity.

A summary of the salient features of each concept appears in Table 4-1.

#### 4.3.2 Hybrid Electric Preferred Concepts

The Downselect process eliminated two (2) concepts. The preferred concepts are Series Concept A and Parallel Concept A.

Series Concept A benefits from fully independent wheel control (rate, torque and regenerative braking). The elimination of the differentials makes this possible. The four (4) motors provide full stealth mobility, drive redundancy and component placement flexibility.

Series Concept B was eliminated primarily due to limitations associated with the differentials (i.e. no rate control, skidding in turns, drivetrain stress on landings, etc.).

Parallel Concept A retains a mechanical link to the wheels to reduce fording risk while maintaining a simple separation of the electrical and mechanical system. This avoids the use of a "summing gear" to combine mechanical and electrical power. The electric motors are used as generators under normal driving. They provide the additional torque for hard acceleration and hill climbing. They provide limited stealth capability when used by themselves.

Parallel Concept B was eliminated primarily due to additional complexity, weight and volume of full mechanical redundancy.

Table 4-1. Hybrid Electric Drive Concepts Advantages/Disadvantages

HED	Advantages	Disadvantages
Series Concept A	Common Components Redundant Drive Motors Independent Wheel Torque Control Exceptional Skid Prevention Simpler Mechanical Systems Most Flexible Control System Most Flexibility for Component Placement IC Engine Sized for RMS Power Most Efficient for Stop and Go Operation	Higher Fording Risks (No Mech Drive) More complex Electrical/Control System Larger Alternator Required More Complex Electrical Motor Cooling
Series	Simple Locking Differentials	Additional Gear/Axle Weight
Concept B	Simple Control System	Higher Fording Risks (No Mech Drive)
Discarded	Simplest Integration	Differential Gears - Torque Control Only
	,	No Independent Traction Control
Parallel	Simple Separation of Mech./Elec Sys	Degraded Climb Traction in Stealth
Concept A	Electric Drive Invoked Under Hand	Degraded Acceleration/Top Speed in Stealth
	Acceleration or Climbing	
	High Survivability - Redundant Systems	Additional Gear/Axle Weight
	Most Efficient at Highway Speed	Only Partial Regenerative Braking
	Fording Risks Reduced	Differential Gears - Poor Torque Control
ļ	Motors Can Be Used as Generators	
	Small or No Alternator	
	Minimum Cooling	
Parallel	Most Acceleration Torque Available	High Complexity (Summing Gears, Mech/Elec Systems)
Concept B	Highest Survivability - Fully Redundant	Most Weight
Discarded	AWD	Most Volume
	Smallest Alternator	Summing Gears - Poor Torque Control
		Largest Battery Pack Required For Same Stealth Range

#### 4.3.3 Hybrid Electric Subsystem Component Sizing

Development of the required capabilities for the individual hybrid electric subsystem components was derived at the top level from the following vehicle performance requirements. These performance criteria were also used for the standard propulsion system analyses discussed in paragraph 4.2.

Drawbar Pull 0.4 TE/GVW
Max Speed 75 mph
Start On Slope 60%
Climb Slope 5% at 40 mph
Tow Equal Weight, 40% Slope
Acceleration 0 to 30 mph 6 sec 0 to 60 mph 15 sec

Sizing of the Hybrid Electric drive major components was based on the vehicle level performance requirements with the following assumptions:

- The parallel configuration will have reduced performance in Stealth Mode since only two wheels are driving.
  - Slope 40%
  - Max speed 30 mph
  - Limited tow/drawbar pull
  - Limited traction
- Motors sized for maximum continuous power required to maintain maximum speed.
- Peak motor power / torque is 4 x continuous (i.e., rated) for 60% slope and acceleration (15 sec 0-60).
- Prime power (IC engine) is sized for maximum continuous power + aux. power + losses.
- Electric range goal is 20 miles.
- 25 kw is available for auxiliary power in hybrid mode with 1 kwh reserved for auxiliary power in stealth mode. Limited auxiliary power available during heavy acceleration / hill climbing.
- Electric motor weight is 2 lbs/hp (non-liquid cooled desired).
- Single gear mesh reductions (98.5% eff.)
- Power distribution efficiency is 99%
- Electric motor efficiency is 93%

Table 4-2 lists the power requirements for the major components for each of the two candidate concepts.

Table 4-2. Hybrid Electric Subsystem Component Power Requirements

Components	Series Concept	Parallel Concept
Prime Mover/Alternator - includes propulsion and 25 kW aux	105 kW (140 Hp)	107 kW (143 Hp)
Motors (ea.) rated - Peak	16.4 kW (22 Hp) - 4 X rated (4 motors)	14 kW (19.5 Hp) - 4 X rated (2 motors)
Drive Train	15:1	30:1 (Electric)
Energy Storage	20 kwh	11 kwh
Power Electronics (peak)	310 kW	137 kW

# 4.3.4 Hybrid Electric Subsystem Technologies

The subsystem component sizing serves as a basis for the assessment of the appropriate technology candidates. The five main areas of focus are:

- Prime mover / alternator diesel, gas turbine, fuel cells, etc.
- Motors (propulsion) brushless technology
- Drive train gearboxes (if any), drive shafts
- Energy storage batteries, ultracapacitors, flywheels
- Power electronics inverter technology, processor architecture, power management

Each of these areas is being investigated to find the leading edge technology and determine the best fit for the application. The selected technology must be available within 2 years to support the RST-V ATD program.

#### 4.3.4.1 Prime Mover

A survey of available energy sources for prime power covered a wide variety of energy conversion technologies. Standard technologies (Diesel engines, Rotary engines and gas turbines) were the focus, however more radical technologies were also reviewed including Solar, Nuclear, and Fuel Cell technology. A synopsis follows:

- Diesel; Mature technology, high efficiency, but heavy, poor emissions, noisy.
- Rotary; High power density, moderate efficiency, but questionable emissions, seals.
- Turbine; Very high power density, low noise, multiple fuels, but high cost.
- Fuel cell; Very efficient, but low power density, complex, fuel logistics complex.
- Solar; Free energy, but low power density, area intensive, weather dependent.
- Nuclear; High power, but dangerous, high volume, complex control.

When considering the internal combustion engines, the rotary technology has advanced in the area of emissions and seals. Rotary technology has a high power-to-weight ratio (about 0.6 hp/lb) because reciprocal motion does not have to be converted into rotary motion as done with conventional engines. Focus is being placed on the latest multi-valve, electronically controlled diesel engines, rotary engines, and small, scaleable turbines.

Of the new technologies fuel cells are the most attractive, but lack the power density (even when considering near term advances) demanded by the operational scenarios for this vehicle. Solar has the obvious limitation of no night time operation as well as weather dependence, in addition to the lack of power density. Nuclear power is dangerous, difficult to control and is scale limited with the current and near term technology considering the prime mover volumetric allocation.

The specific energy density of various fuels is shown in Figure 4-8. A quick look at the fuel options that are available today, indicates that the fuel of choice remains diesel for RST-V applications. Although hydrogen (used for fuel cells) has significantly higher energy density, it is presently not practical as a prime mover fuel since it is difficult to transport in liquid form and there are inefficiencies associated with generating hydrogen from other on-board fuels.

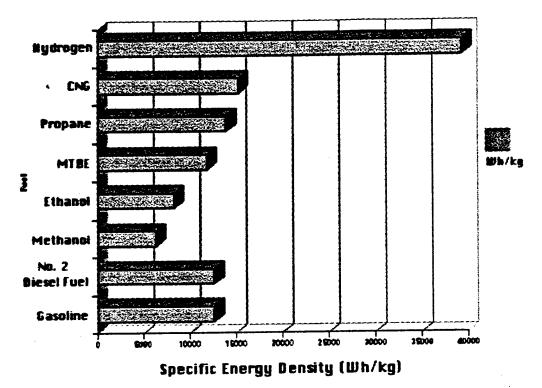


Figure 4-8. Specific Energy Density of Various Fuels

Since no prime mover is ideal, a careful consideration of the respective advantages and disadvantages is in process.

#### 4.3.4.2 Motors

Several types of propulsion motors are being considered. The motors have to be rugged and reliable, hermetically packaged, have a high power density (> 1hp/lb), high efficiency to minimize cooling requirements, and sized for both the peak and continuous power required. The motor must also be high speed to allow the use of a fixed gear ratio transmission. Rotor construction limits must be compatible with motor maximum speed requirements.

All the candidate technologies are brushless due to the higher efficiency and thermal characteristics. However, brushless motors require that commutation of the fields be done electronically which increases the complexity of the electronics. Advances in the motor control technology have made this tradeoff worthwhile. A synopsis follows:

- Induction; Simple rugged design, low cost, high speed, but complex control electronics.
- Switched Reluctance; High performance, high speed, but high cost, complex control.
- Synchronous Reluctance; Emerging technology, uses simpler control electronics.
- Permanent Magnet DC; Easiest brushless to control, low speed efficiency, but exotic magnets are expensive, speed limited.
- Brush DC; Easiest to control, low cost, but low efficiency, brush wear, thermal issues.

Of the above motors reviewed, three are considered suitable:

- AC Induction; Currently favored technology due to motor simplicity and durability (no magnets). Speed control is more complex due to the fact that these motors are normally designed to run at a fixed speed where torque is proportional to rotor to rotating field slip. Sophisticated control techniques have been developed in the last five years which allow smooth rate control. The induction motor is a high speed motor.
- Permanent Magnet DC; Rotor consists of permanent magnets and is synchronized to the field rotation (motor currents are DC at stall). Simplest to control. High efficiency at low speeds. Motor speed limited by adhesion of magnets to rotor shaft. Newer technology rotor designs have increased top speeds.
- Synchronous Reluctance; New technology motors which combine the simple and rugged rotor design of a switched reluctance motor (a single piece) with the simple control of a Permanent Magnet motor with out the magnets. Very high speed. This is a prime candidate.

#### 4.3.4.3 Drive Train

Several technologies are being evaluated for meeting gear reduction requirements. These approaches will minimize volume and have high power density. Consideration has also been given to motor in the wheel technology, however unsprung weight must be minimized for maximum mobility.

#### 4.3.4.4 Energy Storage

Table 4-3 shows the relationships between specific energy, specific power and cost for a variety of energy storage mediums. The specific energy and specific power requirements of this vehicle could require a mix of technologies.

Table 4-3. Energy Storage Comparison

• Requirement: 20 k	Wh, PWR	Selection Criteria Near Term (	(Far Term)
	Specific Energ	Specific Power	Cost
• Batteries	Wh/kg	W/kg	\$ÆWh
- Lead Acid	35 (48)	<b>75</b> 0	150
– NiMH	60 (95)	300	400
- Lithium	120 (500)	500 - 2K	10K (250)
- Zinc/Air	80	100	75
- Sodium/Sulfur	100	500	400 - 500
<ul> <li>Ultracapacitors</li> </ul>			
- Organic (~100 V)	4	2000 (peak)	5K - 10K
- Aqueous (~400 V)	5	2000 (peak)	4K - 8K
• Flywheels (gimbaled	i) 70	200	>10K
(or 2 counter rotating)			
• Superconductors	TBD	TBD	TBD

#### 4.3.4.5 Power Electronics

Four power electronic areas are being studied:

- Power devices
- Processors and low level electronics
- Power management software
- Control software

Power device technology has been making steady progress. The latest advances in IGBT (Insulated Gate Bipolar Transistor) technology makes them the standard device selection for the high voltage end. MOSFETs are still heavily used in the low voltage sections due to efficiency advantages. The most exciting advances are in the Silicon Carbide (SiC) devices, which allow for extremely high voltage and temperatures and therefore, smaller power devices.

Processors and low level electronics today are sufficiently mature to support the planned distributed control architecture. Motors that require electronic commutation algorithms need high speed processing at high motor rates.

Power Management Software is immature in the hybrid electric vehicle arena and will require development focus. Currently power management software is written for each specific hybrid configuration.

Control software will also need customization for the particular topology selected, but the current selection of high speed processors (and related chip sets) will support planned algorithms and architectures.

# 4.3.5 Modeling and Simulation

Some preliminary modeling was developed with the Hybrid Electric Combat Vehicle (HECV) Tool Box developed by the University of Texas. The intent for this software package is to evaluate the HEV concepts with regard to performance and energy management. The HECV Simulink software is very powerful, incorporating very detailed component models and provides an excellent, graphical user interface. Albeit powerful, the tool has two basic disadvantages. First, small changes can have a large ripple effect and secondly, the controller for each configuration needs to be specifically developed. Figure 4-9 is a sample model of a hybrid system and shows the individual blocks which can be mixed and matched to simulate the desired system.

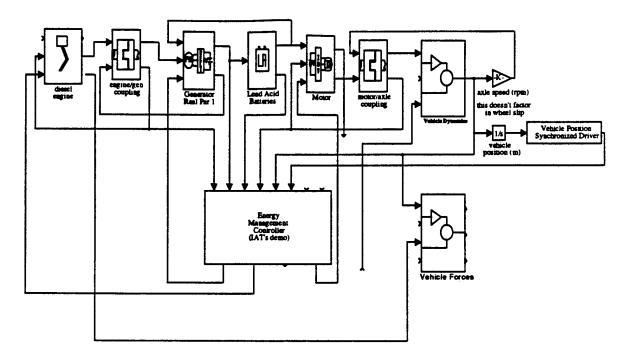


Figure 4-9. Sample HECV Tool Box Hybrid System Model

#### 4.4 Survivability, Firepower, C41

Our approach to date has been a bottoms-up approach based on mobility being the subsystem most critical to a superior vehicle concept. Concept development has focused on ensuring that key mobility requirements are met and that basic crew and firepower needs are met. Once the preferred hybrid electric drive concept has been detailed and incorporated with our RST Vehicle solid model and after incorporating built-in (mission dependent) armor as well as basic crew and firepower elements, future/emerging advanced survivability, weapons and C4I technologies will be incorporated within the remaining space claim areas of the RST Vehicle model.

In parallel with concepting and modeling preferred hybrid electric drive alternates, an independent research and development (IR&D) technology survey is being performed to identify the space, weight, power, and performance characteristics of candidate reconnaissance and light strike vehicle technologies primarily in the areas of survivability, firepower, and C4I. Initial focus will be on acquisition avoidance, threat detection, and target acquisition (reference Appendix G). Appendix H contains preliminary survey results in the survivability areas of acquisition avoidance and threat detection. Appendix I contains preliminary results in the area of firepower.

In reviewing and selecting appropriate technologies from this survey's results, our primary philosophy will be to:

- Focus on implicit stealth design features rather than bulky/heavy add-ons.
- Emphasize passive vs. active techniques due to limited space constraints.
- Pursue C4I technologies that facilitate "seeing before being seen/hit" without significant impacts on space/power claims.

In the armor area, the following considerations have been established:

- Weight, cost, and space claim make it impossible to armor up against all threats:
  - Heavy HMMWV cannot stop 7.62 AP.
  - Full protection against 7.62 ball, mortar fragments, and anti-personnel mines weighs 1700 lbs and costs about \$100,000.
  - Armor detracts significantly from payload and mobility.
- Differing amounts of armor are best for differing missions:
  - Direct actions and shows of force require greater armor protection.
  - Reconnaissance, surveillance, etc. require mine protection combined with avoidance techniques and mobility.
- A mix of integral and configurable armor appears best:
  - Integral armor for wheel wells, floor panels, lower side panels, etc. Where mine protection is needed and attachment is difficult.
  - Add-on panels and windows as required.
  - Transporting, maintaining, and storing configurable armor panels is a logistics burden.
- End user perspective must be incorporated into design trade-offs to ensure appropriate selections are made. It is planned to obtain this perspective by interviewing ground weapons personnel at Quantico. Topics to be discussed are contained in Appendix C.

Additional armor information is contained in paragraph 6.

#### 5. System Trade Studies

The initial topology trade studies which drove the remaining vehicle design concepts were whether the vehicle would be wheeled, tracked or half track. The speed and mobility requirements were assessed and evaluated in order to determine the preferred configuration. Intuitively, tracked and multi-axle wheeled configurations are heavier than a 4x4 AWD configuration and therefore do not score as well in terms of top vehicle speed, dash speed or acceleration. Tracked vehicles also have turning limitations which require that their length to width aspect ratio be between 1.7 and 1.9 which could be a limitation in cargo carrying capacity given the fact that the width is restrained to about 62° to fit the V-22. Conversely, tracked vehicles have better mobility in terms of lower VCI as shown in Appendix J (Extracted from Technical Report GL-88-16; Mobility Analysis for the TRADOC Wheeled Versus Tracked Vehicle Study - September 1988). A summary of similar findings is provided in Table 5-1.

From this study, it was evident that a 4x4 configuration was best in most areas. Mobility in terms of terrain passability and VCI was the only area where the 4x4 configuration was surpassed by all other candidate topologies. It was therefore concluded that a 4x4 configuration would be the best choice if the vehicle could meet the required VCI of 22.

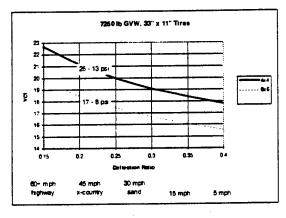
	Topology						
Criteria	Full- Track	Half- Track	4x4 Wheeled	6x6 Wheeled	8x8 Wheeled		
Acoustic Noise	Poor	Poor	Best	Better	Better		
Cost & Complexity	Poor	Good	Best	Better	Poor		
Space Claim	Best	Good	Better	Poor	Poor		
Mobility (Speed)	Poor	Poor	Best	Better	Good		
Mobility (VCI & Terrain)	Best	Better	Poor	Good	Good		
Ride Ouality	Роог	Good	Best	Better	Better		
Vulnerability*	Poor	Good	Good	Better	Best		
Reliability	Poor	Good	Best	Better	Better		

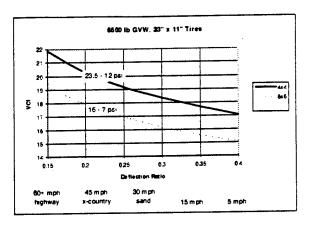
Table 5-1. Tracked vs. Wheeled Qualitative Comparison

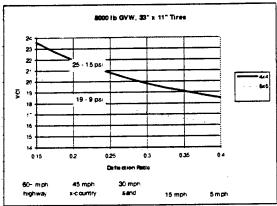
In order to determine if the 4x4 configuration would meet the VCI requirement, a study was performed using the mobility index calculations presented in report AMCP 706-356 and the VCI calculations reported in ADA 297810. Equations used as well as a sample calculation are shown in Appendix K. The following information should help put VCI into perspective. CI refers to soil cone index. The higher the soil cone index, the higher the soil strength. VCI is the minimum value of CI at which a vehicle can successfully complete N passes in the same ruts on level ground at a slow, steady speed. Generally, the lower the VCI, the better the vehicle's mobility.

- VCI is proportional to both ground pressure and tire pressure. As a rule-of-thumb, it is roughly equal to these values (within 25% or so). This is why wheeled vehicles with tire pressures in the range of 20-30 psi have VCIs around 20-30 and why tracked vehicles with ground pressures under 10 psi have VCIs under 10.
- CI less than six leaves a footprint greater than two inches deep due to ordinary walking.
- CI between six and 15 leaves a two inch deep footprint.
- CI between 15 and 23 leaves a one inch deep footprint.
- CI between 25 and 50 leaves a well defined footprint.
- Soil with a CI of about 100 can be easily penetrated with your thumb.
- Soil with a CI of about 150 can be penetrated with moderate thumb pressure.
- Soil with a CI of about 200 is very difficult to penetrate with your thumb.

Overall study results, summarized in Figure 5-1 and Table 5-2, show that a 4x4 vehicle with 33" diameter tires and an 11" tire patch width could meet the VCI requirement of 22 at highway pressures and a GVW of 6,500 pounds or less. However, in order to obtain the desired objective VCI of 15, a 6x6 configuration with CTIS would be required. Based on this fact, it was decided to retain a 6x6 configuration as a backup approach to the 4x4 baseline. The VCI calculations are believed to be conservative since our calculations showed the Surrograte Fast Attack Vehicle to have a VCI of 23.6 versus an empirical value of 22 as reported in "Results of Mobility Tests on the Surrogate Fast Attack Vehicle" (Technical Report GL-84-9, Table 1, R. Gillespie, September 1984).







- Calculations are conservative
- Latest Propulsion System Technology May Improve VCI
- 4x4 can meet VCI at 7250 lb GVW with 11.5" wide tires
- 4x4 can meet VCI at 8000 lb GVW with 12" wide tires

Figure 5-1. VCI vs. Tire Deflection Ratio

Table 5-2. Tracked vs. Wheeled VCI Comparison

4x4 vs 6x6 TRADES										
Drive	Tire Dia x Width (16.5" Wheel Dia)	GVW (lb)	VCI at Highway Pressures (15% deflection)	Highway Pressure (psi)	VCI with CTI at low pressure	CTI Low Pressure (psi)	Comments			
4x4	31" x 11"	8000	.23.6	32	18,5	20.1	Difficulty meeting step climb & ground clearance			
4x4	33" x 11"	8000	23.6	29	18.5	18.4	Tire diameter is not in the VCI equation and only affects pressure			
4x4	33" x 12"	8000	22.0	26.5	17.3	16.8	Meets VCI Spec			
4x4	33" x 11.5"	7250	22.0	26.2	17.2	16.6	Meets VCI Spec			
4x4	33" x 11"	6500	21.9	23.5	17.1	15.0	Meets VCI Spec			
6x6	31" x 11"	8000	20.5	21.2	16.0	13.4	Meets VCI Spec			
6x6	33" x 10"	8000	21.8	21.2	17.0	13.5	Meets VCI Spec			
6x6	33" x 12.5"	8000	19.1	17.0	15.0	10.8	Meets VCI spec and VCI goal at low pressures (possibly achieved with CTI). Tire width is difficult to package			
6x6	33" x 9.5"	7250	21.8	20.3	17.0	12.8	Meets VCI Spec			
6x6	33" x 9.0"	6500	21.8	19.2	17.0	12.2	Meets VCI Spec			
6x6	33" x 11"	6500	19.3	15.7	15.1	10.0	Can meet the goal with CTI.			

#### Notes

- 1. VCI Spec is 22 with a goal of 15.
- 4x4 configurations can meet the 22 spec limit for a GVW range of 6500 8000 lb at reasonable tire widths. VCI can be improved with CTI but will still not reach the desired goal of 15 (meets 17).
- 3. 6x6 configurations can meet the 22 spec limit for a GVW range of 6500 8000 lb at reasonable tire widths and can meet the goal of 15 if supplemented by CT1.

It is planned to confirm the 4x4 selection by applying NRMM as well as to reassess the selection for a hybrid electric drive configuration after hybrid electric drive components are further defined. The first NRMM analysis performed by LMDS has been to consider a modification to an existing and well characterized vehicle (narrowing a HMMWV to fit the V-22 width restrictions). This was accomplished by using an input file for the M1025 Armament Carrier HMMWV (A WES Standard) and running it over a Lauterbach terrain map and comparing these results to an identical simulation, but with the HMMWV's width reduced by two feet to fit the V-22. The results are shown in Appendix L. Note that both vehicles performed similarly in terms of speed under every condition except maneuvering around vegetation & obstacles, sliding on curves and tipping on curves. The narrower vehicle was able to achieve higher speeds around vegetation and obstacles because of its smaller size. However, it performed worse on curves limiting the speed to less than 5 mph and also being limited due to tipping which was never a problem with the baseline HMMWV configuration. These results were expected and are the main reason for either a retractable suspension concept or active suspension to counteract the effects of the narrow track width resulting from the V-22 transportability requirement.

The second bar chart in Appendix L is for the same comparison except it shows terrain "NoGo". As expected, the narrow vehicle has more "NoGo" than the baseline configuration (0.5% vs zero) due to tipping on side slope but also performs better in obstacle clearance interference due to its smaller size.

A concept layout of the RST vehicle is shown in Figure 5-2. It accommodates a three person crew consisting of a driver, commander (front passenger) and an operator in the back situated between the rear wheels. There is plenty of room on either side of the operator (above the wheel wells) for sensors and other necessary reconnaissance equipment and combat gear.

A remote controlled stabilized gun mount can be fitted to the roof, as shown in Figure 5-3 (once the vehicle has been driven out of the V-22) for use with either the MK19 grenade launcher or the M2 HB 50 caliber machine gun as a primary weapon. M240G machine guns could also be mounted to both door posts on 240° articulated swing-arm mounts. Figure 5-3 is also a cross-section of the vehicle layout which shows the occupant leg room and head room. Figure 5-4 is a perpendicular cross-section which shows the shoulder room.

The exterior of the vehicle can be protected by a combination of integral armor which provides floor and wheel well protection from anti-personnel mines & smaller anti-tank mines and door panel armor for personnel protection from small arms fire. Appliqué armor can also be supplied to supplement the integral armor to provide protection from 7.62 mm NATO ball or 7.62 mm AP at 0° obliquity. In addition to armor, the vehicle exterior can be treated with anti-reflective coatings and chameleon-like paints to reduce its visual signature. A skirt could also be added to minimize the visual signature associated with road and trail dust and low road noise tires can be used to minimize the acoustical signature.

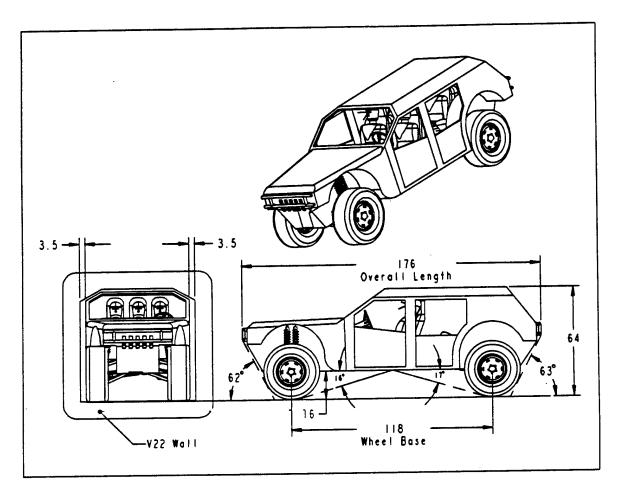


Figure 5-2. RST Vehicle Concept Layout

For maximum grade climbing, the ICE is located in front to shift the CG as far forward as possible. The ICE fits between the front A-arm suspensions as shown in Appendix M. The rear suspension uses progressive independent trailing arms. Due to the narrow vehicle width, and the desired amount of wheel travel, the turning circle can become a difficult requirement to meet. Our experience with conventional steering geometries indicates that a 28° wheel angle is about the maximum that can be accommodated with off the shelf CV joints and axle half shafts. A wheel base of 118" (comparable to the JTEV concept), and a track width of 61", can meet a turning circle radius of 24 feet and keep within the 28° recommended limit. The effects of wheel base and track width on front engine mounting clearance are shown in Figure 5-5 for a 25 foot turning circle.

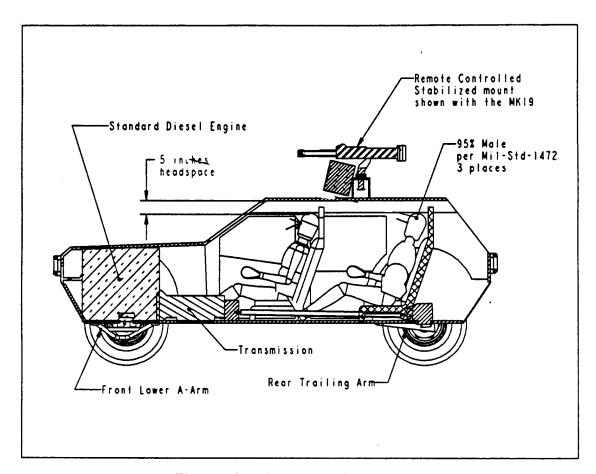


Figure 5-3. RST Vehicle Cross Section

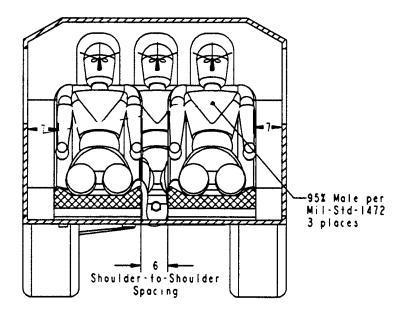


Figure 5-4. RST Vehicle Perpendicular Cross Section

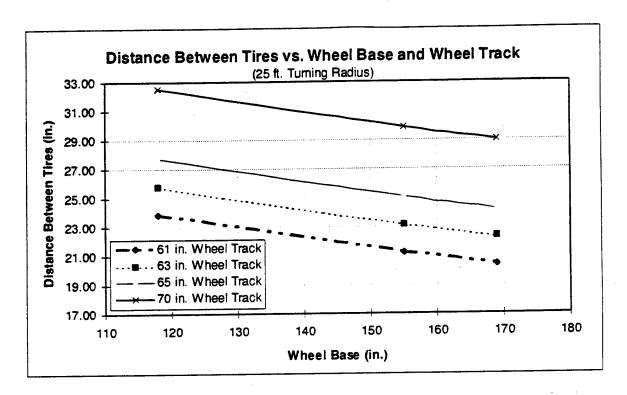


Figure 5-5. Distance Between Tires vs. Wheel Base and Wheel Track

Other configurations which would also fit the V-22 internal cargo dimensions are as follows:

Wheel Base (in)	Ground Clearance (in)	Max Wheel Angle for 25 ft turn (degrees)	
118" (JTEV)	16"	26	
129"	10" stowed (16" ride)	28	
155"	16"	32	
169"	18"	35	

Once the wheel base exceeds 155", the ground clearance must be increased in order to meet ramp over. However, this causes other problems with occupant head room. An optimum configuration appears to be between 118" and 155" wheel base. In order to fit a four person crew, the rear leg room requires a wheel base of at least 142". If the crew is limited to 3 persons, a wheel base of 118" is sufficient. LMDS has a proprietary retractable suspension geometry which allows for a 129" wheel base vehicle with a stowed ground clearance of 10". Once the vehicle exits the V-22 the track width is extended and the ground clearance increases to 16" (see Appendix N). This not only meets ramp over, turning circle, and ground clearance, but it also increases the track width to enhance side slope and cornering stability. Figure 5-6 depicts the concept vehicle on a 40% side slope with a 61" track width.

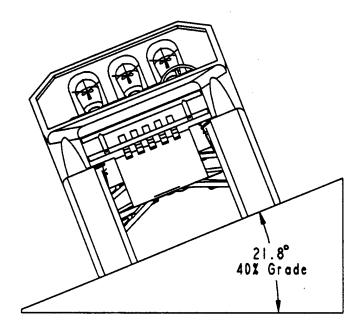


Figure 5-6. RST Vehicle Side Slope Perspective

Statically this concept can survive a 100% grade if its CG is less than 25" off the ground as shown in Figure 5-7. This situation is improved to about 112% if the track width is increased by a retractable mechanism to about 62 inches (measured from tire patch center to tire patch center).

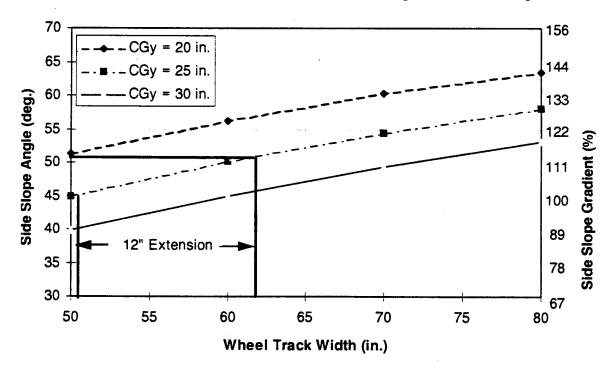


Figure 5-7. Point of Incipient Roll

Track width will also help cornering stability as shown in Figure 5-8. This figure illustrates the effects of CG height on holding a 0.6g lateral acceleration as a function of track width. Note that a vehicle with a 61" track width would require its CG to be less than 25.5" while a 74" wide vehicle would perform similarly with the CG as high as 30.5".

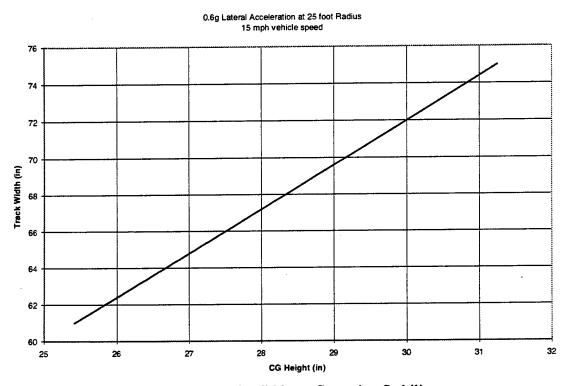


Figure 5-8. Track Width vs. Cornering Stability

A color picture of the concept vehicle with a conventional propulsion system is shown in Figure 5-9 with a translucent chassis so that some of the internal details can be seen. Appendix O contains additional vehicle perspectives including seating capacity, weapons mounted, and a litter variant.

Figures 5-10, 5-11, and 5-12 show initial concept vehicle perspectives that incorporate hybrid electric drive components. These perspectives are preliminary and will be updated based upon on-going hybrid electric drive component tradeoffs.

The hybrid electric drive system differs from the conventional propulsion system approach in that the driver is located between the front tires with the crew seated side-by-side and behind. This potentially improves driver vision and increases the storage space in the back of the vehicle for cargo storage.

The ICE is a combination turbine alternator. It weighs only 175 pounds and measures about 12" x 20" x 30". The batteries are polymer-lithium sheets and are placed along the floor and wheel wells to both lower the vehicle's CG and to augment the vehicles integral armor protection. The locations and thickness of the battery sheets are shown in Figures 5-11 and 5-12. The motors are each about 45 pounds, 0.5 ft<sup>3</sup> volume and shown mounted on top of the spring/damper assemblies. A trade-off is underway to determine if the best motor location is as shown or at the hub assemblies. The hybrid electric vehicle configuration shown has the advantage of having

components being somewhat protected from the road debris by being mounted in the crew areas and might lower the unsprung weight since the motors are attached to the vehicle chassis. In this configuration, the rotary motion from the motors is transmitted through a vertically oriented shaft into a right angle gearbox with a bevel gear set at each wheel to change the rotation from the vertical orientation to horizontally located axles. The motor shaft will have to have telescoping capabilities (about 16") to accommodate the range of wheel travel needed to traverse rough terrain.

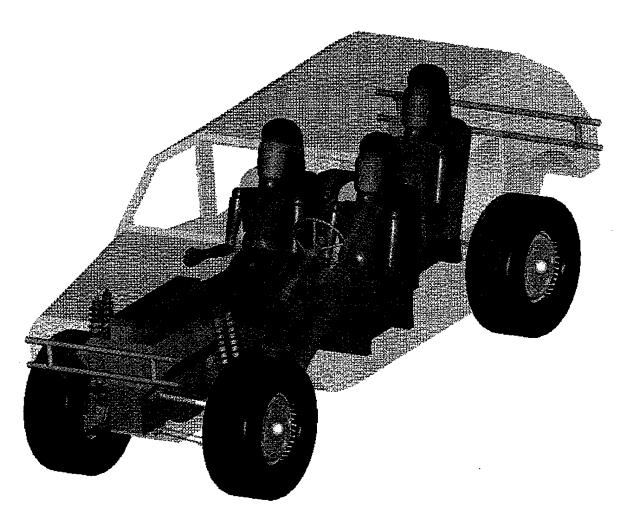


Figure 5-9. RST Concept Vehicle with Standard Propulsion System

The shock tower assemblies include a coil spring, also mounted to the chassis, and located concentrically around the motors to isolate the sprung mass and keep the tires in contact with the terrain as much as possible. Dampers, or shock absorbers, are not shown but could either be located concentric to the motor shaft or multiple absorbers could be nested circumferentially around the motors shaft. Steering is envisioned to be accomplished by rotating the bevel gearbox/axle assembly about the vertical centerline of the shock tower assembly.

around the motors shaft. Steering is envisioned to be accomplished by rotating the bevel gearbox/axle assembly about the vertical centerline of the shock tower assembly.

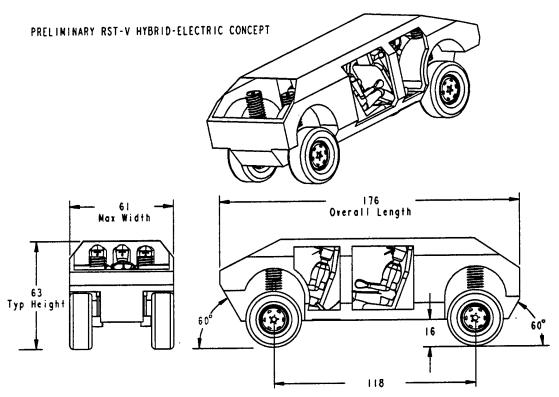


Figure 5-10. RST Hybrid Electric Drive Vehicle Concept Layout

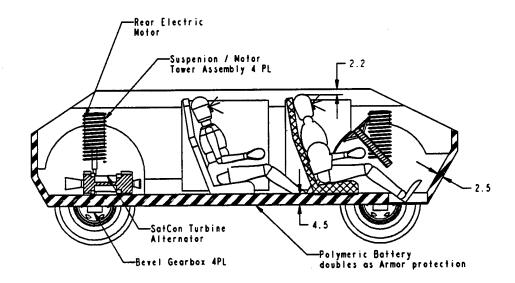


Figure 5-11. Hybrid Electric Drive System - Longitudinal Cross Section

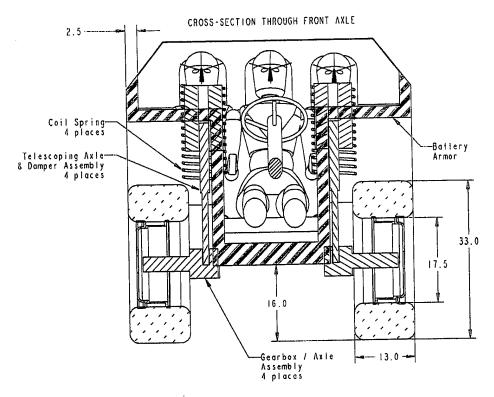


Figure 5-12. Hybrid Electric Drive System - Perpendicular Cross Section

A color picture of the concept vehicle with a conventional propulsion system is shown in Figure 5-13 with a translucent chassis so that some of the internal details can be seen.

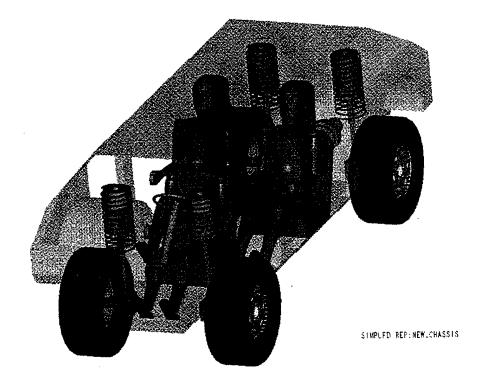


Figure 5-13. RST Concept Vehicle with Hybrid Electric Drive System

### 6. Vehicle Weight Analysis

#### 6.1 Overview

The overall objective of this initial Vehicle Weight Analysis was to compile a detailed list of vehicle component weights and develop a weight allocation profile for the four different vehicle mission variants and three different weight levels. From this profile, the optimum use of the weight allocations and the positioning of the weight can be determined.

### 6.2 Approach

The initial step in the weight analysis was the definition of the terminology for the general weight categories and the specific vehicle components. Then, a spreadsheet was set up in Microsoft Excel to perform the evaluation. The four different mission variants were evaluated separately. Information on individual component weights was gathered and entered into the spreadsheet. A weight analysis was performed with the weight threshold of 6500 pounds as an objective. This threshold was chosen based on engineering judgment of the minimum achievable weight with the desired mobility characteristics. Additional analyses are to be conducted for thresholds of 8000 pounds, the RST-V weight limit, and an intermediate weight of 7250 pounds.

Figure 6-1 displays the general categories and some of the specific components in each category. The Gross Vehicle Weight was separated into two general categories: (1) the Curb Weight, i.e., the basic vehicle weight independent of the mission; and (2) the Payload Weight specific to each mission. The Curb Weight was further broken down into three subcategories: Empty Vehicle, On-Vehicle Basic Equipment, and On-Vehicle Basic Consumables. The Empty Vehicle weight is composed of the bare essentials for a rolling, powered chassis including the frame, body, engine, drivetrain, and the essential auxiliary components like the battery. Also included is built-in armor which is envisioned to be an integral part of the chassis and represents the minimum armor suite common to all mission variants. The On-Vehicle Basic Equipment consists of essential Communications/Navigation equipment, Controls & Displays, and Accessories such as a fire extinguisher. On-Vehicle Basic Consumables includes essential fluids such as a full tank of fuel, engine oil, and coolant fluid in the engine, but does not include spare fluids.

The Payload Weight was broken down into five subcategories: Vehicle-Mounted Weapons & Ammunition, Personnel & Personal Equipment, Consumables, Configurable Armor, and Other Mission Equipment. Vehicle-Mounted Weapons & Ammunition encompasses several weapons options depending on the mission. Personnel & Personal Equipment covers the crew and passengers and their personal equipment such as helmets and armored vests. Consumables includes food rations, cooking stoves, and the water supply. Configurable Armor is armor in addition to the built-in armor that can be added to the vehicle in a mission-dependent configuration. Finally, Other Mission Equipment encompasses Communications/Navigation and Surveillance & Observation equipment that are very much mission-specific, and additional logistics equipment and/or litters for the "Log & Litters" mission variant.

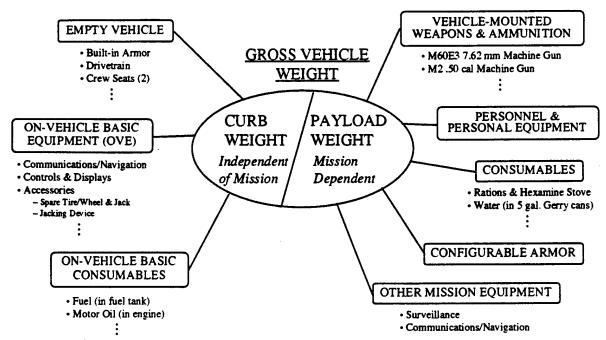


Figure 6-1. Suggested Vehicle Weight Categories

### 6.3 Analysis and Discussion

A summary of the Vehicle Weight Analysis for a minimum weight vehicle is given in Table 6-1. Details of the analysis are provided in Appendix P. This paragraph gives details of the analysis performed to date along with further discussion of several categories that require clarification. The weight estimates included will be further evaluated during the design and integration phase of the contract to develop appropriate baseline weight estimates for each mission variant and each weight threshold.

The Curb Weight is exactly the same for all four missions since the Curb Weight is independent of the mission. The engine and drivetrain weights are based on a standard diesel propulsion system. However, placeholders have been included in the spreadsheet for Hybrid Electric Drive components which are currently being evaluated. Built-in armor weighing 1047 pounds is also incorporated in the Empty Vehicle Weight since this armor may be an integral part of the chassis. This built-in armor includes underbody armor which meets NIJ level IV requirements, along with lighter weight armor around the passenger compartment which meets NIJ level IIIA requirements. Table 6-2 describes the armor selections for each mission variant.

Since the elements of *Payload Weight* were selected for each mission, the estimated weights for the missions vary from one another. The armor selections for each mission are a function of the perceived likelihood of need for armor in each mission. The RST mission required some Vehicle Mounted Weapons and Ammunition, and some additional consumables because of the time duration of the mission, but was not allocated additional armoring, because the theme of the mission is to avoid being detected. The resulting *Gross Vehicle Weight* was less than our objective weight. In contrast, the Strike mission, which requires even more Vehicle Mounted Weapons and Ammunition, is assessed to also require substantial additional armor because of the likelihood of being shot at, thereby driving the *Gross Vehicle Weight* well beyond our objective.

Table 6-1. Vehicle Weight Summary Example

		RST	Strike		Personnel		Log & Litters	
САТЕGORY/ІТЕМ	Qty	Weight (Lbs)	Qty	Weight (Lbs)	Qty	Weight (Lbs)	Qty	Weight (Lbs)
EMPTY VEHICLE								
Hull		722.0		722.0		722.0		722.0
Suspension & Steering		737.0		737.0		737.0		737.0
Engine (Standard Propulsion System)		865.2		865.2		865.2		865.2
Drivetrain (Standard Propulsion System)		464.0		464.0		464.0		464.0
Hybrid Electric Drive System		N/A		N/A		N/A		N/A
Built-in Armor (Mine Attack)		1047.0		1047.0		1047.0		1047.0
Electrical Systems		178.0		178.0		178.0		178.0
ON-VEHICLE BASIC EQUIPMENT								
Communications/Navigation		105.9		105.9		105.9		105.9
Controls & Displays		18.0		18.0	<u> </u>	18.0		18.0
Accessories		220.2		220.2		220.2		220.2
ON-VEHICLE BASIC CONSUMABLES (Fuel, Oil, etc.)		197.0		197.0		197.0		197.0
CURB WEIGHT		4554.3		4554.3		4554.3		4554.3
VEHICLE-MOUNTED WEAPONS & AMMUNITION	<u> </u>		ļ		<b> </b>	04.0		86.8
M60E3 7.62mm Machine Gun		94.8	ļ	118.8		94.8	400	35.0
M60E3 Ammunition (# of rounds)	800	70.0	2000	175.0	800	70.0	400	33.0
M2 .50 Cal Machine Gun		213.2		261.2	ļ			-
M2 Ammunition (# of rounds)	400	116.0	1000	290.0				
PERSONNEL & PERSONAL EQUIPMENT						075	3	525
Crew & Passengers	3	525	3	525	5	875	2	210.6
Personal Equipment	3	317.0	3	317.0	5	592.6	-	210.6
CONSUMABLES (Rations, Stove, Water, etc.)		160.5		85.5		92.5		121.5
CONFIGURABLE ARMOR (Small Arms/Frag Attack)		0		882.0		882.0		0
OTHER MISSION EQUIPMENT								550
Communication/Navigation		127.8		127.8	<b></b>	29.8	<b> </b>	76.8
Surveillance & Observation Equipment		53.9		44.0	<b></b>	39.6	ļ	39.6
Logistics Equipment & Litters		0.0		0.0		0.0		900.0
PAYLOAD WEIGHT	-	1678.2		2826.3		2676.3		1995.3
GROSS VEHICLE WEIGHT		6232.5		7380.6		7230.6		6549.6

The Personnel mission was assessed to require additional armoring, because we anticipate being required to transport personnel into dangerous areas. This weight, along with the additional weight of the personnel being transported, also drives the *Gross Vehicle Weight* for this mission beyond our objective. For the Litters and Logistics mission, we chose not to add the additional armoring, because this mission is more likely to be executed behind friendly forces, althought some cases may exist where additional armoring is desireable for cargos which require additional protection. The *Gross Vehicle Weight* for this mission was very close to our objective weight.

Table 6-2. Vehicle Armor Calculations

### INTEGRAL ARMOR (NIJ IV-underbody, NIJ IIIA-body):

	Area (ft <sup>2</sup> )	Thickness (in)	Weight (lbs)	Applies to Variant*:
Front wheel wells (2)	22.9	1	125.95	R,S,P,L
Firewall	11.9	1	65.45	R,S,P.L
Engine pan	7.3	1	40.15	R,S,P,L
Floor pan (incl. tunnel)	40	1	220	R,S,P,L
Rocker panels (2)	5.8	1	31.9	R,S,P,L
Partial rear wheel wells (2)	18	1	99	R,S,P,L
Rear bulkhead	15.5	1	85.25	R,S,P,L
Partial front 1/4 panel (2)	6	0.25	7.5	R,S,P,L
A post (2)	3	0.25	3.75	R,S,P,L
Front door (2)	13	0.25	16.25	R,S,P,L
B post (2)	3.5	0.25	4.375	R,S,P.L
Rear door (2)	12.5	0.25	15.625	R.S.P.L
Rear 1/4 panel (2)	11.4	0.25	14.25	R,S,P,L
Rear hatch	9.2	0.25	11.5	R,S,P,L
Roof	39.6	0.25	49.5	R.S.P.L
Windshield	7.5	1.33	112.5	R,S,P,L
Front door glass	2.4	1.33	36	R,S,P,L
Rear door glass	2.4	1.33	36	R,S.P.L
Rear glass	4.8	1.33	72	R,S,P.L
		Sum =	1046.95	R,S,P,L

CONFIGURABLE ARMOR (NIJ IV when over integral armor):

Partial front 1/4 panel (2)	6	0.75	25.5	S,P
A post (2)	3	0.75	12.75	S,P
Front door (2)	13	0.75	55.25	S,P
B post (2)	3.5	0.75	14.875	S.P
Rear door (2)	12.5	0.75	53.125	S.P
Rear 1/4 panel (2)	11.4	0.75	48.45	S.P
Rear hatch	9.2	0.75	39.1	S,P
Roof	<sup>1</sup> 39.6	0.75	168.3	S,P
Windshield	7.5	1.7	120	S,P
Front door glass	2.4	1.7	38.4	S,P
Rear door glass	2.4	1.7	38.4	S.P
Rear glass	4.8	1.7	76.8	S.P
Front fascia	7.4	1	48.1	S.P
Hood	22	1	143	S.P
		Sum =	882.05	

\* Weight R (RST) Variant = 1046.95 lbs.
Weight S (Strike) Variant = 1929 lbs.
Weight P (Personnel) Variant = 1929 lbs.
Weight L (Litters/Logistics) Variant = 1046.95 lbs.

Armor is Spectra and Spectra/Boron carbide composite and polycarbonate/glass

Wheelbase is 118"

LMDS - Chenowth

RST-V Concept Study Concept/Requirements Report

## APPENDIX A

RST-V System/Segment Specification Comments

## RST-V System/Segment Specification Comments

Paragraph Number	Paragraph Title	Comment
1.	SCOPE	Scope could be expanded to indicate program objectives. It should be noted that this is a development spec, not an acquisition spec.
1.1	IDENTIFICATION	
1.2	SYSTEM OVERVIEW	
1.3	DOCUMENT OVERVIEW	
2.	APPLICABLE DOCUMENTS	A all annual Addition
2.1	GOVERNMENT DOCUMENTS	With the requirements for Arctic operations MIL-L-46167 Lubricating oil, ICE, Arctic will be required also. (Maybe 21267 for break-in & storage?)
2.2	NON-GOVERNMENT DOCUMENTS	
2.2.1	Precedence of Documents	
3.	SYSTEM REQUIREMENTS	
3.1	DEFINITION	Suggest changing to "RST-V, Weapons/Light Strike, Personnel,
3.1.1	Mission Order of Precedence	and Litters/Logistics". Suggest describing how these priorities should be used in driving vehicle design and how they relate to precedence in section 3.8.
3.1.2	Basetine Platform	Reword to "accommodate a variety of mission variants"
3.2	CHARACTERISTICS	
3.2.1	Performance Characteristics	
3.2.1.1	Weight	Implies vehicle must weigh 8000 pounds. Rewrite to allow for vehicles with curb weight less than 5000 pounds.
3.2.1.1.1	Gross Vehicle Weight (GVW) .	Clarify GVW is a maximum and that payload is a minimum. Don't constrain curb weight.  In Table 1, <i>curb weight</i> and <i>margin of uncertainty</i> should be eliminated as parameters to be specified. <i>Maximum GVW</i> and <i>payload</i> are sufficient. Also, we recommend elimination of separate specifications for weights for separate missions, since the specification applies to a single vehicle (eg, RST-V). We suggest using one basic vehicle, with differences in payload, if different missions are to be accomodated. We suggest using more standard nomenclature, as we defined in IPR 2. These notes also apply to sections 3.2.1.1.2 and 3.2.1.1.3.
3.2.1.1.2	Curb Weight	It would be nice to have an estimate of the weight of the OVE listed. Suggest moving 3.2.3.1 and table 6 to this weight section (3.2.1.1). It is not necessary to itemize payload items in this paragraph.
3.2.1.1.3	Payload	Several items required by the spec are not listed as payload making up GVW. Doors and deep water fording kits need to be accounted in payload somewhere.  The payload weight list is all ROM and needs further definition. Crewman weights are not the same as those used in the weight allocations at IPR 2, nor the same as those provided by Jeff Bradel.  The conversions from pounds to kilograms should be correct if they're included.

Paragraph Number	Paragraph Title	Comment
3.2.1.1.4	Margin of Uncertainty	The weight uncertainty should be dealt with in the design process, not as a specified value. The GVW is a spec value which the design should meet.
3.2.1.2	Mobility	Suggest adding a conditional statement of meeting these requirements with a vehicle at normal operating temperature unless otherwise specified. It is not practical to expect forward motion of a vehicle within 10 sec. after master power with a cold diesel engine and/or torque converter transmission.
3.2.1.2.1	Speed	There are no references to reverse speed capabilities. Should provide a minimum reverse speed requirement.
3.2.1.2.1.1	Forward Speed	60 (75) MPH. Demands a very high power-weight ratio.
3.2.1.2.1.2	Dash Speed	Dash speed is typically time between 2 speeds (i.e. 10 to 30 MPH). 70 (75)MPH for 10 (20) min. Demands a very high power-weight ratio.
3.2.1.2.1.3	Sustained Low Speed	
3.2.1.2.1.4	Acceleration	Acceleration times specified will result in large engines (on the order of 250 hp for 15 sec accel and 450 hp for the goal of 10 sec accel). These engines will be very heavy (about 1,000 lbs). Are these acceleration limits necessary or is max speed more important?  0-30 MPH, 6 (4) Sec - Demands a very high power-weight ratio.  0-60 MPH, 15 (10) Sec - Demands a very high power-weight ratio.
3.2.1.2.1.5	Forward Motion	See temperature related comments above (3.2.1.2).
3.2.1.2.2	Braking	
3.2.1.2.2.1	Service Brakes	Braking times are aggressive and may be difficult to meet. Can they be relaxed? What is the possibility of using ABS for improved performance?  Specify a time increment between the braking operations for the brake fade requirement.
3.2.1.2.2.2	Emergency Braking	
3.2.1.2.2.3	Parking Brake	
3.2.1.2.3	Tractive Effort	
3.2.1.2.3.1	Drawbar Pull	
3.2.1.2.3.2	Mobility Rating	Table 4 has a column for V80. Clarification is needed for the terrain type for this column, or a statement that it applies over all terrain.
3.2.1.2.3.3	Terrain's	
3.2.1.2.4	Obstacles	
3.2.1.2.4.1	Longitudinal Slopes	Typically shutdown of engine for not less than 1 minute and restart and continue up hill without roll-back is required.
3.2.1.2.4.2	Side Slopes	This will require a track width of about 60 inches, assuming a reasonable CG and roll center location, which will conflict with the overall width if it is reduced to 56 inches. This is in a static mode - the situation is worse dynamically or when gun firing conditions exist.
3.2.1.2.4.3	Vertical Step	
3.2.1.2.4.4	Fording	60" fording seems very unlikely, even with a fording kit. Should fording kit be added to OVE table 6.
3.2.1.2.5	Maneuverability	
3.2.1.2.5.1	Turning (Dynamic)	0.6 g acceleration would require a wide track for stability (in the region of 65 inches) and it is optimistic to hold the suspension roll below 5 degrees. Something may have to be relaxed if the vehicle width is reduced from the present 65 inches.
3.2.1.2.5.2	Turning (Static)	Increase the maximum turning radius from 25 feet to 30 feet in order to minimize the tire swept volume thereby enabling increased payload capacity and compliance with the breakover angle and acceleration (engine/motor fit between the front tires) requirements.
	!	( requirements.

Paragraph	B	0
Number	Paragraph Title	Comment
3.2.1.2.6	Interfaces	
3.2.1.2.6.1 3.2.1.2.6.2	Approach Angle Departure Angle	
3.2.1.2.6.3	Ground Clearance	
3.2.1.2.6.4	Break Over Angle	
3.2.1.2.7	Ride Quality	
3.2.1.2.7.1	Ride Limiting Speed	Why not reference MIL-STD-1472 whole body vibration
0.2	The Emang Speed	requirements (5.8.4.1).
3.2.1.2.7.2	Obstacles	
3.2.1.2.8	Range	
3.2.1.2.9	Common Components	
3.2.1.2.9.1	Power Train	
3.2.1.2.9.1.1	Engine	Are we sure we want to limit ourselves to a diesel engine? "Fuel efficient" should be quantified.
3.2.1.2.9.1.2	Transmission	
3.2.1.2.9.1.3	STE/ICE	
3.2.1.2.9.1.4	Seals	
3.2.1.2.9.2	Suspension and Steering	
3.2.1.2.9.2.1	Wheels	Single Rim - Good. Should specify BEAD LOCK.
3.2.1.2.9.2.2	Tires	
3.2.1.2.9.2.3	Central Tire Inflation System	
3.2.1.2.9.2.4	Suspension Lockouts	
3.2.1.2.9.2.5	Steering	Forces not to exceed 5th % female capabilities (permissible increase in force with power drive failure?)
3.2.1.2.9.3	Windshield Wiper and Washers	
3.2.1.2.9.4	Bumpers	
3.2.1.2.9.5	Heater and Defroster	
3.2.1.2.9.6	Rear View Mirrors	
3.2.1.2.9.7	Electrical	
3.2.1.2.9.7.1 3.2.1.2.9.7.2	Electrical System Alternator	Caution: This may be unrealistic to have 100 amps at idle
J.E. 1.2.3.1.2	Alternator	without over- speeding the alternator at high engine speeds (benchmark: BFV is 300 amps at elevated engine rpm (1600?)). A CV drive system may be required which is more costly, consider having a high idle capability to handle generating power requirements when stationary. May be able to offer more versatility by giving a power/noise level requirement keeping both options open to meet the power requirements with survivability requirements in mind.
3.2.1.2.9.7.3	Lighting	
3.2.1.2.9.7.4	Wiring	
3.2.1.2.9.7.5	Light Switch	
3.2.1.2.9.7.6	Ignition Switch	
3.2.1.2.9.8	Wheel Splash and Stone Throw Protection	
3.2.1.2.9.9	Secure Lighting	Should this be part of "Electrical System" above?
3.2.1.3	Survivability	
3.2.1.3.1	Acquisition Avoidance	
3.2.1.3.1.1	Signature Reduction	The wavelengths from 8-14 microns are typically only used in
3.2.1.3.1.1.1	Signature Spectral Bands	space applications with black backgrounds. Is there data which shows this to be useful in land based applications?
3.2.1.3.1.1.2	General Design Characteristics	
3.2.1.3.2	Visual Signature Reduction	
3.2.1.3.2.1	Luminance	
3.2.1.3.2.1.1	Surface Treatments	
	Vertical Height	
3.2.1.3.2.2		

Paragraph Number	Paragraph Title	Comment
3.2.1.3.2.3	Vehicle Silhouette	
3.2.1.3.2.4	Shape	
3.2.1.3.2.5	Glint	
3.2.1.3.2.6	Dust	
3.2.1.3.2.7	Smoke	
3.2.1.3.3	Near Infrared (IR) Signature	Is this criteria realistic for a 2010 vehicle? Gen 3 will most likely
	Performance	be standard equipment by then.
3.2.1.3.3.1	Near Infrared Luminance	
3.2.1.3.4	Thermal Signature	
004044	Performance Edge Effects	
3.2.1.3.4.1 3.2.1.3.4.2	Overall Vehicle	
3.2.1.3.4.3	Engine Top Deck	NA - Why NA? Wheels & hubs make plenty of heat.
3.2.1.3.4.4	Suspension	NA - Why NA? The tail pipe and 'plume' are discernible.
3.2.1.3.4.5	Exhaust Signature Reduction	NA - Willy NA: The tail pipe and pro-
3.2.1.3.5	Acoustic Signature Reduction	
3.2.1.3.5.1	Stationary Requirement	> 70 dB @ 50 ft - Too high a dB level.
3.2.1.3.5.2	Moving Requirements	- 70 GD C 00 N 100 High C D 10 C
3.2.1.3.5.3	Detection Range Hit Avoidance	NA - Why NA? One virtue of a high-performance vehicle with a
3.2.1.3.2	Hit Avoidance	high DASH SPEED is to depart rapidly.
		Paragraph number should be 3.2.1.3.6. Following paragraph
		numbers should also be corrected.
3.2.1.3.3	Penetration Avoidance	
3.2.1.3.3.1	Direct Fire Threats	7/62 @ 300 meters - Criteria seem unrealistic. At 300 meters
3.2.1.0.0.1	Direct file friedle	hits are difficult and the weak rounds cited would not do much
		damage. Consider 100 meters, with more modern ammunition.
		Requirements driving armor configurations should perhaps allow
		for mission specific armor to be added to a basic lightweight
		armor configuration.
3.2.1.3.3.2	Indirect Fire Threats	NA - Why NA? Operators, especially Army Recom, say that
		ranging artillery fires are one of the greatest threats to scouts
		and flank security.
3.2.1.3.3.3	Mine Threats	Protection consistent with weight/mobility Seems to allow the
		next two requirements to be relaxed if weight or mobility
		requirements can't be met.  This requirement wording does not constrain the design.
		Definition should be provided for the minimum requirement, such
		as probability of successful passage through a specified mine
		field without rendering the vehicle or onboard equipment
		inoperable. This would then allow a combination of mine
		detection and under-vehicle armor to meet the requirement.
3.2.1.3.3.3.1	Personnel Area	.50 (1.0) lbs TNT @ any wheel.
3.2.1.3.3.3.1	Fuel Tank(s)	.50 (1.0) lbs TNT @ any wheel.
3.2.1.3.3.3.2	Kill Avoidance	
3.2.1.3.4.1	Fire Prevention and	
J.Z. 1.U.7.1	Suppression	
3.2.1.3.4.1.1	System Design	
3.2.1.3.4.1.2	Design of Occupied Spaces	
3.2.1.3.4.1.3	Portable Fire Extinguisher	Should consider more than one fire extinguisher.
3.2.1.3.4.2	Nuclear, Biological, and	
0.2	Chemical (NBC)	
3.2.1.3.5	Common Components	
3.2.1.4	Firepower	
3.2.1.4.1	Target Acquisition	Too broad; change "all systems" to "systems". Need rough
J.Z. 1.4.1	, a got , rodoloido.	order of magnitude regarding volume, weight, power
		requirements of RST-V sensors.
3.2.1.4.2	Weapons	
3.2.1.4.2.1	Primary Weapon	Did not include STINGER (MANPAD), but did list several anti-
U.C. 1.7.6.1	, , and , troupon	armor weapons. SEALs carried STINGER in Desert Storm.

Paragraph		
Number	Paragraph Title	Comment
3.2.1.4.2.2	Secondary Weapon	M60 is 7.62mm; M240 is 5.56mm. Different ammo.
3.2.1.4.2.3	Individual Weapon	Reference table 3 to provide specifics.
3.2.1.4.3	Ammunition	M240 is 5.56, not 7.62
3.2.1.4.4	Common Components	
3.2.1.5	Command, Control,	Should these systems be capable of tapping off of vehicle power
	Communications,	when mounted within the vehicle, should they be able to
	Computers & Intelligence	recharge from the vehicle system?
3.2.1.5.1	Communication Equipment	Includes older models, not newer (SATCOM??)
		May want to carry amplifiers on vehicle that are not
		dismountable.
3.2.1.5.2	Internal Communication	VVS-1, VVS-2, and VVS-3 all compatible w/CVC
	System (ICS)	Only VVS-2 works with digital comms.
3.2.1.5.3	Navigation Equipment	GPS batteries should charge off vehicle power.
20151		Include a "Whiskey Compass" for backup.
3.2.1.5.4	Common Components	
3.2.2	System Capability Relationships	
3.2.3	External Interface	1
0.2.0	Requirements	
3.2.3.1	On Vehicle Equipment (OVE)	Incorporate table 6 with table 2.
	and the second s	What about deep fording kit, doors, windows etc., winch, arctic
		kit?
3.2.3.2	Towing	
3.2.3.2.1	Tow Eyes	Should specify capacity.
3.2.3.2.2	Tow Pintle	
3.2.3.2.3	Electrical Connector	How about capability of interfacing with electric brakes on trailer
		also?
3.2.3.3	Power Distribution	The vehicle cannot be expected to operate in stealth mode using batteries only, if the main batteries are disconnected.
3.2.3.3.1	Equipment Compatibility	
3.2.3.3.1.1	Steady-State Voltage	
3.2.3.3.2	Slave Receptacle	Is there any other NSN'd item that is usable?
3.2.3.4	Fuel System	Consider specific safety requirements in 3.3.6 (fire resistant foam, spillover requirements, expansion for air transport).
3.2.3.4.1	Fuel/POLs	MIL-L-46167 Arctic oil.
3.2.3.4.2	Fueling	
3.2.3.4.3	Fuel Drain	
3.2.4	Physical Characteristics	
3.2.4.1	Protective Coatings	
3.2.4.1.1	Painting	
3.2.4.1.2	Corrosion Control	
3.2.4.2	Dimensions	Width: May change to 60 inches.
	1	Height: 59.6 in. at the aircraft side wall, not 66 in.
		Length: 250 in. is unreasonable. V22 will also carry other items,
		including vehicle crew, on aircraft seats.
3.2.4.2.1	Operational Dimensions	
3.2.4.3	Federal Motor Vehicle Safety	Probably don't want an airbag because it might be triggered by
	Standards Compliance	rough terrain driving, such as climbing vertical steps.
		This one will take a little time/research to assure it is not picking up on passenger car type regulations which would inhibit
		functionality of the vehicle (incorporation of airbags, side impact
		protection, locks etc.)
3.2.5	System Quality Factors	p. J.
3.2.5.1	Reliability	
3.2.5.1.1	Mean Miles Between	
J.E.G. 1. 1	Operational Mission Failure	e)
3.2.5.2	Maintainability	
3.2.5.2.1	Mean-Time-To-Repair	5 hours at Intermediate level may be low dependent on tasks
	,	(engine teardown and repair) and if most easier tasks are pushed
		to the Org level

Paragraph Number	Paragraph Title	Comment
3.2.5.2.2	Maintenance Ratio	Does this include silent watch time? Engine idle?
3.2.5.2.3	Miles Between Preventative Maintenance	This spec should either specify which PM tasks apply or define operating environment i.e. "Under normal (non-hostile) environments".
3.2.5.2.4	Mean Time To Perform Preventive Maintenance	Objective of "8" seems to be an error.
3.2.5.3	Availability	
3.2.5.4	Additional Quality Factors	
3.2.6	Environmental Conditions	
3.2.6.1	Natural Environment	
3.2.6.1.1	Ambient Air Temperature	
3.2.6.1.1.1	Severe Cold Weather Operation	This can really drive many other requirements! Lots of impact if performance degradation is not permitted in cold etc.
3.2.6.1.1.2	Temperature Variation	
3.2.6.1.2	Solar Radiation	May need desert cammy net.
3.2.6.1.3	Humidity	
3.2.6.1.4	Moisture and Fungus	
3.2.6.1.5	Atmospheric Pressure	
3.2.6.1.6	Elevation	May wish to carry Oxygen bottles for USMC crew members at altitudes above 8000 ft.
3.2.6.1.7	Wind	Driving in 85 knot winds seems risky, especially if it is broadside and the vehicle width is reduced to 60 inches.
3.2.6.1.8	Other	Need clarification on "capable of operating" and the depth of "severe snow". This requirement is an important consideration i weighing wheels versus tracks or in having deployable tracks. Need weather kits & heaters.
3.2.6.2	Induced Environment	
3.2.6.2.1	Internal Acoustical Noise	
3.2.6.2.2	Loads	
3.2.6.2.2.1	Accidental Loads	
3.2.6.2.2.2	Transportation Loads	
3.2.6.2.2.3	Towing, Lifting and Transport Loads	Towing requirements not listed here? Should state towing for a minimum number of miles at a maximum speed without damage. Generally these types of requirements limit or prohibit preparation time prior to towing. (or in paragraph 3.2.3.2).
3.2.6.2.3	Shock	Permit tailoring of the spec?
3.2.6.2.4	Vibration	
3.2.6.2.5	Explosive Atmosphere	
3.2.6.3	Operating Profile	
3.2.6.4	Prepositioning	
3.2.7	Transportability	
3.2.7.1	Amphibious Shipping	
3.2.7.2	Maritime Prepositioning Ships	
3.2.7.3	Rail Transportation	
3.2.7.4	Highway Transportation	
3.2.7.5	Air Transportation	
3.2.7.5.1	Fixed Wing	
3.2.7.5.2	Rotary Wing	See 3.3.6.2 for more stringent requirement. Aircraft loadmaste
3.2.7.5.3	Crew Ingress/Egress	will require access to rear of cabin, for safety of flight.
3.2.7.5.4	Vehicle Ingress/Egress	
3.2.7.6	Air Drop	
3.2.8	Flexibility and Expansion	in the state of th
3.2.8.1	Modular Subsystems and Components	Applicable subsystems should be defined (Does this include power train components).
3.2.8.2	Growth	"Reasonable growth" should be quantified.
3.2.8.3	Special Kits	Storage and permissible degradation factors for these kits is no included throughout the spec.
3.2.8.3.1	Winch Kit	

	· · · · · · · · · · · · · · · · · · ·	<u>,</u>
Paragraph Number	Paragraph Title	Comment
3.2.8.3.2	Modular Armor Kit	Has spec been determined?
3.2.8.3.3	Arctic Kit	
3.2.8.3.4	Deep Water Fording Kit	The spec. originated for large trucks and may not be appropriate for RST-V.
3.2.8.3.5	Soft Top Kit	Will this be carried as aux equipment (stowed), or will it be standard/installed on the vehicle?
3.2.9	Portability	
3.2.9.1	Lifting Eyes and Tiedowns	
3.3	DESIGN AND CONSTRUCTION	
3.3.1	Materials	
3.3.1.1	Toxic Products and Formulations	
3.3.1.1.1	Toxic Fumes	
3.3.1.2	Metric System	
3.3.2	Electromagnetic Radiation	
3.3.2.1	Electromagnetic Interference/ Electromagnetic Compatibility	
3.3.3	Nameplates and Product Marking	
3.3.4	Workmanship	
3.3.5	Interchangeability	
3.3.6	Safety	If this is to include program issues then reference to MIL-STD- 882 and it's individual tasking should be called out as applicable
3.3.6.1	Impact	
3.3.6.1.1	Restraining Devices	
3.3.6.1.2	Padding	
3.3.6.1.3	Roll-Over Protection	
3.3.6.2	Emergency Egress	Are additional equipment's allowed to degrade this evacuation time, i.e. transparent armor sides etc. If so is there a requirement for evacuation times with these accessories in place?
3.3.6.2.1	Crew Evacuation	
3.3.7	Human Engineering	
3.3.7.1	Crew and Personnel	
3.3.7.1.1	Vision	
3.3.7.2	Controls, Instruments,	
	Displays, and Lights	
3.3.7.3	Stowage Accessibility	
3.3.7.4	Ingress/Egress Aids	
3.3.8	Nuclear Control	
3.3.9	System Security	You may wish to make this applicable for RST-V variants that carry high tech sensors.
3.3.10	Government Furnished Property Usage	
3.3.11	Computer Resource Reserve Capacity	
3.4	DOCUMENTATION	
3.5	LOGISTICS	
3.5.1	Maintenance Planning	
3.5.1.1	Two-Level Maintenance	
3.5.1.2	Partitioning (Functional Modularity)	
3.5.2	Support and Test Equipment	
3.5.2.1	Support Equipment	
3.5.2.2	Test Points	
	Tools	
3.5.2.3	1 10013	1

Paragraph		Comment
Number	Paragraph Title	Comment
3.5.3	Supply Support	
3.5.3.1	Supply System Requirements	
3.5.3.2	Source, Maintenance, and	·
	Recoverability (SMR) Codes	
3.5.3.3	Consumable and Bulk Items	
3.5.4	Packaging and Handling	
3.5.4.1	Preparation	
3.5.4.2	Packaging	
3.5.4.3	Special Equipment	
3.5.4.4	Preservation, Packaging, and Packing	
3.5.4.5	Marking for Shipment	
3.5.5	Facilities	
3.5.5.1	Impact on Existing Support Facilities	
3.5.5.1.1	First Echelon Maintenance Facilities	
3.5.5.1.2	Second and Third Echelon Maintenance Facilities	
3.5.5.1.3	Fourth and Fifth Echelon	
0.0.0.1.0	Maintenance Facilities	
3.6	PERSONNEL AND	
0.0	TRAINING	
3.6.1	Personnel	
3.6.2	Training	
3.6.2.1	Operator Training	
3.6.2.2	Maintenance Training	
3.7	CHARACTERISTICS OF	
0	SUBORDINATE ELEMENTS	
3.7.1	Mobility Subsystem	
3.7.2	Survivability Subsystem	
3.7.3	Firepower Subsystem	
3.7.4	C4! Subsystem	
3.8	PRECEDENCE	
3.8.1	Life Cycle Cost and Performance	
3.8.2	Life Cycle Cost (LCC) Elements	
3.8.2.1	Operation and Support (O&S)	
3.8.2.2	Average Unit Rollaway Cost (AURC)	Does this include weapons? Radios? Sensors? Or is this the price of the basic mobility platform?
3.9	QUALIFICATION	
3.10	STANDARD SAMPLE	
3.11	PREPRODUCTION SAMPLE	
0.11	, ALI ROSOTION SAMELE	
	QUALITY ASSURANCE	
4.	PROVISIONS	
11	RESPONSIBILITY FOR	
4.1	INSPECTION	
4.1.1	Inspection	
4.1.2	Analysis	
4.1.3	Demonstration	
4.1.4	Modeling and Simulation	
4.1.4.1	Modeling	
4.1.4.2	Simulation	
4.1.5	Testing	
4.2	SPECIAL TESTS AND EXAMINATIONS	
4.2.1	Object Level	

Paragraph		
Number	Paragraph Title	Comment
4.2.1.1	Component	
4.2.1.2	Subsystem	
4.2.1.3	System	
4.2.2	Model Description	
4.2.2.1	Virtual Model	
4.2.2.2	Physical Scale Model	
4.2.2.3	Physical Full Scale Model	
4.3	REQUIREMENTS CROSS REFERENCE	
4.4	PROCESS	
4.4.1	Duration	
4.4.2	Technique	
4.4.3	Derived Data	
4.4.4	Measured Data	
5.	PREPARATION FOR DELIVERY	
6.	NOTES	
6.1	INTENDED USE	
6.1.1	Missions	
6.1.1.1	Stability Operations	
6.1.1.2	Limited Objective Operations	
6.1.1.3	Conventional Combat Operations	Does USMC only want an amphibious V-22 capable vehicle?
6.1.2	Threat	
Missing	Missing	Driver field of view is currently unspecified. It probably should be specified.

### APPENDIX B

Critical Parameters Matrix Comments

Lockheed Martin Defense Systems 100 Plastics Avenue Pittsfield, MA 01201 October 15, 1996

Naval Surface Warfare Center CD MCPO Carderock Division Headquarters Code 2020 Bethesda, MD 20084-5000 Attn: Jeffrey A. Bradel

Subject:

LMDS-Chenowth Comments to Carderock Critical Parameters Matrix

Reference: RSTA-V Concept Study In-Process Review #1 - Action Item #3

Enclosures: Filled Out Critical Parameters Matrices

A number of individuals from our LMDS-Chenowth project team have reviewed, discussed, and filled out the critical parameters matrix (referred to as Survey) provided by Carderock at the RSTA-V Concept Study In-process Review #1. Participants included systems, safety, and field support engineers. A summary of our findings follows.

#### General comments:

The survey is valuable. Filling it out forces you to define your priorities. As soon as survey participants complete the form, they should be interviewed to ensure the rationale for their priorities is fully understood. This could be an excellent approach for identifying specific user and developer needs.

Rationale for weightings should be documented. Suggest adding a "Rationale" column to the survey to help ensure capturing participants reasons for assigning different weightings.

It may be appropriate to have separate survey forms. Each form would be tailored to a mission area of interest (RSTA-V, LSV, Personnel, Litters/logistics). This could provide for more accurate survey conclusions but would work better if the survey forms only included the detailed requirements that were most closely associated with the mission of interest. Also, this type of survey should only be done after top level vehicle and subsystem concepts have been defined for the mission area of interest.

Prior to the definition of mission specific vehicle and subsystem concepts, the present survey form, which is of a more general nature, is more appropriate. By knowing the background of survey participants, the general survey form provides for determining what is most important to particular users or particular developers. Also, by including parameters like Average Unit

Rollaway Cost (presently not included), presently assumed differences between user and developer communities could be verified.

### Top Level Filters:

A minimum set of mandatory requirements will help to focus developer activities and could reduce development costs (if other specification requirements are not too stringent or defined as goals (non-mandatory)).

If survey participants are asked to identify any missing mandatory requirements (present survey instructions don't specifically request this), a different perspective on priorities might emerge.

Candidate additional top level, mandatory requirements include Range, Forward Speed, Average Unit Rollaway Cost, and Maximum Speed at RMS Course (Reliability/Durability driver that should include duration/duty cycle requirements).

Consider eliminating some Transportability modes since the V-22 is a worst case for certain parameters.

### Weighting of the Four Functional Headings:

Survey participant's functional areas and weightings are shown in Attachment A. Weightings were established from three perspectives (LSV, RSTA-V, and General). As shown, not all participants provided all three perspectives.

### Weighting of the Individual Requirements:

Survey participant's functional areas and weightings are included in Enclosures one through six.

Survey Participant	Representing
Jim Biancolo	LMDS Land Combat Systems Engineering
Steve Brown	LMDS Land Combat Systems Engineering
Jim Gibbons	LMDS Safety Engineering
Ange Castellano	LMDS Field Support Engineering
Chris Johnson	Chenowth - Mission/Vehicle Capabilities Analysis

Present method of making the sum of the sub-values total the subcategory weight within each overall category is somewhat cumbersome. Participant is spending time dealing with math and not as much time may be focused on relative weightings. Suggest using rounded percentages as shown in marked-up sample weighting section (Attachment B). Enclosures four and five include surveys that apply this approach. If this approach is used, a spreadsheet could be setup to perform the math necessary to evaluate participant's percentage inputs.

Enclosure three has been marked up to identify apparent incorrect shadings of some cells.

It is unclear whether the MK 93 Dual Mount entry (Firepower - Survey page 1) is a note or an entry to be weighted.

Signature Reduction entry (Survivability - Survey page 2) appears to be redundant. Also, why not include Radio Frequency Signature Performance (since it is listed in Specification paragraph 3.2.1.3.1.1.1 - Signature Spectral Bands)?

Mean Time to Perform Preventative Maintenance (Criteria Descriptions - Survey page 3) appears to have an error since its objective value is less demanding (8 hours) than the stated requirement value (6 hours).

Consider replacing Availability (Supportability - Survey page 1) with Logistic Delay Time and Administrative Delay Time since the other elements of Availability covered by the already included Reliability and Maintainability parameters.

Hope this is of some help.

Jim Biancolo

LMDS Land Combat Systems Engineering

### Carderock Critical Parameter Matrix

## LMDS-Chenowth Evaluation

### Weight Values for General Headings

LSV

	Survey Participant							
Function	Jim Biancolo (1)	Steve Brown (1)	Joe Rubin (1)	Jim Gibbons (2)	Ange Castellano (3)	Chris Johnson (4)		
Lethality	.40	.35	.30	.40	.30	•		
Supportability	.20	.10	.20	.20	.20	•		
Survivability	.10	.25	.25	.10	.20			
Mobility	.30	.30	.25	.30	.30	•		

RSTA-V

			KOIA-V			
			Survey P	articipant		
Function	Jim Biancolo (1)	Steve Brown (1)	Joe Rubin (1)	Jim Gibbons (2)	Ange Castellano (3)	Chris Johnson (4)
Lethality	.10	.25	.20	.20	.15	•
Supportability	.20	.15	.20	.20	.25	•
Survivability	.40	.30	.30	.40	.25	•
Mobility	.30	.30	.30	.20	.35	

General

			General			
			Survey P	articipant		
Function	Jim Biancolo (1)	Steve Brown (1)	Joe Rubin (1)	Jim Gibbons (2)	Ange Castellano (3)	Chris Johnson (4)
Lethality	.20	.20	.25	.35	-	.14
Supportability	.30	.10	.20	.15	-	.08
Survivability	.20	.30	.25	.20	•	.18
Mobility	.30	.40	.30	.30	-	<b>.6</b> 0

- (1) LMDS Land Combat Systems Engineering
- (2) LMDS Safety Engineering
- (3) LMDS Field Support Engineering (User Perspective)
- (4) Chenowth Mission/Vehicle Capabilities Analysis

# PROCEDURES FOR COMPLETION OF CRITICAL PARAMETERS MATRIX

- 1. Examine the enclosed criteria evaluation sheet and criteria descriptions.
- 2. Determine and comment if the Top Level filters are mandatory requirements.
- 3. Based on your experience, assign a weight value to the four general functional headings of Lethality, Supportability, Survivability, Mobility. The sum of these four values should equal 1.00.
- 4. Next, assign weights to the individual requirements under each functional category (i.e. Lethality). The sum of these sub-values should total the overall category weight. See below for an example of this procedure.

#### CRITERIA FOR EVALUATION

### Sample Weighting Section

LETHALITY = .21	;	Weight	
	170%		
Firepower	0.140		
Primary Weapon (M-2 .50 cal / MK-19 40mm Grenade Launcher)		0.060	40
MK-93 Dual Mount for Primary Weapons			
Secondary Weapon ( M-60E3/M240)		0.040	30
Individual Weapon		0.010	10
Ammunition		0.030	20
Command, Control, Comm, Computers & Intel 30 % -	0.070		
Communication Equipment		0.010	15
Internal Communication System (ICS)		0.010	
Navigation Equipment		0.050	70
Tota	, 0.210	- <del>0-210</del>	A M
	100.70		

## **Enclosure 1**

# Jim Biancolo LMDS Land Combat Systems Engineering

LSV Evaluation

### CRITERIA FOR LSV EVALUATION

LSV

4 KOD

?

TOP LEVEL FILTER (Thresholds)	Yes	No
Weight		
Gross Vehicle Weight (GVW) 8,000 lb		
Payload 3,000 lb		
Dimensions		
Operational Dimensions (V-22 compatible) WAST CASE?	V	
Transportability		
Amphibious Shipping (LHA, LHD, LSD, LSD(CV), LPD)		
Maritime Prepositioning Ships (MPS)		
Rail Transportation		
Highway Transportation		
Fixed wing (C-130 and larger)		
Rotary Wing (V-22 and CH-53E internal, UH-60A, CH-47C	V	
CH-47D, CH-53D, V-22 external)		
* Crew Ingress/Egress		<del></del>
* Vehicle Ingress/Egress		·

LETHALITY = 64	We	ght
Firepower	125	
Primary Weapon (M-2 .50 cal / MK-19 40mm Grenade Launcher)		.08
MK-93 Dual Mount for Primary Weapons		0
Secondary Weapon ( M-60E3/M240)		.05
Individual Weapon		.04
Ammunition		. 08
Command, Control, Comm, Computers & Intel	13-	
Communication Equipment		.06
Internal Communication System (ICS)		,06
Navigation Equipment		.03
Tota	I • H	· u

SUPPORTABILITY = Z		Weight		
Reliability		• 1		
* Mean Miles Between Opn Mission Failure			· ·	
Maintainability		. 1		
* Mean-Time-To-Repair			•05	
* Maintenance Ratio			0	
* Miles Between Preventative Maintenance			.05	
* Mean Time To Perform Preventative Maintenance			0	
Availability		0		
	Total	.2	• 2	

\* FORWARD SPEETO

MAN SPEED AT RMS COURSE (WITH, DURATION & DUTY CYCLE)
- RELINGING DURANCING DRIVER

AUGRACE UNIT ROLLAWAY COST B-8

### CRITERIA FUR LOV EVALUATION

SURVIVABILITY = ,/	SURVIVABILITY = ,/		ight	
			***************************************	
Acquisition Avoidance		• 04.		l
Signature Reduction				一
Visual Signature Reduction			.013	
Near Infrared (IR) Signature Performance			.007	İ
Thermal Signature Performance			.007	l
Acoustic Signature Performance			.013	l
Hit Avoidance		102		ĺ
Penetration Avoidance		.02		
Direct Fire Threats			,01	l
Indirect Fire Threats			Ð	l
Mine Threats			.01	l
Kill Avoidance		200		L
Fire Prevention and Suppression		18/11		Γ,
Nuclear, Biological, and Chemical (NBC)		10////		
	Total	• 1	1 .1	ı

MOBILITY = , 3		Veight
Speed	.05	
Forward Speed		.02
Dash Speed		•02
Sustained Low Speed		0
Acceleration		.01
Forward Motion		0
Braking	.00	7
Service brakes		•02
Obstacles	. 04	<b>X</b>
Longitudinal Slopes		•01
Side slopes		.02
Vertical Step		0
Fording		•02
Maneuverability	.03	7
* Turning (Static)		
Interfaces	.01	1
Approach Angle		.01
Departure Angle		100
Ground Clearance		101
Break Over Angle		101
Ride Quality	.0:	35
Ride Limiting speed		,025
Obstacles		.010
Range	.05	
Suspension / Steering	• 6:	35
Central Tire Inflation System		.015
* Run Flat Capability		• 0ZV)
•	Total 3	. 3

### **CRITERIA DESCRIPTIONS**

<u>Crew Ingress/Egress:</u> The vehicle shall incorporate ingress/egress paths to the aircraft personnel/cargo compartment while the vehicle is loaded aboard a fixed-wing or rotary-wing aircraft (i.e. removable/stowable windshield, a folding windshield, or other means).

Vehicle Ingress/Egress: On landing, the vehicle shall be clear of the aircraft and achieve a "guns up" and ready to fire status within 15 seconds after the ramp is fully lowered. On take-off, the vehicle shall be loaded into the aircraft and achieve a "flight ready" status within 2 minutes after starting to ascend the ramp.

Mean Miles Between Operational Mission Failure: The Mean Miles Between Operational Mission Failure (MMBOMF) shall be 2000 mi (3219 km) or greater. It is an objective that the MMBOMF shall be 3000 mi (4828 km).

Mean-Time-To-Repair: The Mean-Time-To-Repair (MTTR) at the organizational level (1st and 2nd echelon) shall be 3.0 man-hours or less; at the intermediate level (3rd and 4th echelon) the MTTR shall be 5.0 man-hours or less. MTTR is the sum of corrective maintenance time at any specific level of repair, divided by the total number of failures repaired during the particular interval.

Maintenance Ratio: The Maintenance Ratio of the vehicle shall be .16 man-hours or less of maintenance (preventative and corrective) per hour of operation. The Maintenance ratio does not include PMCS.

Miles between Preventative Maintenance. The miles between Preventative Maintenance (MBPM) shall be no less than 3000 mi (4828 km) or 6 months. It is an objective that the MBPM shall be 4000 mi (6437 km) or 8 months.

Mean Time To Perform Preventative Maintenance. The Mean Time To Perform Preventative Maintenance (MTTPPM) shall be no greater than 6 hours. It is an objective that the MTTPPM shall be 8 hours.

<u>Turning (Static)</u>: The vehicle static turning radius shall not exceed 25ft (7.62m) curb to curb, and the vehicle shall complete the turn from a static position in less than 9.5 seconds. It is an objective that the vehicle turning radius shall not exceed 20 ft (6.1m) curb to curb, and that the vehicle complete the turn from a static position in less than 8 seconds.

Run Flat Capability: The tires shall have a threshold run flat capability of 30 miles at 30 mph (48 km at 48 kph) on a hard surface road, after loss of air pressure in any two tires.

## **Enclosure 2**

## Steve Brown LMDS Land Combat Systems Engineering

LSV Evaluation

TOP LEVEL FILTER (Thresholds)	Yes	Na
Weight	1 /	
Gross Vehicle Weight (GVW) 8,000 lb		
Payload 3,000 lb	1 –	
Dimensions		
Operational Dimensions (V-22 compatible)		
Transportability		
Amphibious Shipping (LHA, LHD, LSD, LSD(CV), LPD)		<b></b>
Maritime Prepositioning Ships (MPS)		<b> </b>
Rail Transportation	V	
Highway Transportation		
Fixed wing (C-130 and larger)		
Rotary Wing (V-22 and CH-53E internal, UH-60A, CH-47C	_ V	
CH-47D, CH-53D, V-22 external)	V	
* Crew Ingress/Egress	1/	
* Vehicle Ingress/Egress		
		<u> </u>

LETHALITY = ,35	Wei	ight
Firepower	1.25	
Primary Weapon (M-2 .50 cal / MK-19 40mm Grenade Launcher)		.10
MK-93 Dual Mount for Primary Weapons		11/11/11
Secondary Weapon ( M-60E3/M240)		,075
Individual Weapon		.02
Ammunition		.055
Command, Control, Comm, Computers & Intel	0.10	
Communication Equipment		.05
Internal Communication System (ICS)		.02
Navigation Equipment		-03
Tota	al .35	,35

SUPPORTABILITY = 0.1		Weight	
Reliability		. 05	
* Mean Miles Between Opn Mission Failure			.05
Maintainability		.03	
* Mean-Time-To-Repair			.01
* Maintenance Ratio			.005
* Miles Between Preventative Maintenance			.005
* Mean Time To Perform Preventative Maintenance			.01
Availability		.02	
	Total	,10	-10

Steven Brown 10/10/96 LSV SURVEY LSV

SURVIVABILITY = 0.25	We	Weight	
	104		
Acquisition Avoidance		WILL	
Signature Reduction		.01	
Visual Signature Reduction		.01	
Near Infrared (IR) Signature Performance		<i>X</i> -1	
Thermal Signature Performance		.01	
Acoustic Signature Performance		.01	
Hit Avoidance	1055	+ 05	
Penetration Avoidance	,075	.035	
Direct Fire Threats		.015	
Indirect Fire Threats		102	
Mine Threats	30		
Kill Avoidance	. OB	1009	
Fire Prevention and Suppression			
Nuclear, Biological, and Chemical (NBC)	Total 25	25	

MOBILITY = 0.3	W	Weight	
	0.07		
Speed	0.063	01	
Forward Speed		.025	
Dash Speed		.005	
Sustained Low Speed		,02	
Acceleration		,00	
Forward Motion	0.01		
Braking	0107	•015	
Service brakes	0.055	- 1013	
Obstacles	0,033	کھ ن	
Longitudinal Slopes		102	
Side slopes		.002	
Vertical Step		,007.	
Fording	0,015		
Maneuverability	(1,01)	-0/5	
* Turning (Static)	0.05		
Interfaces	0.03	-007	
Approach Angle		.007	
Departure Angle		.03	
Ground Clearance		.005	
Break Over Angle	0,04	7	
Ride Quality	070 ]	202	
Ride Limiting speed		101	
Obstacles	0.02		
Range	0,04		
Suspension / Steering	0,09	-02	
Central Tire Inflation System		4	
* Run Flat Capability	Total 0,30	130	

.03

Sturm Moseum 10/10/96 LSV Survey

## **Enclosure 3**

# Steve Brown LMDS Land Combat Systems Engineering

**General Evaluation** 

## GINERAL

TOP LEVEL FILTER (Thresholds)	Yes	No
Weight		
Gross Vehicle Weight (GVW) 8,000 lb		V
Payload 3,000 lb		
Dimensions		
Operational Dimensions (V-22 compatible)		
Transportability		
Amphibious Shipping (LHA, LHD, LSD, LSD(CV), LPD)		
Maritime Prepositioning Ships (MPS)		
Rail Transportation		
Highway Transportation		
Fixed wing (C-130 and larger)		
Rotary Wing (V-22 and CH-53E internal, UH-60A, CH-47C		
CH-47D, CH-53D, V-22 external)		
* Crew Ingress/Egress		
* Vehicle Ingress/Egress	V	

LETHALITY = 🔎		Weight	
Firepower	.15		
Primary Weapon (M-2 .50 cal / MK-19 40mm Grenade Launcher)		,05	
MK-93 Dual Mount for Primary Weapons		187711	
Secondary Weapon ( M-60E3/M240)		٠03	
Individual Weapon		.02	
Ammunition		.05	
Command, Control, Comm, Computers & Intel	.05		
Communication Equipment		.03	
Internal Communication System (ICS)		.01	
Navigation Equipment		.01	
Tota	.20	.20	

SUPPORTABILITY =			ght	
Reliability		.05		عمر
* Mean Miles Between Opn Mission Failure			<b>***</b>	}
Maintainability		<b>.</b> 93		
* Mean-Time-To-Repair			.01	
* Maintenance Ratio			,005	]
* Miles Between Preventative Maintenance			-005	1
* Mean Time To Perform Preventative Maintenance			-01	1
Availability		,02		1,0
	Total	.10	.10	

Steven Brown 10/3/96

## 

## GENERAL

SURVIVABILITY = 1.3		Weight		
	WUI	14/1/1/1		
Acquisition Avoidance	./2			
Signature Reduction		11/1/4	n	
Visual Signature Reduction		02		
Near Infrared (IR) Signature Performance		-04		
Thermal Signature Performance		,04		
Acoustic Signature Performance		,02		
Hit Avoidance	.08	3	೦೮	
Penetration Avoidance	.06			
Direct Fire Threats		.03		
Indirect Fire Threats		.01		
Mine Threats		,02		
Kill Avoidance	.04	+ 11/10/	W	
Fire Prevention and Suppression	MAS	720	20ء دام	
Nuclear, Biological, and Chemical (NBC)	VIAN	221	ۋامر 🌃	
	Total .3	0 .30	5	

MOBILITY = ,4	We	ight
	VIVIII	UM
Speed	30.	
Forward Speed		101
Dash Speed		.03
Sustained Low Speed		.005
Acceleration		-03
Forward Motion		-505
Braking	1.015	
Service brakes		.015
Obstacles	1.06	
Longitudinal Slopes		د0.
Side slopes		.02
Vertical Step		.0025
Fording		.0075
Maneuverability	,025	
* Turning (Static)		
Interfaces	.07	
Approach Angle		,02
Departure Angle		,01
Ground Clearance		.035
Break Over Angle		•005
Ride Quality	.055	
Ride Limiting speed		,3,0
Obstacles		.02
Range	.025	
Suspension / Steering	.07	
Central Tire Inflation System	•	.075
* Run Flat Capability		.025
	Total 4	14

Steven JBrown 10/3/96

## **Enclosure 4**

## Jim Gibbons LMDS Safety Engineering

LSV Evaluation

Done with recommended "Percentage Approach"

## LSV

TOP LEVEL FILTER (Thresholds)	Yes	Na
Weight		
Gross Vehicle Weight (GVW) 8,000 lb		
Payload 3,000 lb		
Dimensions		
Operational Dimensions (V-22 compatible)		
Transportability		
Amphibious Shipping (LHA, LHD, LSD, LSD(CV), LPD)		
Maritime Prepositioning Ships (MPS)	/	
Rail Transportation	/	
Highway Transportation		
Fixed wing (C-130 and larger)		
Rotary Wing (V-22 and CH-53E internal, UH-60A, CH-47C	/	
CH-47D, CH-53D, V-22 external)		
* Crew Ingress/Egress		
* Vehicle Ingress/Egress	1/	

LETHALITY = 40		Weight	
Firepower	60		
Primary Weapon (M-2 .50 cal / MK-19 40mm Grenade Launcher)		45	
MK-93 Dual Mount for Primary Weapons			
Secondary Weapon ( M-60E3/M240)		10	
Individual Weapon		25	
Ammunition		<u> చ</u> ేక	
Command, Control, Comm, Computers & Intel	40		
Communication Equipment		40	
Internal Communication System (ICS)		30	
Navigation Equipment		30	
Tota	100		

SUPPORTABILITY = 20	SUPPORTABILITY = 20		Weight	
Reliability		35		
* Mean Miles Between Opn Mission Failure				
Maintainability		15		
* Mean-Time-To-Repair			10	
* Maintenance Ratio			10	
* Miles Between Preventative Maintenance			10	
* Mean Time To Perform Preventative Maintenance			70	
Availability		50		
	Total	100		

JMD.

15V

SURVIVABILITY = 10	We	ght
	15	
Acquisition Avoidance		
Signature Reduction		15
Visual Signature Reduction		15
Near Infrared (IR) Signature Performance		- '3
Thermal Signature Performance		1.4
Acoustic Signature Performance		65
Hit Avoidance	25	
Penetration Avoidance	40	
Direct Fire Threats		35
Indirect Fire Threats		50
Mine Threats		15
Kill Avoidance	20	
Fire Prevention and Suppression	1000	
Fire Prevention and Suppression	シベンバ	<b>-</b>
Nuclear, Biological, and Chemical (NBC)	Total IDD	

MOBILITY = 30	Wei	ight
Speed	20	
Forward Speed		30
Dash Speed		20
Sustained Low Speed		10
Acceleration		30
Forward Motion		10
Braking	15	_
Service brakes		
Obstacles	15	35
Longitudinal Slopes		30
Side slopes		20
Vertical Step		20
Fording		20
Maneuverability	-10	
* Turning (Static)	- 5	
Interfaces	3	10
Approach Angle		10
Departure Angle		60
Ground Clearance		20
Break Over Angle	15	20
Ride Quality		50
Ride Limiting speed		50
Obstacles	10	3
Range	10	
Suspension / Steering		40
Central Tire Inflation System		60
* Run Flat Capability	Total 100	

Jmg

### **Enclosure 5**

### Ange Castellano Field Support Engineering

LSV Evaluation
Done with recommended "Percentage Approach"

### CRITERIA FUR LSV EVALUATION

Ange Castallono + Bub Repaireskei

	No
-	
V	

\* Debenpry, or

LETHALITY = 30	Wei	ght
	67	
Primary Weapon (M-2 .50 cal / MK-19 40mm Grenade Launcher)		30
MK-93 Dual Mount for Primary Weapons		10
Secondary Weapon ( M-60E3/M240)		20
Individual Weapon		10
Ammunition		30
Command, Control, Comm, Computers & Intel	33	
Communication Equipment		25
Internal Communication System (ICS)		40
Navigation Equipment		35
Tota	11	<u></u>

SUPPORTABILITY = 20		Wei	ght
Reliability		45	
* Mean Miles Between Opn Mission Failure			
Maintainability		45	50
* Mean-Time-To-Repair			50
* Maintenance Ratio  * Miles Between Preventative Maintenance			30
* Mean Time To Perform Preventative Maintenance			15
Availability		10	
Availability	Total	l	

Ao which were Logistics Delay Time + ADMIN Delay Time

But Execute Reliability +

Minimanability Firston

SINCE INCLUDED ASSONE

### CRITERIA FOR LSV EVALUATION

SURVIVABILITY = 20		Wei	ght
Acquisition Avoidance		50	
Signature Reduction			25
Visual Signature Reduction			20
Near Infrared (IR) Signature Performance			15
. Thermal Signature Performance			15
Acoustic Signature Performance			マミ
Hit Avoidance		25	
Penetration Avoidance		5	
Direct Fire Threats			36
Indirect Fire Threats			30
· Mine Threats			Gn
Kill Avoidance		20	
Fire Prevention and Suppression		50	
Nuclear, Biological, and Chemical (NBC)		50	
,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,	Total	1	<u> </u>

MOBILITY = 30	Wei	ght
Speed	15	1
Forward Speed		20
Dash Speed		30
Sustained Low Speed		10
Acceleration		30
Forward Motion (?)		10
Braking	10	
Service brakes		10
Obstacles	20	
Longitudinal Slopes		25
Side slopes		25
Vertical Step		25
Fording		25
Maneuverability	15	
* Turning (Static)		
Interfaces	スシ	
Approach Angle		25
Departure Angle		25
Ground Clearance		32
Break Over Angle		25
Ride Quality		
· Ride Limiting speed		60
Obstacles		40
Range	4	
Suspension / Steering	10	40
Central Tire Inflation System		
* Pun Flat Canability		60
Total	aı	<u> </u>

### **Enclosure 6**

### Chris Johnson Chenowth - Mission/Vehicle Capabilities Analysis

**General Evaluation** 

4 October 1996

### CRITICAL PARAMETERS MATRIX NSWC Carderock RSTA-V Study CW Johnson Response

Page 1

NSWC Carderock generated to assign weights to general functions as well as individual capabilities and requirements for the RSTA-V vehicle.

The matrix will be used as a questionaire to USMC, SOCOM and other prospective users. Also, the contractors (LMDS-CC, GDLS and RMM) will fill it out for comparison sake.

CATEGORY		CM JOI	enson
TOP LEVEL FILTER (Thresholds)		YES	NO
Weight Gross Vehicle Weight (GVW) 8000 lb Payload 3000 lb Dimensions		X	
Operational Dimensions (V-22 Compatible)		X	
Transportability Amphibious Shipping (LHA, LHD, LSD, LDS(CV) Maritime Prepositioning Ships (MPS) Rail Transportation Highway Transportation Fixed Wing (C-130 & larger) Rotary Wing (V-22 and CH53E internal, UH-60 CH-47C, CH-47D CH=53D, V-22 external) * Crew Ingress / Egress * Vehicle Ingress / Egress		X X X X X X X	
LETHALITY = 0.14		W	EIGET
Firepower  Primary Weapon (M2 .50 Cal / MK-19 40mm)  . MK-93 Dual Mount for Primary Weapons Secondary Weapon (M60E3/M240G) Individual Weapon Ammunition Command, Control, Comm, Computers & Intel Communications Equipment Internal Communication System (ICS) Navigation Equipment		0.100	0.040 0.005 0.030 0.005 0.020 0.020 0.005 0.015
T	otal	0.140	0.140
SUPPORTABILITY = 0.08		WE:	IGHT
Reliability  * Mean Miles Between Opn Mission Failure  Maintainebility  * Mean-Time-To-Repair  * Maintenance Ratio  * Miles Between Preventive Maintenance  * Mean Time to Perform Preventive Maintenan  Availability	ce	0.010	0.010 0.010 0.010 0.020 0.020 0.020
T		0.080	

P.5.06

Page 2

DC1-U1-95 NUN 15:55

4 October 1996

### CRITICAL PARAMETERS MATRIX NSWC Carderock RSTA-V Study CW Johnson Response

WEIGHT SURVIVABILITY = 0.18 0.085 Acquisition Avoidance 0.020 Signatue Reduction 0.015 Visual Signature Reduction 0.015 Near Infrared (IR) Signature Performance 0.015 Thermal Signature Performance 0.020 Acoustic Signature Performance 0.020 0.020 Hit Avoidance 0.050 Penetration Avoidance 0.020 Direct Fire Threats 0.015 Indirect Fire Threats 0.015 Mine Threats 0.025 Kill Avoidance 0.020 Fire Prevention and Suppression 0.005 Nuclear, Biological and Chemical (NBC) 0.180 0.180 Total WEIGHT MOBILITY = 0.60 0.215 Speed. 0.040 Forward Speed 0.080 Dash Speed 0.010 Sustained Low Speed 0.080 Acceleration 0.005 Forward Motion 0.020 Braking 0.020 Service Brakes 0.105 Obstacles 0.025 Longitudinal Slopes 0.040 Slide Slopes 0.020 Vertical Step 0.020 Fording 0.020 Maneuverability 0.020 \* Turning (Static) 0.110 Interfaces 0.030 Approach Angle 0.030 Departure Angle 0.030 Ground Clearance 0.020 Break Over Angle 0.070 Ride Quality 0.040 Ride Limiting Speed 0.030 Obstacles 0.020 0.020 Range 0.040 Suspension / Steering 0.020 Central Tire Inflation System 0.020 \* Run Flat Capability 0.600 0.600 Total

APPENDIX C

**User Topics** 

### Questions for a RST-V Vehicle User

### Requirements

- 1) Which vehicle capabilities are most important?
- 2) Which vehicle requirements would give the most military advantage if exceeded?

### **Mission Profiles**

- 3) What are typical mission profiles for stealth mode?
  - a) terrain
  - b) duration of stops, duration of moves
  - c) speed
  - d) auxiliary power required
  - e) longest time between battery charges
- 4) What are typical mission profiles for dash mode?
  - a) terrain
  - b) duration
  - c) speed
  - d) auxiliary power required
- 5) What situations drive the turning radius requirement? What is the impact of a smaller or larger turning radius on the ability to conduct operations?
- 6) What situations drive the turning (dynamic) requirement? What is the impact of a smaller or larger turning (dynamic) capability on the ability to conduct operations?
- 7) When is central tire inflation of value? What types of terrain? What transition times are appropriate to go from one inflation level to another?
- 8) What is the required crew size for typical RST-V missions? Unique missions?

### **Mobility**

- 9) What are the most important mobility capabilities? For each mission type?
- 10) What kinds of terrains are presently considered un-crossable? Which of these terrains would give the most benefit if crossable?
- 11) Is there much value in being able to exceed longitudinal slope, side slope, or vertical step requirements? At what speed?
- 12) How important is fording speed?
- 13) What kinds of mobility characteristics would be most valuable, if "blue sky", advanced concepts are developed?

### Survivability

- 14) What are the most important survivability capabilities? For each mission type?
- 15) How much does the user want to depend on having armor, rather than on other means to avoid being attacked or mined, such as threat detection, acquisition avoidance, or hit avoidance? How much armor is appropriate? For what types of threats? What trades are appropriate between types and amounts of armor vs other technologies?
- 16) What types of mines can we detect so we can avoid them? Are mine detectors effective? Airborne mine sensors? Can mine detectors / mine triggering devices be trusted to the extent of not armoring the underside of the vehicle?
- 17) What is desired location for sensors? Are air borne or elevated sensors appropriate? What types? Would the soldier elevate these sensors to see, or keep them low to avoid detection?
- 18) What are dominant characteristics by which vehicles are detectable?
  - a) acoustic
  - b) visual / UV
  - c) infra-red (SWIR, MWIR, LWIR)
  - d) radar / millimeter wave
  - e) magnetic
  - f) bio-chemical
  - g) weight / tracks
- 19) What kinds of stealth characteristics would be most valuable, if "blue sky", advanced concepts are developed?

### On-Vehicle Equipment (OVE)

- 20) What OVE selections are appropriate for each type of mission?
  - a) communications gear
  - b) weapons & ammunition
  - c) special purpose surveillance and target acquisition gear
  - d) position location and navigation gear
  - e) bolt-on armor
- 21) What OVE must be accessible instantly (eg, under attack)?
- 22) Would there be any benefit to a remote-controlled, stabilized gun mount?
- 23) What type of weapons are essential? Which are desireable? If fired from the vehicle, what should the traversing capabilities be?

### **Soldier Preferences**

- 24) In general, do you prefer automatic controls or manual controls, i.e., transmission, differential locks, CTl. sensors, etc.?
- 25) Would users prefer an open cab or an environmentally controlled, closed cab?
- 26) What is minimum acceptable visibility?
- 27) What type of information would the driver like to have displayed?
- 28) What is the user's seating preference?
- 29) How important is a cold weather package, i.e., defroster / heater?
- 30) What are your biggest concerns with existing light weight military vehicles? How about the HMMWV?

### APPENDIX D

Diesel Engine Survey

## DIESEL ENGINES (PARTIAL LISTING SORTED BY BRITISH HORSEPOWER)

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	CYLS	8	4	4	4	4	4	4	4	4	4	4	4	4	4	4	4	4	4	4	4	4	4	4	4	4	ស	4	9	4	∞	4	4	വ	S	<b>9</b>	io i	<u>ب</u>	<b>x</b> 0	ဖ	ဖ	∞	ဖ	9	ဖ	<b>œ</b>	80	ဖ
FUEL	SYSTEM	٥	-	_	_	۵	_	_	۵	_	_	_	-	_	٥	_	_	۵		۵	9	_		۵	_	_	۵	_	_	۵		_	۵	۵.	۵	_	1	٠ د	- 1	ا ۵	۵	٥	۵	۵	۵	_	- 1	۵
HEIGHT	Z	15.0	25.1	25.8	5 <u>6</u> .9	28.2	25.1	26.9	25.1	25.8	22.1		30.6	28.2	28.2	28.2	26.6			23.4	24.5		28.4			26.4	30.7	26.4	28.9	27.0		26.4		30.7		28.9	30.8	80.8 80.8	50			<b>6</b> 0.0	27.2		28.1	30.3	28.2	
Ë	¥ E	381	637	655	683	715	637	684	638	655	560		11	715	716	715	675			595	622		75			671	780	699	735	685		671		780	!	35	782	787	5			1016	<del>6</del>		714	710	717	
핌	Z	14.3	24.6	20.9	21.0	23.6	26.7	21.0	28.0	25.6	22.3		27.2	26.0	30.1	23.6	24.5			23.0	26.4		23.2			24.2	24.9	24.2	24.5	30.9		24.9		24.9	,	24.5	24.2	24.2	35.3			28.0	24.2		22.5	35.3	24.5	
WIDTH	₹	363	625	531	533	598	212	533	711	650	565		069	099	765	900	623			584	671		289			615	632	615	622	783		633		632		622	615	615	/A			11	615		572	897	622	
TH.	Z	20.0	20.9	26.3	29.7	29.1	20.9	29.7	21.5	22.3	26.0		26.3	28.0	30.9	27.2	23.3			27.3	25.2		24.0			<b>26.8</b>	26.5	26.8	35.3	31.0		25.9		26.5		37.9	27.0	27.0	29.6			45.0	35.4		37.5	29.6	30.3	
LENGTH	Σ	508	230	899	754	738	531	754	546	266	9		299	710	785	691	592			694	640		610			<del>8</del> 84	673	989	968	787		658		673		862	989	9	72/			1143	833		923	752	769	
ACE	_	0.81	1.90	2.20	2.50	3.40	1.90	3.00	06.1	1.90	2.00	2.50	2.50	2.90	60.	2.50	2.10	4.00		2.50	2.30	2.50	2.00	2.10	2.50	2.50	2.50	2.50	4.90	2.50		2.00	4.00	2.50	2.70	8	3.10	3.10	6.50	3.20	2.90	6.70	3.75	00.9	4.16	6.50	8	3.20
DISPLACE	CCIN	49	116	134	153	207	116	177	116	116	122	153	153	177	244	153	128	243		153	<del>5</del>	153	122	128	153	153	153	153	299	153		122	243	153	165	000	189	189	£ .	195	360	409	229	365	254	395	302	195
	9	110	286	348	548	572	297	552	297	308	367	519	520	535	719	<b>4</b> 0	376	614		484	393	517	287	395	206	458	440	458	474	460		293	902	8	459	260	220	22/	44	495	1012	1250	639	922	638	687	561	495
WEIGHT	Š	20	<del>1</del> 3	158	249	260	135	251	135	<del>5</del>	167	236	236	243	327	200	171	279		220	178	235	130	180	230	208	200	208	215	508		133	410	210	508	254	220	20.00	787	225	9	568	3	419	290	312	254	225
RPM-PEAK	TORQUE		2200	2000	2200	1800	1700	2000	1900	2250	2500	2100	2100	2000	2000	2000	2000			1800	2400	2000	2500	1800	2400	2200	2250	2150	2000	2000		4200		2000	1800	2000	2000	1800	00/1	1800	400	1800	2000		1800	1700	3000	2300
ш	¥G		12.34	13.57	16.32	22.64	15.61	20.09	20.6	20.09	20.09	23.05	23.05	22.03	26.01	24.48	24.48	38.86		28.05	18.67	24.99	17.03	26.01	25.6	26.72	27.13	29.99	32.84	29.07		17.85	45.39	29.58	32.64	36.62	37.64	40.75	F. 1.51	38.76	57.12	59.47	44.98	62.69	8.7 8.3	53.24	27.54	<b>6</b> 08
TORQUE	Σ		121	133	160	222	153	197	202	197	197	226	526	216	255	240	240	381		275	183	245	167	255	<b>52</b>	<b>5</b> 95	<b>5</b> 9	294	322	<b>582</b>		175	<del>2</del>	38	320	320	9	9	€ (	380	290	583	4	<del>4</del>	4 5	522	270	9
,	LBFT		8	86	118	164	113	145	149	145	145	167	167	159	88	177	177	281		203	135	181	123	188	185	193	<del>1</del> 8	217	237	210		129	328	214	236	192	2/2	7/7	3	280	413	430	325	475	325	382	66	295
	@ RPM	3600	<del>4</del> 000	4250	3000	3400	4000	3000	4000	4000	4500	4000	4000	3800	3250	4200	4300	2600	4000	4000	4800	3800	2200	4300	4100	4200	4250	4200	3600	4300	2000	9300	2600	900	4300	3400	4200	906	3400	4300	2700	2600	4200	2600	3200	3400	600	4000
POWER	₩ ₩	4	8	61	4	73	92	78	91	5	94	6	66	₽	102	90	106	112	113	114	114	116	116	117	117	22	29	125	126	129	136	98	137	<del>5</del>	4	147	<u>ک</u> ز	20.5	701	1/4	177	178	<b>€</b>	<u>≅</u>	<del>1</del>	193	197	200
PO	KW MHP	8	<b>₹</b>	<b>₹</b>	47	54	20	24	29	29	69	23	73	7	22	78	78	82	8	84	28	82	8	86	88	88	88	35	83	8	8	8	<u>5</u>	<u>ද</u>	90 !	<b>8</b> 5	2 :	21	61	128	90	131	132	<del>2</del>	133	42	<del>1</del>	147
	BHB	8	28	8	g	72	75	9/	8	8	93	86	86	8	101	105	105	110	111	113	113	114	114	115	115	118	118	123	125	127	134	34	135	4	142	£ :	5 6	200	2	172	174	176	111	178	178	90	194	197
	MODEL	LOCR 814 SD	1.9 ADG	R2-22	XA-25	SL-35	1.9 ADE	HA-30	1.9 AFD	XUD9TE	4D68	FSD425T	FSD425T	BD30	TF-40	4056	XUD11ATE	PHASER 110T	300 TDI	425 CDIE	LSG 423	8140.47	DOC4201	M 14 TCA	8144.97	425 CLIR	5 CYL TDI	425 CLIE	CSG649	DKSATE	LR 3.6P	DOC420I 16V	PHASER 135TI	5 CYL 140HP	M 15 TCA	CSG 649	531 CLIE	531 CDIE	6.5 V8 U NA	M 16 TCA	8060.25	SERIES 40	638	PHASER 180TI	D 642	6.5L V8 THPD	CSG 850	M 16TCAHD
	MANUFACTURER	RPI	*	MAZDA	MAZDA	MAZDA	8	MAZDA	*	PEUGEOT	MITSUBSH	FORD/EUR	FORD EUR	NISSAN	MAZDA	MITSUBSH	PEUGEOT	PERKINS	ROVER	₹	FORD	IVECO	FORD EUR	STEYR	IVECO	₹	AUDI	₩	FORD EUR	PEUGEOT	LANDROVE	FORD EUR	PERKINS	AUDI	STEYR	FORD	₩ :	<b>M</b>	GMVO	STEYR	IVECO	6	₩	PERKINS	<b>M</b>	GMVO	FORD	STEYR

### APPENDIX E

General Engine Survey

				Diese	Diesel Engines		
Power Range (BHP)	Mfg	Model Number	Weight (Ib)	ВНР х грм	ВНРЛЬ	T (ft-lb) x rpm	Comments
50 - 100	WM	1.9 AFD	297	90 x 4000	0:30	149 x 1900	116 cu in, 1.9L, 21.5"L x 28"W x 25.1"H, 4 cyl Turbo Direct Injection
50 - 100	Peugeot	XUD9TE	308	90 x 4000	0.29	145 x 2250	116 cu in, 1.9L, 22.3"L x 25.6"W x 25.8"H, 4 cyl Turbo Induction
50 - 100	Mitsubishi	4D68	367	93 x 4500	0.25	145 x 2500	122 cu in, 2.0L, 26.0"L x 22.3"W x 22.1"H, 4 cyl Turbo Induction
50 - 100	MΛ	1.9 ADE	297	75 x 4000	0.25	113 x 1700	116u in, 1.9 20.9"L x 26.7"W x 25.1"H, 4 cyl Turbo
101 - 125	Ford Eur	DOC420I	287	114 x 5500	0.40	123 x 2500	122 cu in, 2.0L, 24.0"L x 23.2"W x 28.4"H, 4 cyl Naturally Aspirated
101 - 125	Ford	LSG 423	393	113 x 4800	0.29	135 x 2400	140 cu in, 2.3L, 25.2"L x 26.4"W x 24.5"H, 4 cyl Naturally Aspirated
101 - 125	Steyr	M14 TCA	395	115 x 4300	0.29	188 x 1800	128 cu in, 2.1L, 4 cyl Turbo Direct Injection
101 - 125	Peugeot	XUDIIATE	376	105 x 4300	0.28	177 × 2000	128 cu in, 2.1L, 23.3"L x 24.5"W x 26.6"H, 4 cyl Turbo Induction
126 - 150	Ford Eur	DOC4201	293	134 x 6300	0.46	129 x 4200	16V, 122 cu in, 2.0L, 25.9"L x 24.9"W x 26.4"H, 4 cyl Naturally Aspirated
126 - 150	Auđi		440	141 x 4000	0.32	214 x 2000	153 cu in, 2.5L, 26.5"L x 24.9"W x 30.7"H, 5 cyl Turbo Induction
126 - 150	Steyr	MISTCA	459	142 x 4300	0.31	236 x 1800	165 cu in, 2.7L, 5 cyl Turbo

				Diese	Diesel Engines		
Power Range	Mfg	Model Number	Weight (1b)	ВНР х грш	ВНРЛЬ	T (ft-1b) x rpm	Comments
126-150	Peugeot	DKSATE	460	127 x 4300	0.28	210 x 2000	153 cu in, 2.5L, 31.0"L x 30.9"W x 27.0"H, 4 cyl Turbo Induction
151 - 175	Steyr	MIGTCA	495	172 x 4300	0.35	280 x 1800	195 cu in, 3.2L, 6 cyl Turbo Induction
151 - 175	ВМ	6.5L	644	165 x 3400	0.26	300 x 1700	395 cu in, 6.5L, 30.5"L x 26.2"W x 28.4"H, 8 cyl Naturally Aspirated
176 - 200	RPI	2013R	340	200 x 6000	0.59	175 x 6000	81 cu in, 1.33L, 24.1"L x 29.3"W x 17.7"H, 2 cyl Turbo Induction
176 - 200	Steyr	M16TCAHD	225	197 x 4000	0.40	295 x 2300	195 cu in, 3.2L, 6 cyl Turbo Induction
176 - 200	Ford	CSG850	561	194 x 4000	0.35	199 x 3000	302 cu in, 5.0L, 30.3"L x 24.5"W x 28.2"H, 8 cyl Naturally Aspirated
201 - 250	Ford Eur	CSG850	536	235 x 4800	0.44	278 x 4000	305 cu in, 5.0L, 29.4"L x 26.1"W x 28.7"H, 8 cyl Turbo
201 - 250	GM	6.5L-TD	687	250 x 3600	0.36	400 x 2400	395 cu in, 6.5L, 30.5"L x 26.2"W x 28.4"H, 8 cyl
201 - 250	Ford	NSD873T	860	211 x 3000	0.25	425 x 2000	445 cu in, 7.5L, 33.1"L x 27.8"W x 34.6"H, 8 cyl Turbo
201 - 250	Cummins	B5.9	942	230 x 2200	0.24	605 x 1400	359 cu in, 5.9L, 40"L x 35.5"H x 32.4"W, 6 cyl
250 +	Deere	6081Н	1710	255 x 2200	0.15		496 cu in, 8,1L, 47.6"L x 28.6"W x 40.2"H, 6 cyl
250 +	Deere	1058R	910	375 x ?	0.41		350 cu in, 5.8L, Rotary Diesel

### Notes:

- 1. Only diesel engines, from reputable suppliers, were considered since this is the primary choice of most military services & contractors.
- There are many other engines available. The engines chosen in the table were the top 3 or 4 candidates in each power range in terms of their BHP/lb rating.
- production market share to date, perhaps due to their relatively low torque. These engines are smaller and lighter because they do not have fact, they can run on diesel, gasoline or even jet fuel. Another important advantage to military applications are their stealth advantages. Rotary diesels have lower heat rejection rates (lower IR signature) and the dominant source of noise is the exhaust which can be treated to convert reciprocal motion into rotary motion like conventional engines do. They are also omnivorous, or, intolerant to fuel quality. The highest power-to-weight ratios are the RPI 2013R and the Deere 1058R which are rotary diesel engines. Neither has earned any with silencers.
- The Steyr engines look good but lack logistic support as they have not been made in production quantities to date. Rod Millen uses a 6 cylinder Steyr engine in the HTTMP and has not been impressed by Steyr's support. 4
- Europeans are estimating a 50% increase in future power densities as a result of using multiple valves per cylinder.

				Gasoline Engines	Engines		
Power Range (BHP)	Mfg	Model Number	Weight (1b)	ВНР х грт	ВНРЛЬ	T (ft-lb) x rpm	Comments
50 - 100	Mitsubishi	4G63	315	90 x 2000	0.28	101 x 1800	2L, 4 cyl in line
50 - 100	Peugot	TUSJP	221	100 x 5600	0.22	100 x 3000	1.61, 4 cyl in line, horizontal, liq cooled
101 - 125	Ford	LSG423	319	119 x 5400	0.37	127 x 2800	2.2L, 4 cyl in line, horizontal, liq cooled
101 - 125	GM	2.2L (MPFI)	280	110 x 5200	0.39	123 x 2700	2.2L, 4 cyl in line, horizontal, liq cooled
101 - 125	Peugot	XU92C	283	107 x 6000	0.38	118 x 3000	1.9L, 4 cyl in line, horizontal, liq cooled
126 - 150	GM	3.0L (TBI)	363	148 x 4800	0.41	185 x 3600	2.2L, 4 cyl in line, horizontal, liq cooled
126 - 150	Mitsubishi	6A11	313	133 x 6000	0.42	108 x 3500	1.6L, V6
126 - 150	Peugot	XU10J2TE	298	139 x 4400	0.58	168 x 2200	2.0L, 4 cyl in line, horizontal, liq cooled
151 - 175	Mitsubishi	6G72	375	153 x 5000	0.41	163 x 2500	2.5L. V6
176 - 200	Peugot	XU10J4TE	364	192 x 5000	0.53	214 x 2600	2.0L, 4 cyl in line, horizontal, liq cooled
176 - 200	GM	4.3L (2-BBL)	388	184 x 4400	0.47	230 x 2400	4.3L, V6, horizontal, liq cooled
201 - 250	Ford	CSG649	473	213 x 4200	0.45	245 x 2000	4.9L 6 cyl in line, horizontal, liq cooled
201 - 250	GM	4.3L (TBI)	388	211 x 4600	0.54	265 x 2800	4.3L, V6 horizontal, liq cooled
250 +	Ford	LSG875	713	339 x 4600	0.47	373 x 1700	7.6L V8, horizontal, liq cooled
250 +	GM	5.7L HO (TBI)	473	280 x 4400	0.59	330 x 3200	5.7L, V8, horizontal, liq cooled

### Notes:

<sup>1.</sup> Gasoline engines are typically lighter than equivalent diesel engines due to the lower operating pressures.

### APPENDIX F

**Propulsion System Calculations** 

### Max Speed -- Requirement = 75 mph (level ground)

· Rolling Resistance on Pavement:

$$Frr = 0.03 * 6500 lb = 195 lb$$

• Air Resistance (Far =  $0.26 \text{ Cd A}_{\text{f}} (\text{V}/10)^2$ )

Far = 
$$0.26 * 0.75 * 22 * (75/10)^2 = 241.3 \text{ lb}$$

· Required Engine Power:

$$(\sum F) (V) / 375$$
  
 $(195 + 241.3) * 75 / (375 * 80\% eff) = 109 hp (81 kW)$ 

• Engine Speed:

### Start on 60% Slope

· Rolling Resistance on medium soil:

$$Frr = fw = (0.1) * (6500 lb) = 650 lb$$

· Grade Resistance:

$$Fg = wg = 60\% * (6500 lb) = 3900 lb$$

· Total Force at Wheels:

$$(Frr) + (Fg) = 650 lb + 3900 lb = 4550 lb$$

Torque on Axles:

$$(4550 \text{ lb}) * 1.31 \text{ ft} = 5960 \text{ ft-lb}$$

. Max Engine Torque Required:

$$(5960 \text{ ft-lb}) / (4.56 * 1 * 2.48 * 1.9) = 277 \text{ ft-lb}$$

### Climb 5% Grade at 40 mph

· Rolling Resistance on medium soil:

$$Frr = fw = (0.1) * (6500 lb) = 650 lb$$

· Air Resistance:

Far = 
$$0.26 * 0.75 * 22 * (40/10)^2 = 68.6$$
 lb

· Grade Resistance:

$$F_g = wg = 5\% * (6500 lb) = 325 lb$$

Required Engine Power:

$$(650 + 68.6 + 325) * 40 / (375 * 80\% eff) = 139 hp (104 kW)$$

### Tow Self & Equal up 40% Slope

· Rolling Resistance on medium soil:

$$Frr = fw + fw = 2*[(0.1)*(6500 lb)] = 1300 lb$$

- · Air Resistance Neglected
- · Grade Resistance:

$$Fg = wg = 40\% * (13000 lb) = 5200 lb$$

· Using 139 hp as calculated above:

$$HP = (\sum F)(V) / 375$$

Solving for V:

$$V = (139) * (375) * 80\% / (1300 + 5200) = 6.42 \text{ mph}$$

### Acceleration (ROM estimate only)

· Requirement: 0 to 30 mph in 6 seconds

$$V_f = V_i + at$$

$$a = 44 \text{ fps } / 6 \text{ sec} = 7.33 \text{ ft/sec}^2$$

$$F = ma = w/g * a = (6500 lb) * 7.33 ft/sec^2 / 32.2 ft/sec^2 = 1480 lb$$

Rolling Resistance on pavement:

$$F_{TT} = fw = 0.03 * 6500 lb = 195 lb$$

$$HP = (1480 + 195) * (30) / (375 * 80\%) = 168 \text{ hp} (125 \text{ kW})$$

· Requirement: 0 to 60 in 15 seconds

Determine acceleration from 30 to 60 mph in 9 seconds

$$V_f = V_i + at$$

$$a = (88 \text{ fps} - 44 \text{ fps}) / 9 \text{ sec} = 4.89 \text{ ft/sec}^2$$

Far = 
$$0.26 * 0.75 * 22 * (60/10)^2 = 154.4 \text{ lb}$$

$$F = ma = w/g * a = (6500 lb) * 4.89 ft/sec^2 / 32.2 ft/sec^2 = 988 lb$$

$$Vrms = SQRT [(60^2 + 30^2)/2] = 47.4 mph$$

$$HP = (154.4 + 195 + 988) * (47.4) / (375 * 80\%) = 211 hp (157 kW)$$

$$HP_{0-60} = SQRT [(168^2 + 211^2)/2] = 191 \text{ hp } (142 \text{ kW})$$

### APPENDIX G

Technology Tree for Light Strike/Reconnaissance Vehicles

### Technology Tree for Light Strike / Reconnaissance Vehicles (Top Level)

### **Mobility** A) propulsion 1) engine 2) battery 3) transmission 4) differential 5) hybrid drive 6) motor/wheel

- B) suspension 1) springs
  - 2) shocks
  - 3) active control

- C) vehicle support
  - 1) fuel
  - 2) lubricants
  - 3) coolant
  - 4) OVE items
  - 5) occupant protection

### Survivability

- A) acquisition avoidance \*\*

  (acquisite radar thermal, (acoustic, radar, thermal, visual, magnetic)
  - 1) signature mgmt (paints, materials, covers, elect control)
  - 2) decoys (passive, active, remote, local)
  - 3) sensor jamming (wide or narrow bandwidth)

- B) hit avoidance
  - 1) overload
  - 2) spoofing
  - 3) anti-ballistic weapons
  - 4) high energy defense
- C) penetration avoidance \*\*
  - 1) applicae armor
  - 2) reactive armor
  - 3) ceramics
  - 4) composites 5) spall liner

- D) kill avoidance
  - 1) fire prevention & suppression
  - 2) crew partitioning
  - 3) NBC alarm/protection
- E) threat detection \*\* (wide area sensors, mine detection)
  - 1) IR
  - 2) visual / UV
  - 3) radar / mm wave
  - 4) magnetic
  - 5) acoustic
  - 6) external data

### Firepower

- A) weapons
  - 1) TOW launchers
  - 2) grenade launchers
  - 3) machine guns
  - 4) small caliber
  - 5) high energy

- B) ammo storage
  - 1) TOW missiles
  - 2) grenades
  - 3) machine gun ammo 4) small caliber ammo
  - 5) mines

- C) target acquisition \*\* (accurate tracking sensors)
  - 1) IR
  - 2) visual / UV
  - 3) radar / mm wave
  - 4) magnetic
  - 5) acoustic
  - 6) external data

### C4I

- A) external communication
  - 1) vehicle mounted radio
  - 2) portable radios
  - 3) secure voice radios
  - 4) antennas
  - 5) satellite links

- B) internal communication
  - 1) radios in helmets
- C) position location
  - 1) **GPS**
  - 2) PLRS
  - 3) inertial reference
  - 4) attitude reference
- D) computations
  - 1) ruggedized processor
  - 2) rad-hard capability

\*\* Initial Focus

### APPENDIX H

Survivability Survey Results

## **Assessment of Technologies**

## **Assessment Parameters**

- Ef = Effectiveness in percent success
- \$\$ = Production Cost in dollars per vehicle
- Wt = Weight in pounds per vehicle
- Yr = Year of Technology Readiness

### Quality of Assessment

- 1 = engineering judgement (generally within factor of 2 of correct assessment)
  - 2 = based on application of technology to an existing product
- 3 = calculated using quantitative evaluation of technology as applied to our system
  - 4 = demonstrated by actual application to our system testbed

### **Assessment Example**

(Ef1: 25) (\$\$1: 350) (Wt1: 45) (Yr1: 2001) says that

- (for example, for acoustic signature reduction technologies, if vehicle would have been detected acoustically in 100 scenarios, the Effectiveness rating (using engineering judgement) causes success in 25 percent of situations technology being assessed reduces the detection by 25 percent to 75 scenarios)
  - Production Cost (using engineering judgement) is 350 dollars for one vehicle
    - Weight (using engineering judgement) is 45 pounds for one vehicle
- Year of technology availability (for initiating design) (using engineering judgement) is 2001

# Technology Tree for Light Strike / Reconnaissance Vehicles (Second Level) Survivability Item A1 -- Acquisition Avoidance -- Signature Management

	סמו אוומשווונא ווכוו ליווומשווא ואס	Sai vivability item At Acquisition Avoluance Signature Management	agement
Signature	Selection of Materials	Control Techniques	System Design Approach
Acoustic	<ul> <li>basic paints have very little sound deadening capability; paints which trap air bubbles have minor acoustical benefits (Ef1: &lt;1) (\$\$1: 50) (W1: 10) (Yr1: 1996)</li> </ul>	<ul> <li>electronic noise cancellation can provide excellent attenuation of regular low frequency noise generated at the vehicle, especially if the noise cancellation signal is nearly co.</li> </ul>	<ul> <li>steatth movement using electric drive only will significantly reduce the acoustical signature does not help well with white noise. (E41: 80) (</li></ul>
	<ul> <li>materials constructed with dead air spaces have sound-deadening capabilities; a rough surface breaks up acoustic reflections, as well as attenuating</li> </ul>	located with the source, this applies to noise such as the hum produced by rotating machinery, but does not help well with white	• use of low road noise tires will improve remaining signature (Ef1: 10) (\$\$1: 400) (Wt1: 50) (Yr1: 1998)
	radiated hoise (Eff: 1) (\$\$1: 500) (Wt1: 50) (Yr1: 1996)  • material sandwiches with loose (or indirect) physical coupling between the outer surfaces are very good at	noise. (Ef1: 10) (\$\$1: 1K) (Wt1: 50) (Yr1: 1996)	
	attenuating acoustical transmissions; the center of the sandwich should be composed of lots of small dead air spaces, such as foam (Ef1: 5) (\$\$1: 500) (Wt1: 200) (Yr1: 1996)		
	<ul> <li>covers constructed as material sandwiches provide good acoustical attenuation (Ef1: 5) (\$\$1: 500) (Wt1: 100) (Yr1: 1996)</li> </ul>		
Ultra-violet / Visual	<ul> <li>uttra-violet and visual signatures are mostly dominated by the surface characteristics of the materials; paints or other surface characteristics can match the ultra-violet or visual characteristics of the</li> </ul>	<ul> <li>see note in "materials" column. technology to change surface characteristics is likely to be based on a combination of chemical and electronic technologies</li> </ul>	•
	background; (Ef1: 20) (\$\$1: 500) (Wt1: 10) (Yr1: 1996)  • changeable background characteristics require that the surface coating itself be responsive to the environment; this technology is likely to be in the		
	formative stages at this time, and probably be useable in the 5-8 year time frame. Such technology is probably classified.  (Eft: 80) (\$\$1: 25K) (Wt1: 100) (Yr1: 2002)		

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Signature	Selection of Materials	Control lecuniques	System Design Approach
Infra-red / Thermal	paints can be designed to provide some degree of thermal insulation, thereby reducing the MWIR / LWIR (thermal) signature	electronic transfer of heat to the center of the vehicle will reduce the thermal signature	
	<ul> <li>(ETT: 10) (\$\$1: 400) (WTI: 50) (YTI: 1990)</li> <li>paints can be designed to radiate poorly in the SWIR ranges</li> <li>(EfT: 20) (\$\$1: 400) (WtI: 50) (YTI: 1996)</li> </ul>		
	<ul> <li>responsive surface material see note under ultra- violet / visual materials (Ef1: 20) (\$\$1: 20K) (Wt1: 500) (Yr1: 1996)</li> </ul>		
	<ul> <li>materials which are good insulators provide reasonable thermal signature attenuation (E11: 50) (\$\$1: 2K) (W\$1: 300) (Y\$1: 1996)</li> <li>low signature covers can be designed using the other</li> </ul>		
Radar /	paints which are properly designed have low radar reflectivity, by incorporating materials which do not interact with the electromagnetic environment	electronic control techniques can most likely     be used to enhance the radar absorption of the materials used. Specific materials and	
Wave	<ul> <li>the physical configuration of materials used has a strong influence on radar (&amp; mm wave reflectivity)</li> <li>selected materials provide substantially reduced EM interactions, yielding reduced radar reflectivity</li> </ul>	control techniques are not presently known by our team.	
	<ul> <li>materials which absorb radar energy but do not re-radiate the energy would be optimum for use (details are not presently known by our team)</li> <li>note also that materials which are transparent to radar let the radar signal propagate to the inner materials</li> </ul>		
Magnetic	materials which are non-magnetic can be used to minimize the magnetic signature	<ul> <li>electronic control techniques should be able to provide magnetic shielding to effectively eliminate the external magnetic signature (e.g., below the vehicle)</li> </ul>	
Bio- Chemical	<ul> <li>paints and materials which have cured, and are no longer outgassing should not leave a significant biochemical signature</li> <li>use of very clean burning fuels (such as propane) can substantially reduce the bio-chemical signature</li> <li>bio-chemical signature scrubbers such as activated charcoal can be used to clean up problem areas</li> </ul>	electronic (high voltage) scrubbers can be used to help reduce the bio-chemical signature	
Weight	wide tires can distribute the weight and leave less of an imprint	N/A	

# Technology Tree for Light Strike / Reconnaissance Vehicles (Second Level) Survivability Items A2 and A3 -- Acquisition Avoidance -- Decoys & Sensor Jamming

Observable	Technology Approach for Gener	Technology Approach for Generating Decoys & Sensor Jamming
Signature	Decoys	Jamming
Acoustic	<ul> <li>noise makers are small, lightweight, disposable</li> <li>deployment techniques</li> <li>drop off</li> <li>launch/lob from vehicle</li> <li>tow behind vehicle on tether</li> </ul>	<ul> <li>directed energy (acoustic or other) at the offending sensor is more appropriate than wide bandwidth jamming; operationally unattractive</li> </ul>
Ultra-violet / Visual	<ul> <li>stationary: quick inflation balloons, shaped like vehicles, men,</li> <li>mobile: quick inflation balloons, on wheels, small drive motor, controlled similarly to model planes</li> </ul>	<ul> <li>directed energy at the offending sensor is more appropriate than wide bandwidth jamming; operationally unattractive</li> </ul>
Infra-red / Thermal	same as for UV/visual	<ul> <li>directed energy at the offending sensor is more appropriate than wide bandwidth jamming: operationally unattractive</li> </ul>
Radar / Millimeter Wave	objects with higher reflectivity than the vehicle can be deployed     (equivalent to corner cube for optical reflections)	<ul> <li>straightforward technology, but operationally unattractive</li> </ul>
Magnetic	Not Applicable	Not Applicable
Weight	Not Applicable	Not Applicable

# Technology Tree for Light Strike / Reconnaissance Vehicles (Second Level)

	SULVIVADIII	Survivability Item E Inreat Detection	
Observable	<b>Techn</b>	Technology Approach for Threat Detection	ection
Signature	Omni-direction Sensors	Scanning Sensors	Mine detection
Acoustic	<ul> <li>microphones, along with pattern recognition software can identify known signals even in substantial background noise</li> </ul>	<ul> <li>scanning sensors have little benefit, unless the direction of the threat is known; then directional capability is valuable to enhance the signal relative to noise prior to processing</li> </ul>	<ul> <li>mines which are triggered by vibration could be triggered by acoustics</li> </ul>
Ultra-violet / Visual	<ul> <li>muzzle flash detection (large weapon) can be performed by an omni-directional sensor</li> <li>detection of threats other than muzzle flashes requires image detection and processing, and a sensor with a narrower field of view than an omni-directional sensor</li> </ul>	<ul> <li>tremendous detection capability is available and requires significant image processing; detectability depends on processing capability, stealth techniques used by threat, location of sensor, threat</li> </ul>	<ul> <li>well designed mines can be hidden from UV and visible sensors</li> </ul>
Infra-red / Thermal	<ul> <li>detection requires image detection and processing, and a sensor with a narrower field of view than an omni-directional sensor</li> </ul>	<ul> <li>tremendous detection capability is available and requires significant image processing; detectability depends on processing capability, steatth techniques used by threat, location of sensor, threat; use of MWIR can be most effective, at higher cost</li> </ul>	<ul> <li>well designed mines can be hidden from IR sensors</li> </ul>
Radar / Millimeter Wave	<ul> <li>omni-directional sensors can detect radar being transmitted by a threat</li> </ul>	<ul> <li>scanning radar can identify threrats, but only at the expense of radiating energy, and thereby becoming easily detectable</li> <li>scanning bi-static radar, with the transmitter located remotely from the vehicle is not susceptible to the detectability problem</li> </ul>	<ul> <li>radar can theoretically detect metal objects</li> <li>underground</li> <li>•</li> </ul>
Magnetic			•
Weight	•		•
External Data	•		

					ı
Index	ndex Tech. Category	Det./ Acq.	Vehicle Types	Сотрапу	
2	2 Acoustic	Det	All • Handheld	Jaycor	H
S	53 Bio-Chemical	Det	All	Argonne National Laboratory	Т-
R	54 Bio-Chemical	Det	All	Argonne National Laboratory	_
•	1 Bio-Chemical	Det	AH	Jaycor	T
8	60 Bio-Chemical	Det	All	OCETA Enviro. Tech Barringer Res. Ltd	_
49	49 Bio-Chemical	Det	All	U.S. Army Edgewood RDEC	$\overline{}$
6	3 Bio-Chemical	Det	All * Handheld	Rocketdyne, et al.	Τ-
ଜ	50 Bio-Chemical	Oet	All * Handheld	U.S. Army Edgewood RDEC, Aberdeen - Skunkworks Projects	_
52	52 Bio-Chemical	Det	All * Handheld	U.S. Army Edgewood RDEC, Graesby Dynamics, Ltd.	_
51	51 Bio-Chemical	Det	All * Handheld	U.S. Army Edgewood RDEC, Graesby/FemtoScan, University of Utah - SBIR Program	Т
12	12 Comm. Intercept	Det	All	P.O.C. Mr. John Holzman, DSN 229-6816, COMM 703-349-6816	т
<b>8</b>	69 Infrared	Acq	All		т-
73	73 Infrared	Acq	All		т-
200	79 Infrared	Acq	All		_
4	40 Infrared	Acq	All	MICOM RDEC	т —
33	39 Infrared	Acq	All * Weap. Mount	* Weap. Mount Insight Technology	_
62	62 Infrared	Acq	All * Weap. Mount	* Weap. Mount Rapid Acquisition Targeting Systems	т—
48	48 Infrared	Acq	All * Weap. Mount	Santa Barbara Research Center	_
29	67 Infrared	Det	All	Amber (Raytheon)	т-
65	65 Infrared	Det	All	Amber (Raytheon)	_
98	66 Infrared	Det	All	Amber (Raytheon)	<del>-</del>
8	64 Infrared	Det	All	Amber (Raytheon)	1
34	34 Infrared	Det	All	Inframetrics	1
8	81 Infrared	Det	All	Inframetrics	T
24	24 Infrared	Det	All	Texas Instruments	_
87	82 Infrared	Det & Acq	All	Inframetrics	Т
35	35 Infrared	Det & Acq	All	inframetrics	_
8	36 Infrared	Det & Acq	All * Handheld	Litton	т —
₹	43 Infrared	Det & Acq	All * Weap. Mount	Hughes	Τ-
31	31 Infrared	Det & Acq	All * Weap. Mount Hughes	Hughes	_
42	42 Infrared	Det & Acq	Med, 7LW, SLW? Hughes	Hughes	_
37	37 Infrared	Det & Acq	Med, ?LW, SLW?	Med, ?LW, SLW? Texas Instruments (P) & Hughes (Sub)	Т
7	71 Laser	Acq	All		T
					1

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, april	Dradies News		Malabe
Y DEL		Froduct 10	vergin
	2 Ultrasonic Imaging Technology		Hand held
5	53 Associated-particle, sealed-tube neutron generator	APSTNG	
4)	54 Handheld Portable X-ray Fluoresence Spectrum Analyzer		
	1 Spectrophone		Man-portable
	60 IONSCAN IMS Detection System	IONSCAN	35lbs for Detector Module,
4	49 Joint Services Lightweight Standoff Chemical Agent Detector	LSCAD (JSLSCAD)	Prototype 22 lbs
	3 Ecoscan		Man portable (picture at UR
٠,	50 Lightweight Chemical Detector (LCD) w/ Data Archival	COT	2.6 lbs
	52 Miniature Ion Mobility Spectrometer	S	1.6 lb
,	51 Environmental Vapor Monitor (EVM II)	EWII	Hand held
	12 Lightweight Man-Transportable Radio Direction Finder System LMRDFS AN/PRD-12		Two Man Portable
	69 AIM-1	AIMMLR, AIM/EXL, AIM-1/D, AIM-1/DLR, AIM-1/MLR, AIM-1/EXL Wespon/Turret Mounted	Weapon/Turret Mounted
1	73 Ground Commanders Pointers	GCP-1, GCP-1A NSN: 5855-01-420-0849	4.5 02
	79 Nite Eye		1.1 kg
	40 Small Diameter Seeker (Block II Stinger) and TACAWS Infrared Seeker (		
•"	39 Infrared Aiming Light	ANIPAQ-4. ANIPAQ-4A. ANIPAQ-4B, ANIPAQ-4C, ANIPEQ-1 S	2.7 oz (76.68 g) w/o batteri
9	62 RATS 840 Infra-Red (IR)	RATS 840 IR	
4	48 Staring and Scanning Focal Plane Arrays	SADAII, SADAIIIB, et al.	
9	67 Argus Sentry		Enclosure w/housing = 60 l
9	65 Aurora (Radiance Aurora)		10 lbs
	66 Galileo (MWIR Camera System)		=9lbs
	64 Sentinel		4.5 lbs w/battery, 3.8 lbs w/
(*)	34 MIICAM		4lbs (1.818 kg)
3	81 Thermacam		2.6 kg total, 1.7 kg camera
	24 Driver's Vision Enhancer (DVE)	AN/VAS-5 (D1000)	<17 lbs
3	82 Infracam		2.93 ibs total
63	35 Long Range Infrared System	LORIS	35 lbs (15.9 kg)
63	36 Mini Eyesafe Laser Infrared Observation SET (MELIOS	AN/PVS-6	4 fbs, 6.5 lbs w/tripod, 13 lb
4	43 Multipurpose Thermal Sight	NIPAS-19	Man Portable
פא	31 Thermal Weapon Sight	AN/PAS-13	Man Portable
4	42 Hughes Infrared Equipment (HIRE), GMHE Integrated TOW Sight	SJ	Light weight - modular
(4)	37 Horizontal Technology Integration Second Generation FLIR	HTISGF	
_	71 Ground/Vehicle Laser Locator Designator	AN/TVQ-2 (G/VLLD)	Man Portable

Ž	ndex Unit Price	Space Reds	Power Ress	Avelahitt
[		Hand Held		Colores
1	T	Public Public		
3	Inexpensive	relatively small		Prototype - needs ruggedization and additional
R		Hand Heid		Current
1				Future- Chemical Fingerprints (baselines) nee
90		14" x 12.5" x 12.5" Detector, 8.5" x 12.5" x 12.5" Pump	110/220 VAC, DC option available	Current
49		0.3 cu ft + turret scanner 10 cu inches, + a 17" x 13" oval 28 VDC at 2 amps	28 VDC at 2 amps	Prototype
e.	3 1/10th of current Hand Held	Hand Held		Current -1995
20	\$100's/unit (+dat	50 \$100's/unit (+dat   Prototype: 8" x 3.5" x 3", Next Generation: 7" x 2" x 1.5"	<1 watt	Prototype available
52	52 relatively low	.02 cu ft	Battery or External Power	Current
51		Hand Portable	24VDC - Battery or External Power supply	Current
12				1996
69		Weapon/Turret Mounted		Currently Fielded
73		Used with Night Vision devices	2 - AA Batteries	Currently Fielded
79			2 AA Batteries	Currently Fielded
40		2.75" dlameter, 5.85" dlameter		
36	39 \$402	5.5"L x 2.5"H x 1.2"W sight, 8" x 12" x 10" Transit Case	1 BA-5567 or 2 AA batteries (BA-3058)	Currently fielded
62		.48"W x .40"H x .50"L	6VDC, PX28L Battery Type	Current
48		Weapon Integrated		Current
29		Camera = 4.4"W x 10.3"D x 7.2" H, Rain Shield 6" x 10"	120V AC/60 HZ or 230V AC/ 50 HZ, 350 Watts max,	Current
65		5.1"W×9"D×5.4"H	19V to 36V DC, 55 watts	Current
8		4.2'W x 6.75"D x 5.7"H		Current
8		4.5"W×3.5" H×5" L	6VDC typ, 6.8VDC max. Standard Battery - SONY NP Current	Current
क्ष		8.25" x 3.75" x 4.0"	6V Camcorder battery or 120/240 VAC 50/60 HZ, <5 w Current	Current
19		10.3"L x 4.5"W x 3.8" H total, 8.3"L x 4.5"W x 3.3"H came	8.3"L x 4.5"W x 3.3"H came 6VDC, battery operation >2 hours with 4.5 AH Camcor	Current
24		10.2"Wx14.8"Hx7.2"D	28 VDC nominal, 16 VDC-32 VDC operating. Typical	2nd Quarter 1997 Full Production, Currently in
82		5.3" x 9.7" x 2.5"	<5 watts, 2 hour life with 6 VDC Camcorder battery, A	Current
35		9" Diam. x 24"L	<15 Watts @12VDC	Current
98		6"x12"x14"		Current
₽		Weapon Mounted, Individual and/or Crew Served		Сител
31		Weapon Mounted		Low Rate Production 1995, Full Production 19
42		Picture-http://www.cerf.net:80/hac/products/hire.htm		Current production since 1991. Second gener
37				Testing in 1996 for M2A3 and M1A2, Decisio
71				Currently Flekded

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Index	Index Tech. Category	Det./ Acq.	Vehicle Types	Company
75	75 Laser	Acq	All	
74	74 Laser	Acq	All	Cadillac Gage Textron (P)
4	41 Laser	Acq	All	Hughes
7	7 Laser	Acq	All	Hughes Aircraft
72	72 Laser	Acq	All	Navy
16	16 Laser	Acq		Radio Corp. et al.
2	70 Laser	Acq	All • Handheld	
9/	76 Laser	Acq	All * Handheld	
11	77 Laser	Acq	All • Handheld	RCA
8	86 Laser	Det & Acq.	All • Handheld	Leica
47	47 LIDAR	Det	All	Orca Photonics
4	46 LIDAR	Det	All	Science and Engineering Services Inc.
S	59 Mine	Det Det	All	A&S Company
8	56 Mine	Det	All	GDE - P.O.C. Ray Garriot (619) 592-5567
•	8 Mine	Det	All	Jaycor
S	59 Mine	Det	A	Science Applications International Corp., Humanitarian Demining BAA, US Army CECOM (NVESD)
28	28 Mine	Det Det	All • Handheld	Schiebel Instruments, Inc.
2	21 Mine	Set C	All * Man Portable	Man Portable P.O.C. Mr. Mark Locke, DSN 654-2418, COMM 703/704-2418
57	57 Mine	Det O	Med, 7LW, SLW?	7LW, SLW7 DARPA, LLNL, NRaD, SAC, SCC, TRA, Univ. of Hawaii
8	68 Mine	Det O	Med, 7LW, SLW?	7LW, SLW7 U.S. D.O.D. ESTCP
61	61 Mine	Det & Acq	Med, 7LW, SLW?	?LW, SLW?   Martin Marietta
87	87 Multi - A, BC, M, S, V, W	Det	All * Man Portable	Man Portable DARPA, CMO, AF, Marines, USSOCOM, USCENTCOM
14	14 Multi - A, Data, IR, Las, V	Det & Acq	ם ו	7LW, SLW? Dismounted BattleSpace Battle Lab - IEW, ASM
13	13 Multi - A, IR, M, S	Det	All	Lockheed Martin Control Systems
88	88 Multi - Data, IR, N, V	Det	All * Man Portable	Dismounted Battlespace BattleLab - IEW
8	38 Multi - IR, Las, V	Det & Acq	All	P.O.C. Maj. Byrnside, DSN 654-2915, COMM703/704-2915
8	80 Multi - IR, Las, V	Det & Acq	All	System Planning Corporation
8	89 Muiti - IR, N	Det	All	Dismounted Battle Lab
22	25 Multi - IR, N	Det	All • Handheld	NightVision Engineering & Manufacturing, L.L.C.
85	85 Multi - IR, N, R, SIGINT, V	Det & Acq	Med, 7LS, SLW?	Motorola
83	83 Mutti - IR, N, V	Det & Acq		7LW, SLW7 Boeing (P), Hughes (Sub)
97	97 Multi - IR, R	Det & Acq		Alliant Techsystems
92	92 Multi - IR, R	Det & Acq	All	Thomson-CSF (France), Vision Abell (Australia)

Index Product Name	Product ID	Aleks Selection of the selection of the
75		
74Light Armored Vehicle 105 mm gun laser range finder	LAV-105	
41 Eyesafe Laser with Integrated Telescope Equipment	ELITE	
7 Modular Universal Laser Equipment (MULE)	AN/PAQ-3	42lbs (19.07 kg) Daytime; 1
72 Compact Laser Designator	CLD	Lightweight
16 Laser Rangefinder (Infrared Observation Set)	AN/GVS-5 5 tbs (	5 lbs (2.27 kg) w/battery
70 Laser Target Designator	AN/PAQ-1 (LTD) Handf	Handheld
76 Tactical Laser Pointer	LPL-30 Man P	Man Portable
77 Mini Laser Rangefinder	AN/PVS-X (MLRF) lightw	lightweight
86 Leica Enhanced Laser Rangefinder	\$Let	Handheld Binoculars
47 Trasnportable Raman Lidar System		
46 Micro Pulse Lidar	50 kg	Kg.
59 Mine Detecting Set	AN-19/2 Man P	Man Portable Detector
56 Integrated GPR/EMI Mine Detection	Man-	Man-portable
8 Standoff Mine Detection Radar System	SMDRS	Current Prototype 200 lbs.
58 Thermal Netron Analysis Mine Detector	TNA Mine Detector	
28 Metallic Mine Detector	AN/PSS-12	8.5 lbs (3.8 kg) w/o case, 1
21 Close in Man Portable Mine Detector	CIMMD (ATD) 30 lbs	30 lbs for IRTI, 26 lbs for G
57 Hyperspectral Mine Detector	HWD	
68 Mutti-Sensor Towed Array Detection System		Vehicle Towed
61 Off Road Smart Mine Clearance	ORSMC	Remote HMMWV integrate
87 Unattended Ground Sensor ATD	nes	Expected - Manportable
14 Hunter Sensor (Surrogate) Suite ATD	HSS, HS3	HMMWV Integrated - Appr
13 Improved - Remotely Monitored Battle Field Sensor System (I-REMBASS AN/PSQ-7(V)		
88 Remote Sentry ATD	oedv3	Expected - Manportable
38 Lightweight Laser Designator Rangefinder (LLDR)	LLDR 30 lbs	30 lbs, broken into 2- 15 lb i
80 Hunter Sensor Surrogate Vehicle same as Magic Warrior and Night Stalk		
89 Own The Night	Includes AN/PVS-7B (NVG), AN/PAS-13 (TWS), AN/VAS-5 (DVE)	
25 Night Roamer Pocket Scope Model B	14.7 02	7 02
85 Joint Surveillance Target Attack Radar System, Light Ground Station Mo	Joint STARS, LGSM	HMMWV Portable with Trail
83 Avenger infrared System		2568 lbs (1165.87 kg) - ove
97 Dual-Mode MMW/Uncooled IR Seeker	MMW/IR SADARM (spinoff)	Lightweight
92 RAdar Plus Thermal Observation and Recognition system	RAPTOR	Lightweight, Modular

Index	Index Unit Price	Space Regs	Power Regs	Availability
22				Currently Fielded
74				Currently Fielded
41				Current
7	7 \$218,000	Man-portable, tripod mounted or shoulder fired	24 volt DC, NI-CD, Run time 10 minutes, Recharge tim Currently Fielded	Currently Fielded
72		Small		Currently Fielded
16	16\$5370	9"L×8"W×4"H	24 volts (battery or vehicle power)	Currently Fielded
02		Handheld	Battery Operated	Currently Fielded
9/				Currently Fielded
11		handheid		Currently Fielded
88		Handheld Binoculars		Prototype - Experimentation with Specail Purp
47				Current
46		Controller and Display: 49 x 23 x 51 cm, Tranceiver: 41 x	Tranceiver: 41 x 115V/60 Hz, ~5A	Curmet IR&D
29	59 \$4899.00		Dry Cell Batteries (IEC standard size LR20 or ANSI sta Current	Current
ሄ				
8		Prelim Specs: System 2 cu.ft., Antennas 6 cu. ft.	Prelim Specs: 100 watts	Prototype successfully field tested February 19
85		Metal Array -2 x 2 meter array for prototype + sensors		Prototype
28	28 \$1196	Hand Held	4 - 1.5 V batteries, 70 hour operating time	Current
21		Carried - Each system is a backpack and helmet configur		1996 Demonstration
22				
89				Prototype
61				1996 ACTD - Prototype Expected 1997
87	87 Throw Away	"Miniature"	Battery	ATD FY97 Start, 3 year funding
14		HMMVVV Integrated - Approximate - for Prototype		1998 Prototype
13		6' x 6' x 6'	Lithium or Alkaline Batteries	1996
88	88 Max<114.6K, Go		Battery Operated, Remotely Activated	ATD FY93 Start, Delivery of Prototype 1Q97,
38				Demonstrator unit scheduled for 1st quarter 19
8	80 < 500k productio		Standard Issue Batteries, Generator, and Vehicle Elect Current Prototype	Current Prototype
88				NVG Currently fielded, TWS and DVE full rate
25	25 \$1060 +	5"L x 1.75"W x 2.625" H	3 V battery, 25 hour life	Current
82		HMMWV Portable w/Trailer		Currently Fielded
83	\$617,000 (per sy	83 \$617,000 (per sy   182" L x 104" H x 87" W (4.62m x 2.64m x 2.21m) - overal		Current
26		6.3 inches squared transceiver, plus antenna which attach		Current
92				Developmental - 1996

78 N N			Ail	
101	78 Multi - IR, V	₽ V		
	10 Multi - IR, V	Det	All * Weap. Mount	Weap. Mount Texas Instruments
45 8	45 Mutti - IR, V	Det & Acq	All * Handheld	Texas instruments
8	9 Multi - IR, V	Det & Acq	Med, 7LW, SLW?	?LW, SLW? Texas Instruments, Systems Group, Defense Systems and Electronics
9	6 Multi - IR/Las, R(MTI, MM)	Acq	Med, 7LW, SLW?	?LW, SLW? Mounted Battlespace BattleLab - IEW, ASM
91₹	91 Multi - Lad, RSTA, V	& Acq	Med, 7LW. SLW?	?LW. SLW? ARPA, DOD, DOE, DOT
80€	90 Mutti - Las, N	¥	All	Dismounted Battlespace Battle Lab - Stingray
55 N	55 Multi - M, R, S	Det Coet	All	Stano
118	11 Multi - N, V	Det O		Ft. Hood/Lewis
63	63 Night Vision	Acq	All * Weap. Mount	Rapid Acquisition Targeting System
8	84 Night Vision	Acq.	All * Weap. Mount Hughes	tughes
8	30 Night Vision	Det	All * Handheld	IMO,VARO,ITT, Litton
82	29 Night Vision	Det Det	All * Handheld	ITT, Litton
27/	27 Night Vision	<b>Set</b>	Helmet Mount	Littor
32	32 Night Vision	Det & Acq	All	MO, VARO
171	17 Night Vision			Trilicon
4	44 Night Vision	Det & Acq	All • Weap. Mount	Hughes
33	33 Night Vision	Det & Acq	Weap. Mount	IMO, VARO
28	26 Night Vision	Det & Acq	All • Weap. Mount	Weap. Mount NightVision Engineering and Manufacturing
8	96 Radar	Acq	All	Thomson-CSF
936	93 Radar	Det Det	A	Thomson-CSF
18	18 Radar	Det & Acq	All	Eaton/Telephonics, AlL Systems, subsidiary of EATON
क्र	94 Radar	Det & Acq	All	Thomson-CSF
22	22 Radar	Det & Acq	Med, 7 LW, SLW?	7 LW, SLW? P.O.C. Mr Larry Bovino, DSN 995-4225, COMM 908/544-4226
8	20 Radar	Det & Acq	Med, 7LW, SLW?	7LW, SLW? Hughes Aircraft Company, Grumman upgrade with Radstone technology
191	19 Radar	Det & Acq		?LW, SLW? Hughes Aircraft Company, Grumman upgrade with Radstone technology
5	5 Radar - MW	Det	All	TRW Space and Electronics Group
4	4 Radiation	Det	All	AlL Systems, Consolidated Edison Company of NY, Industrial Quality Inc, and PMX
23(	23 UltraViolet	Det	All	Littlemore Scientific
15/	15 Visible	Det	A⊪	IMO,VARO
8	95 Visible	Det	All	Thomson-CSF

### LMDS- Chenowth

Index	Product Name	Product ID	Weight
	78 Target Designator	TD-100	Man Portable
	10 Target Acquisition System	AN/PXX-X TAS	Man Portable
	45 Nightsight 15 & 9 degree lenses	W1000, H1000	15deg: 3.1 lbs, 9deg: 3.4 lb
	9 Improved Target Acquisition System	ITAS	
	6 Target Acquisition ATD		
	91 Unmanned Ground Vehicle/Demo II	UGV/Demo II	HMMWV Integrated
	90 Target Location and Observation System	1108	Lightweight - M16 mounted
	55 Platoon Early Warning System	ANTRS-2	23 lbs
	11 Lightweight Video Reconaissance System	LVRS	Out Station (Reconaissanc
	63 RATS 635 Visible Light	RATS 635	
	84 Elcan Blackcat		Lightweight, rifle mounted
	30 Night Vision Goggles	AN/PVS-5	30 oz (.85 kg)
	29 Night Vision Goggles	AN/PVS-7B	24 oz (.68 kg)
	27 M982 /M983 Monocular with Gen II / Gen III Image intesifier tube	M982/M983	Helmet, Face Mask Mounte
	32 Crew Served Weapon Night Sight	AN/TVS-5	8 lbs (3.63 kg)
	17 Advanced Combat Optical Gunsight ACOG	TAO1, NSNM4A1	Weapon Mounted/Aluminu
	44 Stinger Night Sight	AN/PAS-18	Man Portable
	33 Individual Weapon Night Sight	AN/PVS-4	4lbs (1.818 kg)
	26 Coyole II Weapon Sight		1 lb 14 oz
	96 Flycatcher	Flycatcher	
	93 Ground Surveillance Radar	Rasit - G	Lightweight
	18 Modular Lightweight Battlefield Surveillance System	AN/PPS-5	125 lbs (currently trying to r
	94 Ground Surveillance Radar	RB 12 B	Lightweight
	22 Bistatic Radar For Weapons Location	BRWL (ATD)	Towed Reciever
	20 Firefinder Artillery Locating Radar	AN/TPQ-37	Heavy
	19 Firefinder Mortar and Artillery Locating Radar	AN/TPQ-36	Heavy
	5 Passive Millimeter-Wave Camera	PMMW Camera	
	4 Gamma Ray Imaging System (GRIS)	Gamma Cam	Man-portable
	23 763 Crawford Monitor		14 oz (390 g)
	15 M49 Telescope		2.75 lbs (1.25 kg)
	95 Mobile Optronic Surveillance System	MOS2	Lightweight

Index	Index Unit Price	Space Regs	Power Regs	Availability
78				Currently Fielded
10		Weapon Mounted		Production 3rd Querter 1997
45		15 deg: 3.8"Wx4.3"Hx12.9"L; 9 deg: 4.3"Wx4.6"Hx14.3"	9 deg: 4.3"Wx4.6"Hx14.3" nominal: 4-9 VDC Battery, operating 8 VDC - 40 VDC,	Current
6				
9				4th Quarter 1997
91				Demo II performed June 1996
8		Weapon Mounted		Fielded 1098
55	55 \$995	Fits within a carrying bag	Operates of 9 VDC batteries	Current
11		Out Station (Reconaissance Vehicle) 6"x24"x36"		1996 Prototype
ස		.48"W x .40"H x .56"L	6VDC, PX28L Battery Life	Current
8			3.1 or 6 power (VDC?)	Current
œ	30 \$5111	6.5°L x 6.8°W x 4.7°H (16.51cm x 17.27cm x 11.94cm)	2.7V Mercury Battery (BA-1567/U, BA-5567/U, BA-305 Current	Current
8	29 \$6000	5.9°L x 6.1°W x 3.9°H (14.99cm x15.49cm x 9.91cm)	Mercury NI-CD or Li 2.7 V batteries (BA-5567 or AA ce Current	Current
27	27 \$5000 +			Current
32	32 \$4,005	15"L x 6"W x 6"H (38.1cm x 15.24 cm x 15.24cm)	2.7V Mercury Battery (BA-1567/U, BA-5567/U, BA-305 Current	Current
17	17 \$706-\$1102			Current
44		Weapon Mounted		Current
33	33 \$4815	12"L x4"W x 4.5"H (30.5cm x 10.2cm x 11.4cm)	2.7V Mercury Battery (BA-1567/U, BA-5567/U, BA-305 Current	Current
58	26 \$1500 +	8.25"L x 8.25"W x 2.5"H	3 V battery, 25 hour life	Current
96				
83				Currently Fielded
18	Upgrade1/9th of	18 Upgrade 1/9th of   Carried, Tripod or Vehicle Mounted Adaptations, Modular	inted Adaptations, Modular   24 VDE - rechargeable batteries (BB-622) or external p Only Available through repair of existing - mod	Only Available through repair of existing - mod
8				
22		Towed Receiver		9661
8		Currently Fielded on an M035 series truck, M-1048 Trailer	series truck, M-1048 Trailer MEP-115A 60 kwatt, 400 hertz Generator Set	Last fielded in 1992
19		Currently Fielded on one M-35 2 1/2 ton Truck, and an M1	2 1/2 ton Truck, and an M1 2 MEP-112s 10 kwatt, 400 htz, diesel generators	Current
5				September 1998
4	4 \$50-75k	(picture in URL)		September 1995
23		7.7" x 3.9" x 1.6" (195mm x 100mm x 40mm)	Alkaline or Mercury 8 -9V PP3 type or External 7-16V	Current
15	15\$511	13.5 inches	None	Current
8				

### Armor Technology Alternatives

	l	200			
Z	Threat, Example	Type of Armor	lbs/ft	\$/ft <sup>2</sup>	thickness
class					(in)
II	357 mag. JSP,	Fiberglass composite	3	30	·/1
	9mm FMJ	Glass+Polycarbonate	7	09	11/16
		(transparent)			
IIIA	44 mag. SWC	Ballistic steel	5.25	16	8/1
		Kevlar composite	1.6	80	1/4
		Spectra composite	1.25	99	1/4
		Glass+Polycarbonate	15	06	1 1/3
		(transparent)			
III	7.62x39 FMJ,	Ballistic steel	11	33	1/4
	81mm mortar fragments	Kevlar + alumina	7	350	3/4
		Spectra composite	4.75	250	
		Glass+Polycarbonate	18.25	125	1 2/3
		(transparent)			
IV	5.56x45 FMJ,	Ballistic steel	16	48	8/8
	7.62x51 AP,	Spectra + alumina	6.5	350	
	30-06 AP,	Spectra + Boron carbide	5.5	400	_
	anti-personnel mine	Glass+Polycarbonate	25	200	3
		(transparent)			

JSP- jacketed soft point FMJ- full metal jacket SWC-semi wad cutter AP- armor piercing

RST-V Concept Study Concept/Requirements Report

### APPENDIX I

Firepower Survey Results

Category	Weapon	Caliber	Gun Weight (lb)	Range (meters)	Gun Size (in)	Comments
Personal	M16A1/A2 Rifle	5.56 mm	8.9 lb including 30 round loaded magazine	274 (this short range is the users biggest complaint)	38.9" long with flash suppresser 36.4" barrel length	700 - 900 rpm rate of fire.  Over 5 million in use.  Gas operated, optional bipod, telescope and night sight
Personal	M9 handgun	9 mm	2.6 lb (loaded)	50	8.6" long 4.92" barrel	390 m/s muzzle velocity 15 round magazine
Personal	M79 Grenade Launcher	40 mm	6 lb empty 0.61 lb projectile	400 m (max) 350 m (area targets) 150 m (point targets)	29" long 14" barrel	76 m/sec muzzle velocity. Single shot. Replaced by M203 grenade launcher.
Personal	M203 Grenade Launcher	40 mm	3 lb launcher	350 m (area) 150 m (point) 400 m max		Attaches to the M16 rifle. Approx cost = \$600.
Personal	MP-5N	6 mm	7.44 lb with a 30 round mag	100 m	26" extended	For special purpose forces.  800 rpm  Approx cost = \$800.
Personal	M249 Squad Automatic Weapon (SAW)	5.56 mm	15.5 lb empty 22 lb with 200 round magazine	1,300 m	39.4" long 18.5" barrel	750 rpm. 924 m/s muzzle velocity

-	Weapon	Caliber	Gun Weight (lb)	Range (meters)	Gun Size (in)	Comments
M6(	M60 Machine	7.62 mm	23 lb with bipod	900 (boqid)	43.3" long	550 rpm cyclic rate, 200 rpm automatic.
<b>5</b>	_		39.6 lb with	1,800 (tripod)	22" barrel	Gas operated.
			tripod			Uses 100 round ammo belts.
	•					Barrel changed after 500 rounds.
						Can mount on vehicles with the M142 mount.
						Being replaced by the M240G.
M2,	M240G	7.62 mm	24.2 lb	1,800 m (tripod	47.5" long	650 - 950 rpm cyclic rate of fire.
X X X	Medium Machine Gun			mounted)		100 rpm sustained.
	•					Approx $cost = $6.6K$
						Can be pintle mounted for vehicular applications.
M2	M2 HB	12.7 mm (0.5")	83.3 lb	6,830m max	65.07" long	Gas operated.
Mar	Machine Gun		127.8 lb with tripod		45" barrel	Fires ball, tracer, armor piercing and incendiary rounds.
MK	MK19-3	40 mm	72.5 lb	1,500 m (point	40.4" long	350 rpm, gas operated.
S S	Grenade			targets)		240 m/sec muzzle velocity.
Lau	Launcher			2,200 m (area targets)		5 m casualty radius.
						Can penetrate 2" of armor at 2,200 m range.

134

Category	Weapon	Caliber	Gun Weight (lb)	Range (meters)	Gun Size (in)	Comments
Primary	M29A1	81 mm	28 lb barrel	4,700 m		40° to 85° elevation.
	mortar		25 lb baseplate			+/- 4° traverse.
			40 lb bipod			30 rpm for one minute, 4 - 12 rpm
			115 lb total with			sustained.
			sight			Being replaced by the M252.
Primary	M252 Mortar	81 mm	35 lb assembly	5,700 m		33 rpm max rate of fire, 16 rpm sustained.
			26 lb bipod			45° to 85° elevation.
			25.5 lb			Approx $cost = $25K$ .
			baseplate			
			2.5 lb sight			
			89 lb total			
Primary	M224	60 mm	46 lb total	3,500 m		Indirect fire weapon.
	Lightweight Mortar					Replaced the M29 in non-mechanized infantry. Often used with the AN/GVS-5 hand held laser range finder.
						30 rpm maximum, 20 rpm sustained.
						Approx cost = \$11K.
Missile	M72A2 Light	96 mm	4.75 lb complete	325 m	20" rocket	145 m/sec muzzle velocity.
Launchers	Anti-Tank Weapon (LAW)		2.2 lb rocket.		length 25.7" launcher	Shoulder fired. HEAT round can penetrate 11.8" of armor.
					35" extended	Being replaced by the AT-4 light weight multi-purpose weapon.

Weapon	uo	Caliber	Gun Weight (lb)	Range (meters)	Gun Size (in)	Comments
Lightweight Multi-Purpose Weapon, AT4	Δ.	84 mm	14.6 lb complete	300 m	39.7" long	Single shot, throwaway, squad anti-tank rocket. 290 m/s muzzle velocity Getting old, cannot defeat the latest Soviet
Advanced Anti- Tank Weapon System-Medium	-‡ - <u></u>		45 lb complete 32 lb missile	2,000 m		armor. Fire and forget anti-tank missile. Imaging infra-red seeker.
FIM-43A Redeye			29 lb complete 18 lb missile	3,300 m	48" L x 2.75" dia	Shoulder fired, infantry surface-to-air missile. Initial optical aiming, IR homing. Mach 2.5 flight speed. No IFF, must wait for aircraft to attack, then fire at their exhaust. Guidance is vulnerable to IRCM. Flight speed is just enough to catch
FIM-92A Stinger			35 lb complete 24 lb missile	5,000 m	60" missile x 2.75" diameter	Portable air defense missile.  Passive IR homing, includes IFF.  Mach 2 flight speed.  Replaced the Redeye.  Used in HMMWVs, carries 8 missiles.

Category	Weapon	Caliber	Gun Weight (lb)	Range (meters)	Gun Size (in)	Comments
Missile Launchers	TOW-2B BGM-71		46.1 lb launch weight	3,750 m	45" - 75" long x 6" diameter	Heavy anti-tank missile. Used in HMMWVs, carries 6 missiles. 1003 km/hr flight speed. Optically tracked (day or night)
Missile Launchers	Dragon M47 FGM-77A		24.4 lb launch weight 5.4 lb shaped charge warhead	1,000 m	29.3" long x 4.5" diameter	Infantry anti-tank/assault missile.  370 km/hr flight speed.  Designed as a medium range complement to the TOW.  Throw-away launch tube.
Missile Launchers	Shoulder Launched Multipurpose Assault Weapon (SMAW)	83 mm	16.6 lb carry weight 30.5 lb ready to fire	250 m (1 x 2 m target) 500 m (tank sized target)	29.9" carry length 54" ready to fire	Unit cost approx \$13K.  Consists of the MK153 launcher, MK3 encased HEDP rocket, MK6 encased HEAA rocket and the MK217 spotting rifle cartridge. Uses the MK42 day sight and the AN/PVS-4 night sight.

### Notes:

- Deadeye-40/50 is a remote controlled, stabilized mount for use with the MK19 and M2 HB weapons. It uses 24Vdc, is Mil-Std-1275 compatible, requires 100 watts for average operation (50 watts standby) and has day and night sight capabilities. Max dimensions are 65.5" long x 32" wide and weighs 243 pounds. Weight is 439 pounds including the M2 HB machine gun and ammo.
- The MK93 is a dual purpose, soft mount for the MK19 Grenade Launcher and M2 HB Machine Gun. The MK93 is designed for use with either a tripod change-overs much easier. The use of a soft mount improves the accuracy of the M2 Machine Gun by attenuating the recoil. The MK93 consists of a carriage and cradle assembly, train stop bracket, ammunition can holder, a bolt-on small pintle, a bolt-on large pintle, and stowage bar assembly. It or a vehicular mount (using the MK175 pintle pedestal). The MK93 requires no external adapters or tools to mount either weapon system, making weighs 30 pounds, measures 6.8" high x 10" wide x 22" long and costs about \$3.2K. ri
- 3. A typical RSTA mission would most likely employ the following weapons:

Personal: Each person wearing an M9 handgun on their belt, an M16 machine gun with an M203 attachable grenade launcher in readily accessable

Secondary: The vehicle will have M240G machine guns mounted on both right and left doorposts.

Primary. The vehicle will most likely require a universal mount to accomodate either the MK19 or the M2 HB machine gun.

### MINES

are unfortunately in wide spread use. The fuse may incorporate a tripwire, an anti-handling device or some form of electronic sensor. Anti-personnel mines Anti-personnel mines are designed to kill or maim personnel who happen to walk over the buried device. They are very inexpensive (\$3 to \$30 a piece) and are equipped with 10-250 g of explosive, and detonate under pressures ranging from 0.5 to 50 kg. There are roughly 360 types of anti-personnel mines available, produced in some 55 countries, and can be divided into two (more or less) distinct groups: blast mines and fragmentation mines.

as the mine casing or surrounding dirt or gravel is blasted at the victim. Modern anti-personnel blast mines have a plastic watertight casing; only the detonator, Blast mines are surface or sub-surface laid, activated by pressure and rely mainly on blast for their effectiveness. Secondary fragmentation injuries are possible springs and strikers are metal, making them difficult to detect and clear.

When a fragmentation mine explodes, shrapnel, pre-cast fragments or steel balls are projected outwards over a given radius (10-50 m) at approximately twice the velocity of a bullet, producing puncture wounds, blinding and death. Fragmentation mines can be divided into three sub-groups.

- (a) Stake mines: in order to increase the shrapnel effect, these mines are mounted on stakes to raise them above the ground
- (b) Directional fragmentation mines: these mines fire small fragments in a predetermined direction. The most notorious of these is the Claymore mine.

(c) Bounding mines: a small propelling charge first lifts the mine to about stomach height before the main charge explodes, making it much more lethal (the kill rate for those triggering the mine is 100%) and shooting the fragments over a much wider area.

against there use. First, he directed the U.S. military immediately stop using "dumb" anti-personnel land mines -- those that remain active until detonated or these mines will save the lives of U.S. service members in some battlefield situations. Third, the president directed DoD to examine tactical alternatives to than 4 million anti-personnel mines by 1999. Second, the United States reserves the right to use "smart" anti-personnel mines, which self-destruct, because cleared. The only exception is on the Korean Peninsula, where he judged the risk to U.S. interests was too great. U.S. forces will rid the arsenal of more It is unlikely that any LSV applications would be required to carry and deploy anti-personnel mines as President Clinton announced four unilateral actions mines. Finally, he directed DoD to expand efforts to develop better mine-detection and mine-clearing technology.

to civilians than anti-personnel mines because they are not usually detonated by pedestrians. A typical anti-tank mine uses 5.4 kg (12 lb) of TNT and is buried Anti-tank mines usually contain between 2 and 9 kg of explosive and are activated by pressures between 100 and 300 kg. Anti-tank mines are less dangerous a few inches below the surface. Modern anti-tank mines are triggered by seismic fuses which can differentiate between fried or foe signatures.

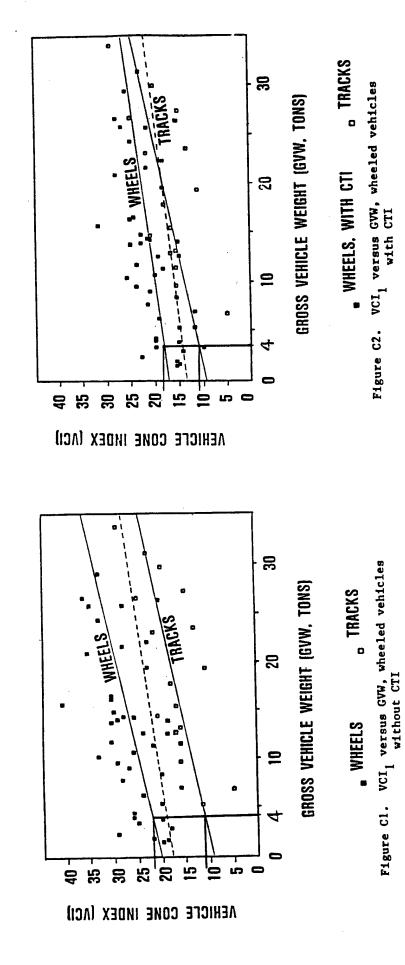
### State of the art mine technology includes:

- quantities from aircraft or helicopter underwing pods or from artillery, mortar or rocket warheads. Most scatterable mines are small and many are intended to be dual-purpose anti-personnel/anti-tank devices. One vehicle-mounted mine-scattering system dispenses up to 1,750 anti-personnel mines per minute, Remotely delivered mines - First used on a large scale in Viet Nam, remotely delivered mines (also known as scatterable mines) are strewn in large while a helicopter-mounted system, is designed to drop 2,080 anti-personnel mines in three to 16 minutes. Unless they are self-destructing or selfneutralizing, it is almost impossible to use them in a manner which complies with Protocol II (Clinton's statement above).
- held by some to be a viable alternative to the mapping and fencing of minefields. However, they have not yet shown sufficient reliability under battlefield conditions. To provide acceptable protection for the civilian population, the reliability rate of self-destructing and self-neutralizing mines must be 99.6%, mines», as opposed to the traditional «dumb» or conventional mines. Designed to self-destruct or self-neutralize after a predetermined period, they are Self-neutralization and self-destruction mechanisms - Mines whose fuses incorporate these types of mechanism are commonly referred to as «smart hat is, the level achieved by humanitarian mine-clearance operations. 'n
- Anti-handling and anti-disturbance devices Mines fitted with simple tilt switches which make them explode as soon as they are tilted or moved are already being produced. Thus if a prodder hits the mine during a clearance operation, the mine explodes in the mine-clearers face. Other fuses are designed to detonate the mine if an electronic mine detector is passed over them, or if they are exposed to daylight. ₩.

### APPENDIX J

VCI Comparison - Tracks vs. Wheels

# Tracked vs Wheeled Tradeoff - VCI Comparison; Tracks vs Wheels



APPENDIX K

VCI Calculations

## Tracked vs Wheeled Tradeoff - VCI Calculations

# Tracked vs Wheeled Tradeoff - VCI Sample Calculations

### Surrogate Fast Attack Vehicle has a VCI = 22 (1)

$$MI = \left[ \left( \begin{array}{cccc} \frac{3500}{11.5 \times 15 \times 4} & \times & 0.9 \\ \hline 1.25 \times 11.5 & \times & 1.00 \\ \hline 1.05 \times 11.5 & \times & 1.00 \end{array} \right. + \frac{2.75}{4} - \frac{11.75}{10} \right) 1.0 \times 1.05 \right] + 20 = 52.8$$

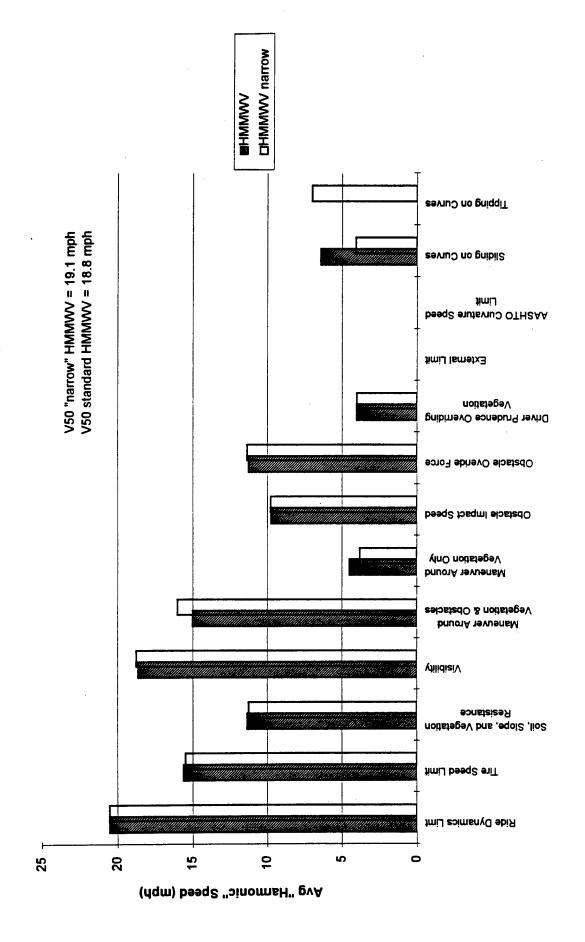
VCI = 
$$\left(11.48 + 0.2 \times 52.8 - \frac{39.2}{52.8 + 3.74}\right) \left(\frac{.15}{.10}\right)^{.25} = 23.6 \,^{(2)}$$

- (1) Results of Mobility Tests on the Surrogate Fast Attack Vehicle (Technical Report GL-84-9, Table 1, R. Gillespie, September 1984)
  - (2) Calculations are Conservative

### APPENDIX L

Initial NRMM Results

### NRMM Study (Lauterbach Terrain)



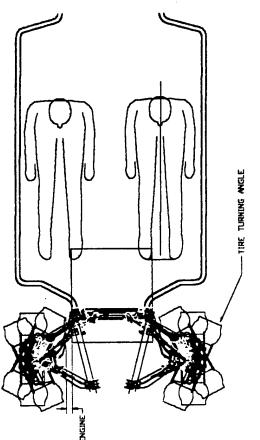
L-3

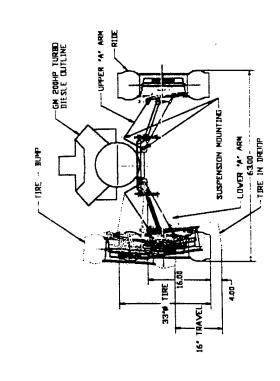
### APPENDIX M

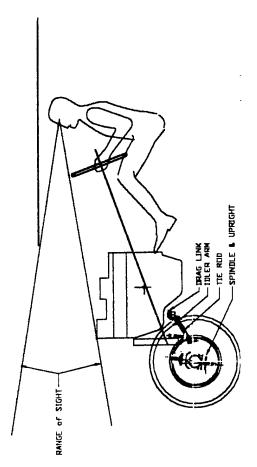
Suspension / Engine Relationships

### Subsystem Concept Development Mobility - Standard Suspensions

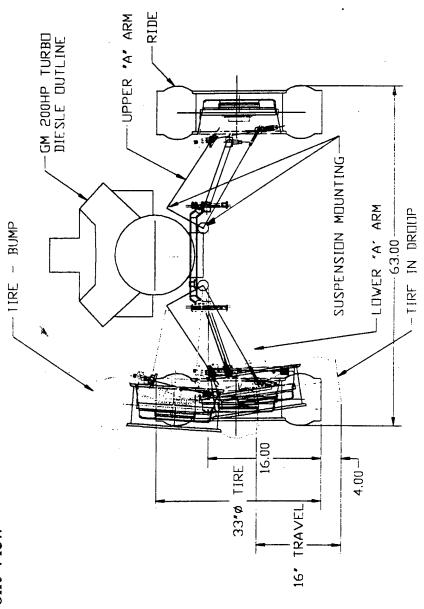
Suspension/Engine Relationships 2000 THE TO ENGINE Front, Plan & Side Views



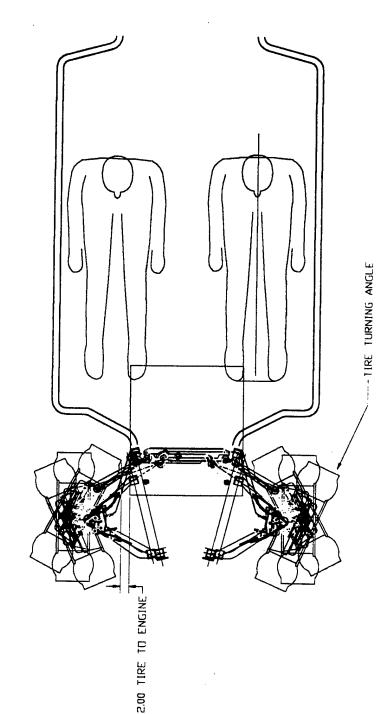




### Subsystem Concept Development Mobility - Standard Suspensions Suspension/Engine Relationships Front View



### Subsystem Concept Development Mobility - Standard Suspensions Suspension/Engine Relationships Plan View

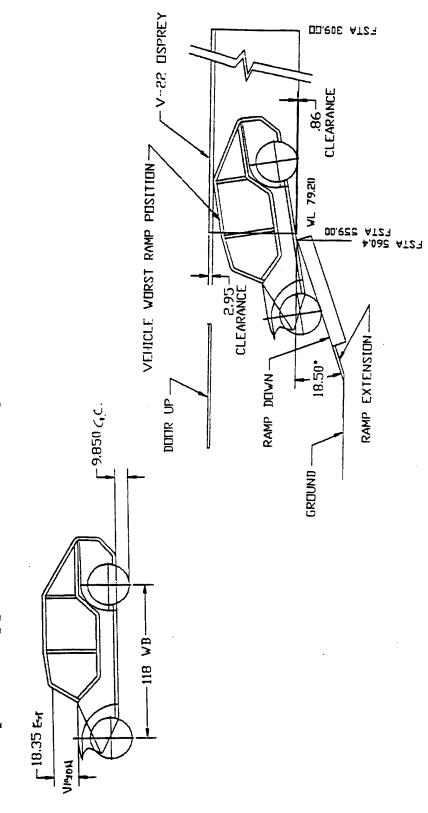


### APPENDIX N

Wheel Base Assessment

## Initial Vehicle Assessment - Wheel Base

• 118" Wheel Base was chosen to be compatible with our retractable suspension approach based on 10" ground clearance

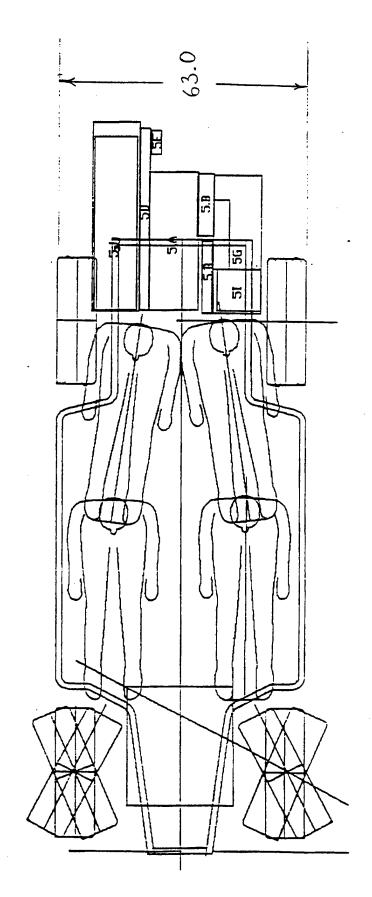


RST-V Concept Study Concept/Requirements Report

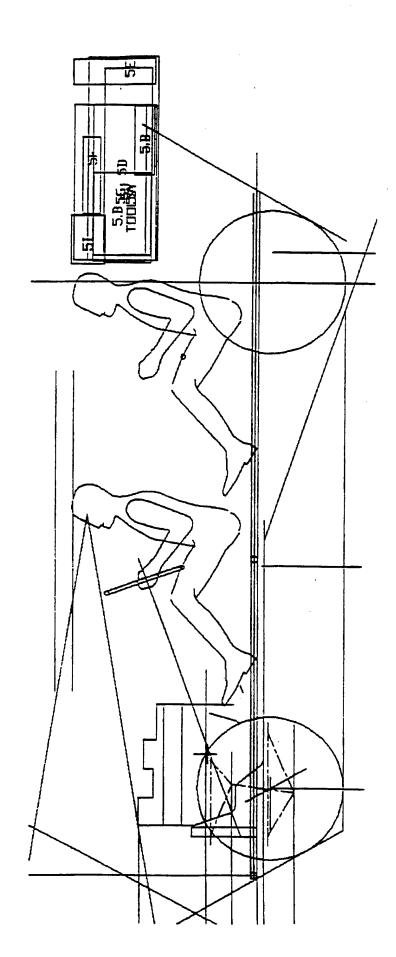
**APPENDIX O**RST-V Variants

## Initial Vehicle Assessment - Seating Capacity

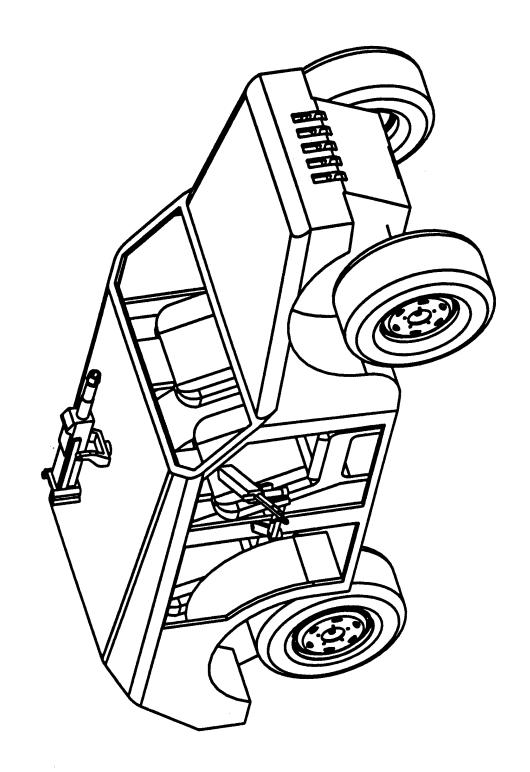
• 4 Man crew is too tight at 118" wheel base. Wheel base would have to be increased to about 142" minimum to allow for room of the back seat passengers.



## Initial Vehicle Assessment - Seating Capacity

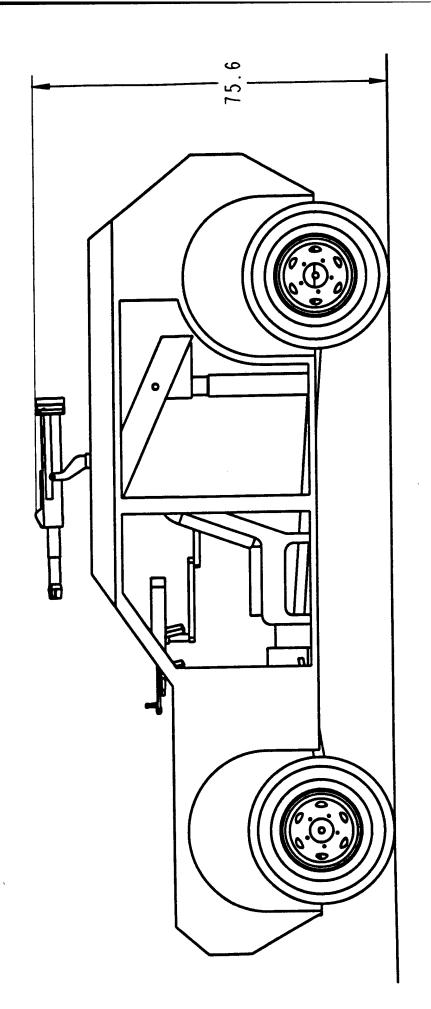


Subsystem Concept Development Preliminary Vehicle Concept Models

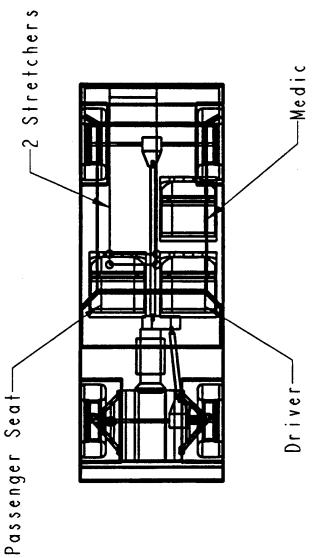


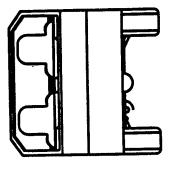
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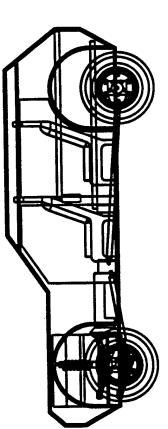
### Subsystem Concept Development Preliminary Vehicle Concept Models



### Subsystem Concept Development Preliminary Vehicle Concept Models







### APPENDIX P

Detailed Vehicle Weight Breakdown

### APPENDIX P DETAILED VEHICLE WEIGHT BREAKDOWN

	/Atem	Oty.	Weight Each (Lbs)	Total Weight (1
TY VEHICLE (	Same for all Mission Variants)			
<u>Hull</u>				
Chassis /	Assembly	1	315.0	315.0
Body		1	120.0	120.0
Dash		1	10.0	10.0
* Front Bu	mper	1	15.0	15.0
Rear Bur	nper	1	38.0	38.0
Front Tre	ms. Skid Plate	1	3.0	3.0
Transmis	ssion Skid Plate	1	8.0	8.0
Engine S	kid Plate	1	11.0	11.0
Floor Par	n, Front	1	16.0	16.0
Floor Par	n, Rear	1	16.0	16.0
Front Du	st Wall	1	10.0	10.0
Side Basi	ket	0	29.0	0.0
Gas Tani	k	2	15.0	30.0
Straps, T	ank Holddown	4	2.0	8.0
Seat, Dri	ver & Navigator	2	22.0	44.0
Seat Adj	uster	3	3.0	9.0
Seat Belt	Assembly	3	4.0	12.0
Passenge	er Seat	1	22.0	22.0
Clutch Po	edal Assembly w/ Cyls.	0	4.0	0.0
Gas Peda	ıi .	1	1.0	1.0
Brake Pe	edal w/ Master Cyls.	1	8.0	8.0
Headligh	nt	4	2.5	10.0
Blackout	Lights	4	3.0	12.0
Hand Br	ake Assembly  Hull Weight	1	4.0	722.0
			20.0	80.0
	ront (Coil-over)	4		
Shock R	ear	4	20.0	80.0
Shock Re Front Co	ear vil Springs	4	20.0 5.0	<b>8</b> 0.0 <b>2</b> 0.0
Shock Re Front Co Rear Coi	ear ill Springs il Springs	4 4	20.0 5.0 5.0	80.0 20.0 20.0
Shock Re Front Co Rear Coi Trailing	ear vil Springs il Springs Arm, Rear	4 4 4 2	20.0 5.0 5.0 26.0	80.0 20.0 20.0 52.0
Shock Re Front Co Rear Coi Trailing Tie Rod	ear  il Springs il Springs Arm, Rear (set)	4 4 4 2 1	20.0 5.0 5.0 26.0 6.0	80.0 20.0 20.0 52.0 6.0
Shock Ro Front Co Rear Coi Trailing Tie Rod A-Arm S	ear  il Springs il Springs Arm, Rear (set) Set	4 4 4 2 1 2	20.0 5.0 5.0 26.0 6.0 34.0	80.0 20.0 20.0 52.0 6.0 68.0
Shock R Front Co Rear Co Trailing Tie Rod A-Arm S A-Arm F	ear  il Springs il Springs Arm, Rear (set) Set Pivot Bushing Assembly	4 4 4 2 1 2 8	20.0 5.0 5.0 26.0 6.0 34.0 0.5	80.6 20.0 20.0 52.0 6.0 68.0 4.0
Shock R Front Co Rear Coi Trailing Tie Rod A-Arm S A-Arm F Spindle	ear  il Springs il Springs Arm, Rear (set) Set Pivot Bushing Assembly Assembly (set)	4 4 4 2 1 2 8 2	20.0 5.0 5.0 26.0 6.0 34.0 0.5	80.0 20.0 20.0 52.0 6.0 68.0 4.0 20.0
Shock R Front Co Rear Coi Trailing Tie Rod A-Arm S A-Arm F Spindle & Bushing	ear  il Springs il Springs Arm, Rear (set) Set Pivot Bushing Assembly Assembly (set) , IRS	4 4 4 2 1 2 8 2 4	20 0 5.0 5.0 26.0 6.0 34.0 0.5 10.0	80.0 20.0 20.0 52.0 6.0 68.0 4.0 20.0
Shock R. Front Co Rear Coi Trailing Tie Rod A-Arm S A-Arm F Spindle . Bushing, Sleeve, I	ear  il Springs il Springs Arm, Rear (set) Set Pivot Bushing Assembly Assembly (set) , IRS IRS Bushing	4 4 4 2 1 2 8 2 4 2	20.0 5.0 5.0 26.0 6.0 0.5 10.0 0.3 0.5	80.0 20.0 20.0 52.0 6.0 4.0 20.0 1.0
Shock R. Front Co Rear Coi Trailing Tie Rod A-Arm F Spindle . Bushing, Sleeve, I Wheel B	ear  iil Springs iil Springs Arm, Rear (set) Set Pivot Bushing Assembly Assembly (set) , IRS IRS Bushing Learing Assembly, Front	4 4 4 2 1 2 8 2 4 2 2	20 0 5.0 5.0 26.0 6.0 34.0 0.5 10.0 0.3 0.5 9.5	80.0 20.0 20.0 52.0 6.0 68.0 4.0 1.0 1.0
Shock R. Front Co Rear Coi Trailing Tie Rod A-Arm F Spindle . Bushing, Sleeve, I Wheel B	ear  il Springs il Springs Arm, Rear (set) Set Pivot Bushing Assembly Assembly (set) , IRS IRS Bushing	4 4 4 2 1 2 8 2 4 2	20.0 5.0 5.0 26.0 6.0 0.5 10.0 0.3 0.5	80.0 20.0 20.0 52.0 6.0 68.0 20.0 1.0 19.0
Shock Ro Front Co Rear Coi Trailing Tie Rod A-Arm S A-Arm F Spindle Bushing Sleeve, I Wheel B	ear  il Springs il Springs Arm, Rear (set) Set Pivot Bushing Assembly Assembly (set) , IRS IRS Bushing Bushing Assembly, Front Bearing Assembly, Rear	4 4 4 2 1 2 8 2 4 2 2	20 0 5.0 5.0 26.0 6.0 34.0 0.5 10.0 0.3 0.5 9.5	80.0 20.0 20.0 52.0 6.0 68.0 20.0 1.0 19.0
Shock Referent Co Rear Coi Trailing Tie Rod A-Arm F Spindle Bushing, Sleeve, I Wheel B Wheel B	ear  oil Springs il Springs Arm, Rear (set) Set Pivot Bushing Assembly Assembly (set) , IRS IRS Bushing Learing Assembly, Front learing Assembly, Rear ont	4 4 4 2 1 2 8 2 4 2 2 2 2	20 0 5.0 5.0 26.0 6.0 34.0 0.5 10.0 0.3 0.5 9.5	80.0 20.0 20.0 52.0 68.0 4.0 1.0 1.0 19.0 99.0
Shock Referent Co Rear Coi Trailing Tie Rod A-Arm S A-Arm S Spindle a Bushing, Sleeve, I Wheel B Wheel B Tire, Fro	ear  oil Springs il Springs Arm, Rear (set) Set Pivot Bushing Assembly Assembly (set) , IRS IRS Bushing learing Assembly, Front learing Assembly, Rear ont ar	4 4 4 2 1 2 8 2 4 2 2 2 2	20.0 5.0 5.0 26.0 6.0 34.0 0.5 10.0 0.3 0.5 9.5 9.5	80.0 20.0 20.0 52.0 68.0 4.0 20.0 1.0 19.0 99.0
Shock R. Front Co Rear Coi Trailing Tie Rod A-Arm F Spindle . Bushing, Sleeve, I Wheel B Wheel B Tire, Fro Tire, Re.	ear  iil Springs iil Springs Arm, Rear (set) Set Pivot Bushing Assembly Assembly (set) , IRS IRS Bushing Bearing Assembly, Front Bearing Assembly, Rear  ont ar Front (Al)	4 4 4 2 1 2 8 2 4 2 2 2 2 2 2 2	20.0 5.0 5.0 26.0 6.0 0.5 10.0 0.3 0.5 9.5 9.5 49.5	80.0 20.0 20.0 52.0 68.0 4.0 20.0 1.0 19.0 99.0 99.0
Shock R. Front Co Rear Coi Trailing Tie Rod A-Arm S A-Arm S Spindle . Bushing, Sleeve, 1 Wheel B Wheel B Tire, Fro Tire, Re; Wheel, I	ear  pil Springs  il Springs  Arm, Rear (set)  Set  Privot Bushing Assembly  Assembly (set)  IRS  IRS Bushing Bearing Assembly, Front Bearing Assembly, Rear  pont  ar  Front (Al)  Rear (Al)	4 4 4 2 1 2 8 2 4 2 2 2 2 2 2 2 2 2 2	20.0 5.0 5.0 26.0 6.0 34.0 0.5 10.0 0.3 0.5 9.5 9.5 49.5 49.5	80.0 20.0 20.0 52.0 6.0 6.0 4.0 1.0 19.0 19.0 99.0 48.0
Shock Referent Co Rear Coi Trailing Tie Rod A-Arm F Spindle Bushing, Sleeve, I Wheel B Wheel B Tire, Fro Tire, Rei Wheel, I Wheel, I	ear  iil Springs iil Springs Arm, Rear (set) Set Pivot Bushing Assembly Assembly (set) , IRS IRS Bushing Bearing Assembly, Front Bearing Assembly, Rear  ont ar Front (Al)	4 4 4 2 1 2 8 8 2 4 4 2 2 2 2 2 2 2 2 2	20 0 5.0 5.0 26.0 6.0 34.0 0.5 10.0 0.3 0.5 9.5 9.5 49.5 49.5 24.0 24.0	80.0 20.0 20.0 52.0 6.0 68.0 4.0 19.0 19.0 99.0 48.0 0.0
Shock Referent Co Rear Coi Trailing Tie Rod A-Arm F Spindle & Bushing, Sleeve, I Wheel B Wheel B Tire, Fro Tire, Rei Wheel, I Wheel, I Inner Tu	ear pil Springs il Springs Arm, Rear (set) Set Pivot Bushing Assembly Assembly (set) , IRS IRS Bushing Learing Assembly, Front Learing Assembly, Rear point ar Front (Al) Rear (Al) Libes, Front Libes, Rear	4 4 4 2 1 2 8 2 4 2 2 2 2 2 2 2 2 2 2	20 0 5.0 5.0 26.0 6.0 34.0 0.5 10.0 0.3 0.5 9.5 9.5 49.5 49.5 24.0 24.0 4.0	80.0 20.0 20.0 52.0 6.0 68.0 4.0 19.0 19.0 99.0 99.0 48.0 0.0
Shock Referent Co Rear Coi Trailing Tie Rod A-Arm F Spindle Bushing, Sleeve, I Wheel B Wheel B Tire, Fro Tire, Rei Wheel, I Wheel, I Inner Tu	ear  sil Springs il Springs Arm, Rear (set) Set Pivot Bushing Assembly Assembly (set) , IRS IRS Bushing learing Assembly, Front learing Assembly, Rear ont ar Front (Al) Rear (Al) libes, Front libes, Rear teering Sys. Lines	4 4 4 2 1 2 8 2 2 2 2 2 2 2 2 2 0 0	20.0 5.0 5.0 26.0 6.0 34.0 0.5 10.0 0.3 0.5 9.5 9.5 49.5 49.5 24.0 24.0 4.0	80.0 20.0 20.0 52.0 6.0 68.0 4.0 10.0 19.0 19.0 99.0 48.0 0.0 0.0
Shock Right Front Co Rear Coi Trailing Tie Rod A-Arm S A-Arm S Spindle . Bushing, Sleeve, I Wheel B Wheel B Tire, Fro Tire, Rei Wheel, I Inner Tu Inner Tu Inner Tu Power S Steering	ear  iil Springs iil Springs Arm, Rear (set) Set Pivot Bushing Assembly Assembly (set) , IRS IRS Bushing learing Assembly, Front learing Assembly, Rear ont ar Front (Al) Rear (Al) libes, Front libes, Front libes, Front libes, Rear libering Sys. Lines Box	4 4 4 2 1 2 8 2 2 2 2 2 2 2 2 2 0 0	20.0 5.0 5.0 26.0 6.0 34.0 0.5 10.0 0.3 0.5 9.5 9.5 49.5 24.0 24.0 4.0 3.0	80.0 20.0 20.0 52.0 6.0 68.0 1.0 1.0 19.0 99.0 48.1 48.0 3.3
Shock Reserved to the control of the	ear  iil Springs iil Springs Arm, Rear (set) Set Privot Bushing Assembly Assembly (set) , IRS IRS Bushing Bearing Assembly, Front Bearing Assembly, Rear  ont ar Front (Al) Rear (Al) Bes, Front Beering Sys. Lines Box Beering Sys. Lines Box Beering (continued)	4 4 4 2 1 2 8 2 2 2 2 2 2 2 2 2 0 0	20.0 5.0 5.0 26.0 6.0 34.0 0.5 10.0 0.3 0.5 9.5 9.5 49.5 24.0 24.0 4.0 3.0	80.0 20.0 20.0 52.0 6.0 68.6 4.0 1.0 19.0 99.0 99.0 48.0 0.0 3.1
Shock Referent Co Rear Coi Trailing Tie Rod A-Arm F Spindle A-Bushing, Sleeve, I Wheel B Wheel B Tire, Fre Wheel, I Wheel, I Inner Tu Inner Tu Power S Steering	ear  sil Springs il Springs Arm, Rear (set) Set Pivot Bushing Assembly Assembly (set) , IRS IRS Bushing Bearing Assembly, Front Bearing Assembly, Front Bearing Assembly, Rear out ar Front (Al) Rear (Al) Bes, Front Bes, Rear Bearing Sys. Lines Box Box Beering (continued) , Steering	4 4 4 2 1 2 8 2 4 2 2 2 2 2 2 2 2 0 0 1 1	20.0 5.0 26.0 6.0 34.0 0.5 10.0 0.3 0.5 9.5 49.5 49.5 24.0 24.0 4.0 3.0 35.0	80.0 20.0 20.0 52.0 6.0 68.0 4.0 10.0 19.0 19.0 99.0 48.0 0.0 3.0 35.0
Shock Referent Co Rear Coi Trailing Tie Rod A-Arm F Spindle A Bushing, Sleeve, I Wheel B Wheel B Tire, Fro Tire, Rei Wheel, I Wheel, I Inner Tu Inner Tu Power S Steering	ear  sil Springs il Springs Arm, Rear (set) Set Pivot Bushing Assembly Assembly (set) , IRS IRS Bushing Assembly, Front Bearing Assembly, Front Bearing Assembly, Rear ont ar Front (Al) Rear (Al) Bushes, Front Bushes, Front Bushes, Rear Reering Sys. Lines Box  Reering (continued) , Steering Shaft	4 4 4 4 2 1 2 8 2 2 2 2 2 2 2 2 2 0 0 0 1 1	20.0 5.0 5.0 26.0 6.0 34.0 0.5 10.0 0.3 0.5 9.5 49.5 24.0 24.0 4.0 4.0 3.0 35.0	80.0 20.0 20.0 52.0 6.0 68.0 4.0 10.0 19.0 19.0 99.0 48.0 0.0 3.3 35.0
Shock Referent Co Rear Coi Trailing Tie Rod A-Arm F Spindle Bushing Sleeve, I Wheel B Wheel B Tire, Fro Tire, Rei Wheel, I Unner Tu Inner Tu Inner Tu Power S Steering Suspension & Si Coupler Steering Ball Join	ear  il Springs il Springs Arm, Rear (set) Set Pivot Bushing Assembly Assembly (set) , IRS IRS Bushing learing Assembly, Front learing Assembly, Rear out ar Front (Al) Rear (Al) labes, Front labes, Rear teering Sys. Lines (Box  Lines) (Box	4 4 4 4 2 1 2 8 8 2 4 4 2 2 2 2 2 2 2 2 2 2 1 1 1 1 1 1 1	20 0 5.0 5.0 26.0 6.0 34.0 0.5 10.0 0.3 0.5 9.5 9.5 49.5 24.0 24.0 4.0 4.0 3.0 35.0	80.0 20.0 20.0 52.0 68.0 4.0 10.0 19.0 99.0 99.0 48.1 48.0 0.0 3.0 3.5 48.0 1.0 2.0 48.0 48.0 48.0 48.0 48.0 48.0 48.0 48
Shock Referent Co Rear Coi Trailing Tie Rod A-Arm F Spindle Bushing Sleeve, I Wheel B Wheel B Tire, Fro Tire, Rei Wheel, I Unner Tu Inner Tu Power S Steering Suspension & Si Coupler Steering Ball Join Steering	ear  sil Springs il Springs Arm, Rear (set) Set Pivot Bushing Assembly Assembly (set) , IRS IRS Bushing Assembly, Front Bearing Assembly, Front Bearing Assembly, Rear ont ar Front (Al) Rear (Al) Bushes, Front Bushes, Front Bushes, Rear Reering Sys. Lines Box  Reering (continued) , Steering Shaft	4 4 4 4 2 1 2 8 8 2 4 4 2 2 2 2 2 2 2 2 2 0 0 0 1 1 1	20.0 5.0 5.0 26.0 6.0 34.0 0.5 10.0 0.3 0.5 9.5 9.5 49.5 24.0 4.0 3.0 35.0	80.0 20.0 20.0 52.0 6.0 68.0 4.0 10.0 19.0 19.0 99.0 48.0 0.0 3.3 35.0

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### RST-V Concept Study Concept/Requirements Report

Category/Item	Oty.	Weight Each (Lbs)	Total Weight (L.
ngine (Standard Propulsion System)			
Engine Assembly	1	687.0	687.0
Engine Accessories	1	90.2	90.2
Radiator	1	35.0	35.0
Radiator Hose	2	5.5	11.0
Fan Assembly	1	16.0	16.0
Oll Cooley Assembly			0.0 8.0
Oil Cooler Assembly Oil Filter	1	8.0	8.0 2.0
	1	2.0	2.0
Oil Filter Mount	1	1.0	1.0 0.0
Air Cleaner Assembly	i	7.0	7.0
Air Intake Box	1	2.0	2.0
Air Intake Screen	1	3.0	3.0
Hose, Airbox to Air Cleaner	1	3.0	3.0
Engine Weight Subto	otal		865.2
instanta (Standard Branchica Coston)			
ivetrain (Standard Propulation System) Transaxle	1	172.0	172.0
Front Differential	1	172.0	135.0
Driveshaft	1	15.0	15.0
CV Joint, Front Outer	2	8.0	16.0
CV Joint, Prom Outer  CV Joint, Rear Outer	2	8.0	16.0
	2	3.0	6.0
CV Joint, Front Inner (Turbo)	_	•	6.0
CV Joint, Rear Inner (Turbo)	2 2	3.0	16.0
Drive Axles, Front	_	8.0	8.0
Drive Axles, Rear	2	4.0	
CRP Shifter Assembly	1	4.0	4.0
Caliper w/ Pads, Front	2	3.5	7.0
Caliper w/ Pads, Rear	2	3.5	7.0
Brake Rotor, Front	2	13.0	26.0
Brake Rotor, Rear	2	13.0	26.0
Park Brake Caliper	2	2.0	4.0
Drivetrain Weight Subto	ial		464.0
brid Electric Drive System - TBD			
Hybrid Electric Drive System Weight Subto	ıal		N/A
ilt-in (Integral) Armor (Mine Attack)			
Built-in Armor Weight Subto	ial		1047.0
ectrical Systems			
Battery	2	34.0	68.0
Wire Harness	1	35.0	35.0
Alternator, Heavy-Duty	1	75.0	75.0
Electrical Systems Weight Subio		73.0	178.0

### RST-V Concept Study Concept/Requirements Report

Category/Item	Qtv.	Weight Each (Lbs)	Total Weight (L.
-VEHICLE BASIC EQUIPMENT (Same for all Mis	sion Variant	is)	
Communications/Navigation Equipment			
VHF radio (Transceiver, antenna, kit)	1	75.0	75.0
Intercom Harness & Control Boxes	. 1	16.5	16.5
Cables	1	10.0	10.0
Global Position System w/Antenna	1	4.4	4.4
Communications/Navigation Eqpt. Weight Sui	btotal		105.9
Controls & Displays			
Speedometer	1	1.0	1.0
Oil Pressure Gauge	1	0.5	0.5
Voltage Gauge	1	0.5	0.5
Water Temp. Gauge	1	0.5	0.5
Trans. Temp. Gauge	1	1.0	1.0
Hour Meter	i	1.0	1.0
Turn Signal Indicator	1	1.0	1.0
Compass	1	2.0	2.0
Fuel Gauge	2	1.0	2.0
Tachometer	1	1.0	1.0
Master Switch	1	1.0	1.0
Switch, Blackout	1	2.0	2.0
Switch, Headlight	i	0.5	0.5
Switch, Ignition	i	1.0	1.0
External Start Terminal	i	3.0	3.0
Controls & Displays Weight Su	·	1	18.0
(			
Accessories			
Spare Wheel & Tire		65.0	65.0
Jack		33.0	33.0
Тоwторе		1 4.0	4.0
Shovel		1 4.0	4.0
Axe		1 5.5	5.5
Tool Kit with Tools		1 25.0	25.0
Spare Parts		1 25.0	25.0
Fire Extinguisher and Mount		1 6.6	6.6
First Aid Kit		1 1.5	1.5
Lubricant Bottle		1 0.6	0.6
Camouflage Net w/Support  Accessories Weight Su		1 50.0	50.0 220.2
•			
ON-VEHICLE BASIC EQUIPMENT WEIGHT SUBT	OTAL		344.1
-VEHICLE BASIC CONSUMABLES (Same for all	Mission Va	riants)	
Fuel (In Fuel Tank)	2		136.5
Motor Oil (In Engine)	_	4 2.8	11.0
Motor Oil (Spare Quarts)		4 2.8	11.0
Other Lubricants (In Vehicle)		4 2.8	11.0
Other Lubricants (Spare Quarts)		4 2.8	11.0
Coolant (In Radiator)		6 2.8	16.5
VEHICLE BASIC CONSUMABLES WEIGHT SUBT			197.0

VEHICLE CURB WEIGHT (Same for all Mission Variants) 4554.3 Lbs.

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### RST-V Concept Study Concept/Requirements Report

		RST-V		Ca-il.a		ersonnel	l ac	& Litters
Category/Item	Qtv	Weight (Lbs)	Oty	Strike Weight (Lbs)	Oty	Weight (Lbs)	Oty	Weight (Lbs)
EHICLE-MOUNTED WEAPONS & AMMUNITION								
M60E3 7.62mm Machine Gun								
M60E3	1	20.8	1	20.8	1	20.8	1	20.8
Scissors Swing Arm Mount	1	20.0	1	20.0	ı	20.0	1	20.0
Gun Mount w/200rd Ammo Can	1	12.0	1	12.0	ī	12.0	1	12.0
Cans, Ready Ammo	2	8.0	2	8.0	2	8.0	2	8.0
Cans, Stowed Ammo	2	8.0	8	32.0	2	8.0		0.0
Day / Night Sight	1	11.0	1	11.0	1	11.0	ì	11.0
Spare Barrel	1	12.0	1	12.0	1	12.0	ì	12.0
Weapon CES	1_	3.0 94.8	1	3.0 118.8	1	3.0 94.8	1_	3.0 86.8
M60E3 7.62mm Mochine Gun Weight Subtotal		27.0		110.0		74.5		33.0
M60E3 7.62mm Ammunition Ammunition - Ready (8.75#/100)	400	35.0	400	35.0	400	35.0	400	35.0
Ammunition - Ready (8.75#/100) Ammunition - Stowed (8.75#/100)	400	35.0	1600	140.0	400	35.0	0	0.0
M60E3 Ammunition Weight Subtotal	800	70.0	2000	175.0	800	70.0	400	35.0
MO to Cal Marking Com								
M2 .50 Cal Machine Gun		84.0	1	84.0				
M2 MG w/QCB		40.0	1	40.0				
Pedestal Mount Gun Mount w/200 rd Ammo Can		20.0	i	20.0				
Cans, Ready Ammo		16.0	2	16.0				
		16.0	8	64.0				
Cans, Stowed Ammo		11.0	1	11.0				
Day / Night Sight Spare Barrel		24.0	i	24.0				
Weapon CES		2.2	i	2.2				
M2 .50 Cal Machine Gun Weight Subtotal		213.2		261.2				
M2.50 Cal Ammunition								
Ammunition - Ready (.29# ea)	200	58.0	200	58.0				
Ammunition - Stowed (.29# ca)	200	58.0	800	232.0				
M2 .50 Cal Ammunition Weight Subiotal	400	116.0	1000	290.0				
Mk19 40mm AGL								
Mk19 AGL			1	75.6				
Pedestal Mount			1	40.0				
Gun Mount w/50 rd Ammo Can			1	20.0				
Cans, Ready Ammo			2	16.0				
Cans, Ready Ammo			6	48.0				
Day / Night Sight			1	11.0				
Spare Barrel			1	23.9				
Weapon CES			1	2.2				
MK19 40mm AGL Weight Subtotal (Not Implemented)				236.7				
Mk19 40mm Ammunition			0.0	70.0				
Ammunition - Ready (.75# ea)			96	72.0				
Ammunition - Stowed ( 75# ea)  MK19 40mm Ammunition Weight Subtotal (Not Implemented)			288 384	216.0 288.0				
GAU-19,50 Cal Gatling Gun GAU-19			1	76.0				
			i	40.0				
Pedestal Mount			i	20.0				
Gun Mount w/600 rd Ammo Can			2	16.0				
Cans, Ready Ammo			18	144.0				
Cans, Stowed Ammo			1	11.0				
Day / Night Sight			i	23.9				
Spare Barrel			i	2.2				
Weapon CES (1-19.50 Cal Gailing Gun Weight Subtotal (Not Implemented)			····	333.1				
GAU-19 .50 Cal Ammunition Ammunition - Ready ( 29# ea)			200	58.0				
GAU-19 .50 Cal Ammunition Ammunition - Ready (.29# ea) Ammunition - Stowed (.29# ea)			200 1800	58.0 522.0				

Category/item	Oty	RST-V Weight (Lbs)	<u>Oty</u>	Strike Weight (Lbs)	P. Oty	ersonnel Weight (Lbs)	Log Oty	& Litters Weight (Lbs)
VEHICLE-MOUNTED WEAPONS & AMMUNITION (continu	ued)							
M230B LF 30mm Chain Gun								
M230B			1	384.0				
Pedestal Mount Gun Mount w/90 rd Ammo Can			] 1	<b>80</b> .0 <b>80</b> .0				
Cans, Ready Ammo			3	24.0				
Cans, Stowed Ammo			12	96.0				
Day / Night Sight			1	11.0				
Spare Barrel			1	60.0				
Weapon CES			1	3.0				
M230B LF 30mm Chain Gun Weight Subtotal (Not Implemented)				738.0				
M230B LF 30mm Ammunition								
Ammunition - Ready (.78# ea)			90	70.2				
Ammunition - Stowed (.78# ea)			360	280.8				
M230B LF 30mm Ammunition Weight Subtotal (Not Implemented)			450	351.0				
IICLE-MOUNTED WEAPONS & AMMO WEIGHT SUBTOTAL		494.0		845.0		164.8		121.8
PERSONNEL & PERSONAL EQUIPMENT								
TERSONNEE & TERSONAL EQUIPMENT								
Crew & Passengers								
Crew & Passengers Weight Subtotal	3	525.0	3	525.0	5	875.0	3	525.0
(NATO Average)		•						
Personal Equipment								
Personal Weapon	3	24.0	3	24.0	5	40.0	2	16.0
Personal Weapon CES	3	2.4	3	2.4	5	4.0	2	1.6
Ammunition (5x30 rd magazines)	6	30.0	6	30.0	23	115.0	4	20.0
Helmet	3	13.2	3	13.2	5	22.0	2	8.8
Armored Vest	3	48.0	3	48.0	5	80.0	2	32.0
Personal equipment & clothes	3 2	195.0 4.4	3 2	195.0 4.4	5	325.0	2	130.0
Binoculars  Personal Equipment Weight Subtotal		317.0		317.0	3	592.6	1	2.2 210.6
PERSONNEL & PERSONAL EQUIPMENT WEIGHT SUB	OTĂ	842.0		842.0		1467.6		735.6
·		012.0		542.5		1407.0		733.0
CONSUMABLES Combat Ration	9	9.0	9	9.0	15	15.0	6	6.0
Hexamine Stove	3	1.5	3	1.5	5	2.5	6	3.0
Water (in 5 Gal Jerrycan)	4	150 0	2	75.0	2	75.0	3	112 5
CONSUMABLES WEIGHT SUBTOTAL		160.5		85.5		92.5		121.5
CONFICURABLE ARMOR (Amount qualitable under mainte		ald)						
CONFIGURABLE ARMOR (Amount available under weight configurable ARMOR WEIGHT SUBTOTAL	inresn	0.0	-	882 0	-	882.0	-	0.0
OTHER MISSION EQUIPMENT				302 0		002.0		0.0
Communication/Navigation						*		
MILSATCOM Terminal, kit, antenna & CES	1	40.0	1	40.0				
MILSATCOM Modem	i	· 11.0	í	11.0				
Batteries - VHF (Lithium 8Ah)	i	3.3	i	3.3	1	3.3	1	3.3
Remote Equipment	1	14.0	1	14.0	0	0.0	1	14.0
COMSEC Interface	1	17.6	1	17.6	0	0.0	1	17.6
COMSEC Device, Mount & Adapter	!	4.4	1	4.4	0	0.0	1	4.4
Data input device	1	11.0	1	11.0	0	0.0	1	11.0
Cables Manpack and VHF CES	1	10.0 16.5	i i	10.0 16.5	1	10.0 16.5	1	10.0 16.5
Communication/Navigation Weight Subtotal	<u>'</u>	127.8	<u>'</u>	127.8	<del></del>	29.8		76.8
( )								
Surveillance & Observation Equipment								
Mission Equip (e.g. Laser Designator)	1	5.5		0.0		0.0		0.0
Device Tripod	1	4.4		0.0	_	0.0	•	0.0
Night Vision Goggles	3	13.2 4.4	3 1	13.2 4.4	2	8.8 4.4	2 1	8.8 4.4
Night Vision Binoculars Laser Range Finder	1	4.4 4.4	i	4.4	,	0.0	•	0.0
Batteries for Observation Devices	i	22.0	1	22.0		0.0		0.0
Searchlight w/Power Supply	0	0.0	0	0.0	1	26.4	1	26.4
Surveillance & Observation Eqpt. Weight Subtotal		53.9		44.0		39.6		39.6
Logistics Foundance & Listana								
Logistics Equipment & Litters  Logistics Equipment & Litters Weight Subtotal				<del></del>			-	900.0
OTHER MISSION EQUIPMENT WEIGHT SUBTOTAL		181.7		171.8		69.4		1016 4
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PAYLOAD WEIGHT (lbs)		1678.2	Lbs	2826.3	LDS.	2676.3	Lbs	1995.3